# Software Solutions to Problems on Heat Transfer 

Radiation Heat Transfer - Part I
Dr. M. Thirumaleshwar


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# Software Solutions to Problems on Heat Transfer RADIATION HEAT TRANSFER - PART-I (Laws of Radiation and Radiation properties, View factors \& Radiation energy exchange between black bodies) 

Software Solutions to Problems on Heat Transfer: Radiation Heat Transfer - Part I
(Laws of Radiation and Radiation properties, View factors \& Radiation energy exchange between black bodies)
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## Contents

Preface to vol. 5 6

About the Author 8

About the Software used 10

To the Student 11
$5 \quad$ Radiation heat transfer 13

5A Laws of Radiation and Radiation properties: 57

5B View factors (or, Shape factor, configuration factor) and Radiation energy exchange between Black bodies: 98

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5C Electrical network method \& Radiation energy exchange between gray surfaces:

## Part II

5D
Radiation shields
Part II

5E
'Radiation error' in temperature measurement

References


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## Preface to vol. 5

This is Vol. 5 of the book "Software Solutions to Problems on Heat Transfer".

In Vol. 1, problems on various aspects of CONDUCTION heat transfer were solved with Mathcad, EES, FEHT and EXCEL.

Vol. 2 contained solved problems on the topics of CONVECTION (i.e. Forced convection, Natural or Free convection).

In Vol. 3, we gave solved problems on the topics of BOILING and CONDENSATION. And,

In Vol. 4, we dealt with problems on the topic of HEAT EXCHANGERS.

## Present Vol. viz. RADIATION HEAT TRANSFER contains problems solved on following topics:

## Vol. 5. RADIATION HEAT TRANSFER

Radiation Heat Transfer - Equation summary

## 5.A. Laws of Radiation \& Radiation properties

Problems solved with Mathcad (Prob. 5A. 1 to 5A.10)
Problems solved with EES (Prob. 5A. 11 to 5A.12)
Problems solved with EXCEL (Prob. 5A. 13 to 5A.15)
5.B. View Factors \& Radiation energy exchange between Black bodies:

Problems solved with Mathcad (Prob. 5B.1 to 5B.7)
Problems solved with EES (Prob. 5B. 8 to 5B.10)
Problems solved with EXCEL (Prob. 5B. 11 to 5B.18)
5.C. Electrical network method \& Radiation energy exchange between gray surfaces
5.C. 1 Radiation energy exchange in 2-zone enclosures
5.C. 2 Radiation energy exchange in 3-zone enclosures
5.D. Radiation Shielding

## 5.E. Radiation error in temperature measurement

## References

Here also, problems are solved using the popular software, viz. "Mathcad", "Engineering Equation Solver (EES)", and MS EXCEL spreadsheet. Comments are included generously in the codes so that the logic behind the solutions is clear. An introductory chapter in Part-I gives a brief overview of the software used.

Many Functions are written on View Factors for various geometries, Radiation exchange in 2_zone and 3-zone enclosures, Radiation shields etc. in Mathcad, EES and EXCEL-VBA. These Functions enable us to make the complicated calculations very easily.

As in Vol. I, II , III and IV, emphasis is given not only to solving a given problem but also to parametric analysis and graphical representation of results. Advantage of using Software to solve a variety of problems thus becomes evident. Also, problems with EXCEL spreadsheet are solved in greater details with relevant screen shots for immediate help to students, since EXCEL is available in practically every Personal Computer.

Acknowledgements: Firstly, I thank my students, since it is they who inspired me and motivated me.

Next, my thanks are due to the authorities at St. Joseph Engineering College, for their constant encouragement.

Also, my sincere thanks to Bookboon.com for publishing this book on the Internet. Ms. Sophie and her editorial team have been extremely helpful.

Finally, my heartfelt appreciation to my wife, Kala, but for whose unfailing and thoughtful support and encouragement, this work would not have been possible.
M. Thirumaleshwar

Author
November 2013

## About the Author

Dr. M. Thirumaleshwar graduated in Mechanical Engineering from Karnataka Regional Engineering College, Surathkal, Karnataka, India, in the year 1965. He obtained M.Sc (cryogenics) from University of Southampton, U.K. and Ph.D.(cryogenics) from Indian Institute of Science, Bangalore, India.

He is a Fellow of Institution of Engineers (India), Life Member, Indian Society for Technical Education, and a Foundation Fellow of Indian Cryogenics Council.

He has worked in India and abroad on large projects in the areas involving heat transfer, fluid flow, vacuum system design, cryo-pumping etc.

He worked as Head of Cryogenics Dept. in Bhabha Atomic Research Centre (BARC), Bombay and Centre for Advanced Technology (CAT), Indore, from 1966 to 1992.

He worked as Guest Collaborator with Superconducting Super Collider Laboratory of Universities Research Association, in Dallas, USA from 1990 to 1993.

He also worked at the Institute of Cryogenics, Southampton, U.K. as a Visiting Research Fellow from 1993 to 1994.

He was Head of the Dept. of Mechanical Engineering, Fr. Conceicao Rodrigues Institute of Technology, Vashi, Navi Mumbai, India for eight years.

He also worked as Head of Dept. of Mechanical Engineering and Civil Engineering, and then as Principal, Vivekananda College of Engineering and Technology, Puttur (D.K.), India.

Presently, he is Professor and coordinator of Post-graduate program in the Dept. of Mechanical Engineering in St. Joseph Engineering College, Vamanjoor, Mangalore, India.

A book entitled "Fundamentals of Heat and Mass Transfer" authored by him and published by M/s Pearson Education, India (2006) has been adopted as a Text book for third year engineering students by the Visweswaraya Technological University (V.T.U.), Belgaum, India.

He has also written and published three book-lets entitled as follows:

1. Towards Excellence... How to Study (A Guide book to Students)
2. Towards Excellence... How to teach (A guide book to Teachers)
3. Towards Excellence... Seminars, GD's and Personal Interviews (A guide book to Professional and Management students)

Dr. M. Thirumaleshwar has attended several National and International conferences and has more than 50 publications to his credit.

## About the Software used

Following three software are used while solving problems in this book:

1. Mathcad 2001 (Ref: www.ptc.com)
2. Engineering Equation Solver (EES) (Ref: www.fchart.com), and
3. Finite Element Heat Transfer (FEHT) (Ref: www.fchart.com)
4. MS EXCEL - (2007) Spreadsheet (Ref: Microsoft)

Trial versions of the first three software and detailed Instruction Manuals may be downloaded from the websites indicated. EXCEL is a very popular spreadsheet which comes bundled with MS OFFICE software, and is generally available in every computer.

## See Part-I of this book for brief introduction to these four software.

While the information given there is enough to get going, for detailed instructions one should consult the respective Instruction manuals.


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## To the Student

## Dear Student:

I would like to remind you that Heat Transfer is an important subject useful in many branches of engineering. It is also a subject in which you can score high marks in the examinations, since the question paper generally consists of derivations and numerical problems, almost in the ratio 50:50. Therefore, it requires that:

1) you are thorough with the derivations, and
2) skillful in solving numerical problems.

To be thorough with derivations, you should refer to well known, standard Text books on the subject of Heat Transfer (See References at the end of this book). And, to develop your skill in solving problems... well, that is where I think that this book will help you.

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Effectiveness

This book contains solutions to problems on heat transfer using four popular softwares, viz. Mathcad, Engineering Equation Solver (EES), Finite Element Heat Transfer (FEHT), and EXCEL spreadsheet. Trial versions of Mathcad, EES and FEHT can be downloaded from the websites indicated. EXCEL, which is a part of MS OFFICE, is generally preinstalled in most of the Personal Computers. Problems are chosen from the University question papers and standard heat transfer Text books.

Use of Software in solving problems has many advantages:

1. It helps in logical thinking
2. Problems are solved quickly and accurately
3. Parametric solutions (or 'what-if' solutions) are obtained easily
4. Solutions can be presented in tabular or graphical form, very easily and quickly
5. Once a particular type of problem is solved, solving a similar problem with different data input becomes very easy
6. Ease of getting solutions to problems in tabular or graphical form creates further interest and curiosity on the subject in the minds of students and encourages them to be creative and work further

## How to use this Book?

You need not worry if you don't know about these softwares. Since each problem is solved systematically step by step, and is well commented, just reading through the solution will make the logic of the solution clear to you. That is the most important thing in solving the problems. Then, you must work out the problem yourself, by hand or using the software. Of course, use of software has the above-mentioned advantages. Simply reading the book won't do. Have your favorite Text book nearby, in case you need to refer to it for any formulas or clarifications. There is no other 'easy method'.

As they say, 'there is no gain without pain!'

Lastly, I would like to tell you how greatly I enjoyed solving the problems presented in this book using the softwares mentioned.

I hope that you too will enjoy as much as I did in solving these problems and get benefitted.

Good Luck!

Author

## 5 Radiation heat transfer

## Learning objectives:

1. Radiation mode of heat transfer is radically different from conduction and convection modes in the sense that there is no need for a medium to be present for heat transfer to occur. But, just as conduction and convection heat transfers occur when there is a temperature gradient, net radiation heat transfer also occurs from a higher temperature level to a lower temperature level.
2. Radiation heat transfer is proportional to the fourth power of absolute temperature of the radiating surface. Therefore, radiation becomes the predominant mode of heat transfer when the temperature of the body is high.
3. Our interest in this chapter is on 'thermal radiation', i.e. radiations in the wave length range of 0.1 to 100 microns.
4. Few important applications of radiation heat transfer:
i) industrial heating, such as in furnaces
ii) industrial air-conditioning, where the effect of solar radiation has to be considered in calculating the heat loads
iii) jet engine or gas turbine combustors
iv) industrial drying
v) energy conversion with fossil fuel combustion etc.
5. In this Chapter, first, we solve problems on: fundamental laws governing radiation heat transfer, radiation properties of surfaces, such as absorptivity $(\alpha)$, emissivity $(\varepsilon)$ and transmissivity $(\tau)$. Calculation of 'view factor' for different geometries by analytical / graphical methods and by view factor algebra.
6. Next, we solve problems related to radiation heat transfer between surfaces in two-surface and three-surface enclosures, using the radiation network method. This method greatly simplifies the analysis and gives a 'physical feel' of the problem. Important practical examples of twosurface enclosure are: two infinite, parallel planes, long concentric cylinders and concentric spheres. Furnaces with re-radiating (insulated) surfaces are examples of three-surface enclosure.
7. 'Radiation shielding' to reduce the radiation heat transfer between surfaces is studied next. Importance of radiation shielding in reducing the radiation error in temperature measurement is also studied.
8. Radiation has to be generally considered when the operating temperature level is high; as a rule, it will be prudent to check its relevance in problems involving natural convection and forced convection at high temperatures. Typical example is heat transfer from walls and doors of furnaces. In such problems, concept of 'radiation heat transfer coefficient' simplifies the numerical calculations.
9. Problems on above topics are worked out using Mathcad, EES and EXCEL software. Several Functions/Procedures are written to make the complicated calculations very easy.

## Formulas used: (Ref: 1)

## Absorptivity, Reflectivity and Transmissivity:

$$
\begin{equation*}
\alpha+\rho+\tau=1 \tag{13.1}
\end{equation*}
$$

where,$\quad \alpha=$ absorptivity $=$ fraction of incident radiation absorbed
$\rho=$ reflectivity $=$ fraction of incident radiation reflected
$\tau=$ fraction of incident radiation transmitted

If $\tau=1$, entire radiation passes through the body; such a body is 'transparent' or 'diathermaneous'.

If $\alpha=1$, the body absorbs all the incident radiation and such a body is called a 'black body'.

If $\rho=1$, all the incident radiation is reflected, and it is a perfectly 'white body'.

Emissive power (E): The 'total (or hemispherical) emissive power' is the total thermal energy radiated by a surface per unit time and per unit area, over all the wave lengths and in all directions.



Solid angle: "Solid angle" is defined as a region of a sphere, which is enclosed by a conical surface with the vertex of the cone at the centre of the sphere. See Fig. below:


If a plane area $A$ intercepts the line of propagation of radiation such that the normal to the surface makes an angle $\theta$ with the line of propagation, then we project the incident area normal to the line of propagation, such that, the solid angle is now defined as:

$$
\begin{equation*}
\omega=\frac{\mathrm{A} \cdot \cos (\theta)}{\mathrm{r}^{2}} \mathrm{sr} . \tag{13.3}
\end{equation*}
$$

Note that, $\mathrm{A} \cdot \cos (\theta)=\mathrm{A}_{\mathrm{n}}$ is the projected area of the incident surface, normal to the line of propagation.

Intensity of radiation $\left(I_{b}\right)$ : Intensity of radiation for a black body, $I_{b}$ is defined as the energy radiated per unit time per unit solid angle per unit area of the emitting surface projected normal to the line of view of the receiver from the radiating surface.

Mathematically, this is expressed as:

$$
\begin{align*}
& \mathrm{I}_{\mathrm{b}}=\frac{\mathrm{dQ}_{\mathrm{b}}}{(\mathrm{dA} \cdot \cos (\theta)) \cdot \mathrm{d} \omega} \mathrm{~W} /\left(\mathrm{m}^{2 . s r}\right)  \tag{13.4}\\
& \text { i.e. } \quad \mathrm{I}_{\mathrm{b}}=\frac{\mathrm{dE}_{\mathrm{b}}}{\cos (\theta) \cdot \mathrm{d} \omega} \mathrm{~W} /(\mathrm{m} 2 . \mathrm{sr}) \tag{13.5}
\end{align*}
$$

Note that Emissive power $E_{b}$ of a black body refers to unit surface area whereas Intensity $I_{b}$ of a black surface refers to unit projected area.
$I_{b} \lambda$ is the intensity of blackbody radiation for radiation of a given wave length $\lambda$. And, $I_{b}$ is the summation over all the wave lengths, i.e.

$$
\begin{equation*}
\mathrm{I}_{\mathrm{b}}=\int_{0}^{\infty} \mathrm{I}_{\mathrm{b} \lambda} \mathrm{~d} \lambda \mathrm{~W} /(\mathrm{m} 2 . \mathrm{sr}) \tag{13.6}
\end{equation*}
$$

Lambert's cosine law: "A diffuse surface radiates energy such that the rate of energy radiated in a direction $\theta$ from the normal to the surface is proportional to the cosine of the angle $\theta$ ". i.e.

$$
\mathrm{Q}_{\theta}=\mathrm{Q}_{\mathrm{n}} \cdot \cos (\theta)
$$



### 13.3 Laws of black body radiation:

## Planck's Law for spectral distribution:

Radiation energy emitted by a black surface depends on the wave length, temperature of the surface and the surface characteristics.

Planck's distribution Law relates to the spectral black body emissive power, $\mathbf{E}_{\mathrm{b}} \lambda$ defined as 'the amount of radiation energy emitted by a black body at an absolute temperature T per unit time, per unit surface area, per unit wave length about the wave length $\lambda$. Units of $\mathrm{E}_{\mathrm{b}} \lambda$ are: $\mathrm{W} /\left(\mathrm{m}^{2} . \mu \mathrm{m}\right)$. The first subscript ' b ' indicates black body and the second subscript ' $\lambda$ ' stands for given wavelength, or monochromatic. Planck derived his equation for $\mathrm{E}_{\mathrm{b}} \lambda$ in 1901 in conjunction with his 'quantum theory'.

Planck's distribution law is expressed as:

$$
\begin{equation*}
\mathrm{E}_{\mathrm{b} \lambda}(\lambda)=\frac{\mathrm{C}_{1} \cdot \lambda^{-5}}{\exp \left(\frac{\mathrm{C}_{2}}{\lambda \cdot \mathrm{~T}}\right)-1} \tag{13.7}
\end{equation*}
$$

where,

$$
\mathrm{C}_{1}=3.742 \cdot 10^{8} \quad \mathrm{~W} \cdot \mu \mathrm{~m} 4 / \mathrm{m}^{2}
$$

and,

$$
\mathrm{C}_{2}=1.4387 \cdot 10^{4} \quad \mu \mathrm{~m} . \mathrm{K}
$$

Plots of $\mathrm{E}_{\mathrm{b}} \lambda$ vs. $\lambda$ for a few different temperatures are shown in Fig. below:


## Wein's displacement law:

It is clear from Fig. 13.7 that the spectral distribution of emissive power of a black body at a given absolute temperature goes through a maximum. To find out the value of $\lambda_{\text {max }}$, the wave length at which this maximum occurs, differentiate Planck's equation w.r.t. $\lambda$ and equate to zero. Finally, we get:

$$
\begin{equation*}
\lambda_{\max } \mathrm{T}=2898 \quad \mu \mathrm{~m} . \mathrm{K} \tag{13.11}
\end{equation*}
$$

Wein's displacement law is stated as: "product of absolute temperature and wave length at which emissive power of a black body is a maximum, is constant".

Value of maximum monochromatic emissive power of a black body at a given temperature is obtained by substituting this value of $\lambda_{\max } \mathrm{T}(=2898 \mu \mathrm{~m} . \mathrm{K})$ in Planck's equation. And, we get:

$$
\begin{equation*}
\mathrm{E}_{\mathrm{b} \lambda_{\max }}=1.287 \cdot 10^{-5} \cdot \mathrm{~T}^{5} \quad \mathrm{~W} / \mathrm{m}^{3} \tag{13,12}
\end{equation*}
$$

## Stefan-Boltzmann Law:

$$
\begin{equation*}
E_{b}=\sigma \cdot T^{4} \quad W / m^{2} \tag{13.13}
\end{equation*}
$$

$$
\text { where, } \quad \sigma=5.67 \cdot 10^{-\circ} \quad W /\left(\mathrm{m}^{2} . \mathrm{K}^{4}\right)
$$

$\sigma$ is known as 'Stefan-Boltzmann constant'.

Net radiant energy exchange between two black bodies at temperatures $T_{1}$ and $T_{2}$ is given by:

$$
\begin{equation*}
\mathrm{Q}_{\mathrm{net}}=\sigma \cdot\left(\mathrm{T}_{1}^{4}-\mathrm{T}_{2}^{4}\right) \mathrm{W} / \mathrm{m} 2 \tag{13.14}
\end{equation*}
$$



## Radiation from a wave band:

Often, it is required to know the amount of radiation emitted in a given wave band, i.e. in a wave length interval between $\lambda_{1}$ and $\lambda_{2}$. This is expressed as a fraction of the total emissive power and is written as $\mathrm{F} \lambda_{1-} \lambda_{2}$. Then, we can write:

i.e. $\quad F_{\lambda 1_{-} \lambda 2}=\frac{1}{\sigma \cdot T^{4}} \cdot\left(\begin{array}{ll}\lambda_{2} & \\ & E_{\mathrm{b} \lambda} \mathrm{d} \lambda-\int_{0}^{\lambda_{0}} 1 \\ & \mathrm{E}_{\mathrm{b} \lambda} \mathrm{d} \lambda\end{array}\right)$
i.e.

$$
\begin{equation*}
\mathrm{F}_{\lambda 1 \_\lambda 2} \mathrm{~F}_{0 \_\lambda 2}-\mathrm{F}_{0 \_\lambda 1} \tag{13.5}
\end{equation*}
$$

Above formula is not very convenient to use, since $\mathrm{E}_{\mathrm{b}} \lambda$ depends on absolute temperature T , and it is not practicable to tabulate $\mathrm{F}_{0-\lambda}$ for each T . This difficulty is overcome by expressing $\mathrm{F}_{0-\lambda}$ as follows:

$$
\mathrm{F}_{0-\lambda}=\frac{\int_{0}^{\lambda} \mathrm{E}_{\mathrm{b} \lambda} \mathrm{~d} \lambda}{\sigma \cdot \mathrm{~T}^{4}}=\frac{\int_{0}^{\lambda \cdot \mathrm{T}} \mathrm{E}_{\mathrm{b} \lambda} \mathrm{~d}(\lambda \cdot \mathrm{~T})}{\sigma \cdot \mathrm{T}^{5}}
$$

i.e.

$$
\begin{equation*}
\mathrm{F}_{0 \_\lambda}=\mathrm{f}(\lambda \cdot \mathrm{~T}) \tag{13.16}
\end{equation*}
$$

i.e. Now, $\mathrm{F}_{0} \lambda$ is expressed as a function of the product of wave length and absolute temperature ( $=\lambda . \mathrm{T}$ ) only.

Values of $\mathrm{F}_{0 . \lambda}$ vs. $\lambda . \mathrm{T}$ are tabulated in Table 13.2 below:

|  | Radiation Functions |  |  |
| :---: | :---: | :---: | :---: |
| 2. ( $\mu \mathrm{m} . \mathrm{K}$ ) | $\mathrm{F}_{0-\lambda}$ | 2. ( $\mu \mathrm{m} . \mathrm{K}$ ) $^{\text {a }}$ | $\mathrm{F}_{0-\lambda}$ |
| 400 | 0 | 7000 | 0.8081 |
| 600 | 0 | 7200 | 0.8192 |
| 800 | 0.000016 | 7400 | 0.8295 |
| 1000 | 0.00032 | 7600 | 0.848 |
| 1200 | 0.00213 | 7800 | 0.848 |
| 1400 | 0.0078 | 8000 | 0.8563 |
| 1600 | 0.0197 | 8500 | 0.8746 |
| 1800 | 0.0393 | 9000 | 0.89 |
| 2000 | 0.0667 | 9500 | 0.9031 |
| 2200 | 0.1009 | 10000 | 0.9142 |
| 2400 | 0.1403 | 10500 | 0.9237 |
| 2600 | 0.1831 | 11000 | 0.9319 |
| 2800 | 0.2279 | 11500 | 0.9399 |
| 3000 | 0.2732 | 12000 | 0.9451 |
| 3200 | 0.3181 | 12500 | 0.9505 |
| 3400 | 0.3617 | 13000 | 0.9551 |
| 3600 | 0.4036 | 13500 | 0.9592 |
| 3800 | 0.4434 | 14000 | 0.9628 |
| 4000 | 0.4809 | 14500 | 0.9661 |
| 4200 | 0.516 | 15000 | 0.9689 |
| 4400 | 0.5488 | 16000 | 0.9738 |
| 4600 | 0.5793 | 17000 | 0.9776 |
| 4800 | 0.6075 | 18000 | 0.9808 |
|  |  |  |  |
| 5000 | 0.6337 | 19000 | 0.9834 |
| 5200 | 0.659 | 20000 | 0.9855 |
| 5400 | 0.6804 | 25000 | 0.9922 |
| 5600 | 0.701 | 30000 | 0.9953 |
| 5800 | 0.7201 | 35000 | 0.9969 |
| 6000 | 0.7378 | 40000 | 0.9979 |
| 6200 | 0.7541 | 45000 | 0.9985 |
| 6400 | 0.7962 | 50000 | 0.9989 |
| 6600 | 0.7832 | 75000 | 0.9997 |
| 6800 | 0.7961 | 100000 | 0.9999 |

Table 13.2.

Note in the above Table that the units of product $\lambda . T$ is (micron.Kelvin).

Therefore,

$$
\begin{equation*}
\mathrm{F}_{\lambda 1 \_\lambda 2}=\mathrm{F}_{\lambda 1 . \mathrm{T}}^{-} \lambda 2 . \mathrm{T}=\frac{1}{\sigma} \cdot\left[\int_{0}^{\lambda 2 \cdot \mathrm{~T}} \frac{\mathrm{E}_{\mathrm{b} \lambda}}{\mathrm{~T}^{5}} \mathrm{~d}(\lambda \cdot \mathrm{~T})-\int_{0}^{\lambda 1 \cdot \mathrm{~T}} \quad \mathrm{E}_{\mathrm{b} \lambda} \mathrm{~d}(\lambda \cdot \mathrm{~T})\right] \tag{13.17}
\end{equation*}
$$

i.e.

Relation between Radiation intensity and Emissive power:

$$
\begin{equation*}
\mathrm{E}_{\mathrm{b}}=\pi \cdot \mathrm{I}_{\mathrm{b}} \tag{13.21}
\end{equation*}
$$

i.e. Total emissive power of a black (diffuse) surface is equal to $\pi$ times the intensity of radiation.

This is an important relation, which will be used while calculating the view factors required to determine net energy exchange between surfaces.

Total hemispherical emissivity $(\varepsilon)$ of a surface is the average emissivity over all directions and all wave lengths and is expressed as:

$$
\begin{equation*}
\varepsilon(T)=\frac{E(T)}{E_{b}(T)}=\frac{E(T)}{\sigma \cdot T^{4}} \tag{13.22}
\end{equation*}
$$

where $E(T)$ is the emissive power of the real surface.

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Similarly, spectral emissivity is defined as:

$$
\begin{equation*}
\varepsilon_{\lambda}(T)=\frac{E_{\lambda}(T)}{E_{b \lambda}(T)} \tag{13.23}
\end{equation*}
$$

Emissivity values for a few surfaces at room temperature are given in handbooks.

A surface is said to be gray if its properties are independent of wave length, and a surface is diffuse if its properties are independent of direction.

## To calculate the average emissivity:

Let the variation of spectral emissivity with wave length be as follows:

$$
\begin{aligned}
& \varepsilon_{1}=\text { constant, } 0 \leq \lambda \leq \lambda_{1} \\
& \varepsilon_{2}=\text { constant, } \lambda_{1} \leq \lambda \leq \lambda_{2} \\
& \varepsilon_{3}=\text { constant, } \lambda_{2} \leq \lambda \leq \infty
\end{aligned}
$$

Then, the average emissivity is calculated using eqn (13.24) as follows:

$$
\begin{equation*}
\text { i.e. } \varepsilon(\mathrm{T})=\varepsilon_{1} \cdot \mathrm{~F}_{0} \lambda_{1}(\mathrm{~T})+\varepsilon_{2} \cdot \mathrm{~F} \lambda_{1-} \lambda_{2}(\mathrm{~T})+\varepsilon_{3} \cdot \mathrm{~F} \lambda_{2-} \infty(\mathrm{T}) \tag{13.26}
\end{equation*}
$$

Factors $\mathrm{F}_{0} \lambda_{1}(\mathrm{~T})$ etc. can easily be determined using Tables in handbooks.

### 13.3.7 Kirchoff's Law:

$$
\varepsilon(T)=\alpha(T) \ldots .(13.27)
$$

Kirchoff's Law states that the total hemispherical emissivity, $\varepsilon$ of a gray surface at a temperature T is equal to its absorptivity, $\alpha$ for black body radiation from a source at the same temperature $T$.

## The View factor:

View factor is also known by other names such as: 'configuration factor', 'shape factor,' 'angle factor' etc.

View factor is defined as the fraction of radiant energy leaving one surface which strikes a second surface directly.

Here, 'directly' means that reflection or re-radiated energy is not considered. View factor is denoted by $F_{12}$, where the first subscript, 1 stands for the emitting surface, and the second subscript, 2 stands for the receiving surface.

## View factor between Infinitesimal areas:

(Such a situation is encountered even when finite areas are involved, when the distance between these two areas ' $r$ ', is very large.)

$$
\begin{equation*}
\mathrm{F}_{\mathrm{dA} 1_{-} \mathrm{dA} 2}=\frac{\cos \left(\phi_{1}\right) \cdot \cos \left(\phi_{2}\right) \cdot \mathrm{dA}_{2}}{\pi \cdot \mathrm{r}^{2}} \tag{!3.31}
\end{equation*}
$$

Note that the View factor involves geometrical quantities only.

## View factor between Infinitesimal and finite area:

$$
F_{d A 1-A 2}=\int_{A 2} \frac{\cos \left(\phi_{1}\right) \cos \left(\phi_{2}\right) d A_{2}}{\pi r^{2}} \ldots \ldots(13.32)
$$

Practical situation of calculating view factors between infinitesimal to finite areas are encountered the case of a small thermocouple bead located inside a pipe or a small, spherical point source radiator located by the side of a wall etc.

## View factor from Finite to finite area:

$$
F_{A 1-A 2}=\frac{1}{\pi A_{1}} \int_{A 1} \int_{A 2} \frac{\cos \left(\phi_{1}\right) \cos \left(\phi_{2}\right)}{r^{2}} d A_{1} d A_{2} \cdots \cdots(13.33)
$$

In general, we write eqn. (13.33) compactly as:

$$
F_{12}=\frac{1}{\pi A_{1}} \int_{A 1} \int_{A 2} \frac{\cos \left(\phi_{1}\right) \cos \left(\phi_{2}\right)}{r^{2}} d A_{1} d A_{2} \cdots \cdots(13.34)
$$

## Reciprocity theorem:

$$
\begin{equation*}
A_{1} \cdot F_{12}=A_{2} \cdot F_{21} \tag{13.36}
\end{equation*}
$$

This is a very useful and important relation. It helps one to find out one of the view factors when the other one is known. In practice, one of the view factors which is easier to calculate is obtained first, and the other view factor is found out next, by using the reciprocity theorem.

Note: It is easier to remember the view factor relation given in eqn. (13.34) as:

$$
A_{11} F_{12}=\int_{A 1} \int_{A 2} \frac{\cos \left(\phi_{1}\right) \cos \left(\phi_{2}\right)}{\pi r^{2}} d A_{1} d A_{2} \cdots \cdots(13.37)
$$

Radiation energy exchange between black bodies:

$$
Q_{\text {net }}=A_{1} \cdot F_{12} \cdot \sigma \cdot T_{1}^{4}-A_{2} \cdot F_{21} \cdot \sigma \cdot T_{2}^{4}
$$

$$
\text { But, } \quad A_{1} \cdot F_{12}=A_{2} \cdot F_{21} \quad \text { by reciprocity theorem }
$$

Therefore,

$$
Q_{\text {net }}=A_{1} \cdot F_{12} \cdot \sigma \cdot\left(\mathrm{~T}_{1}^{4}-\mathrm{T}_{2}^{4}\right\}=\mathrm{A}_{2} \cdot \mathrm{~F}_{21} \cdot \cdot \cdot\left(\mathrm{~T}_{1}^{4}-\mathrm{T}_{2}^{4}\right\} \quad W \ldots(13.40)
$$

## Properties of View factor and view factor algebra:

## Properties of view factor:

1) The view factor depends only on the geometrics of bodies involved and not on their temperatures or surface properties.
2) Between two surfaces that exchange energy by radiation, the mutual shape factors are governed by the 'reciprocity relation', viz. $A_{1} \cdot F_{12}=A_{2} \cdot F_{21}$.

3) When a convex surface $\mathbf{1}$ is completely enclosed by another surface $\mathbf{2}$, it is clear from Fig. 13.14(a) that all of the radiant energy emitted by surface 1 is intercepted by the enclosing surface 2 . Therefore, view factor of surface $\mathbf{1}$ w.r.t. surface $\mathbf{2}$ is equal to unity. i.e. $\mathbf{F}_{12}=\mathbf{1}$. And, the view factor of surface 2 w.r.t. surface 1 is then easily calculated by applying the reciprocity relation, i.e. $\mathrm{A}_{1} \cdot 1=\mathrm{A}_{2} \cdot \mathrm{~F}_{21}$, or, $\mathrm{F}_{21}=\mathrm{A}_{1} / \mathrm{A}_{2}$.
4) Radiation emitted from a flat surface never falls directly on that surface (see Fig. 13.14 (b)), i.e. View factor of a flat surface w.r.t. itself is equal to zero, i.e. $\mathbf{F}_{11}=\mathbf{0}$. This is valid for a convex surface too, as shown in Fig. 13.14(c).
5) For a concave surface, it is clear from Fig. 13.14(d) that $F_{11}$ is not equal to zero since some fraction of radiation emitted by a concave surface does fall on that surface directly.

(a) Surface 1 completely enclosed by surface 2

(b) Flat surface
(c) Convex surface
(d) Concave surface

Fig. 13.14 View factors for a few surfaces
6) If two, plane surfaces A1 and A2 are parallel to each other and separated by a short distance between them, practically all the radiation issuing from surface 1 falls directly on surface 2, and vice -versa. Therefore, $\mathbf{F}_{12}=\mathbf{F}_{21}=\mathbf{1}$


Fig. 13.15 View factors for subdivided surfaces
7) When the radiating surface 1 is divided into, say, two sub-areas A3 and A4, as shown in Fig. 13.15(a), we have:
$\mathrm{A}_{1} \cdot \mathrm{~F}_{12}=\mathrm{A}_{3} \cdot \mathrm{~F}_{32}+\mathrm{A}_{4} \cdot \mathrm{~F}_{42}$
Obviously, $\mathrm{F}_{12} \neq \mathrm{F}_{32}+\mathrm{F}_{42}$.
8) Instead, if the receiving surface A2 is sub-divided into parts A3 and A4 as shown in Fig. 13.15 (b), we have:

$$
\begin{equation*}
\mathrm{A}_{1} \cdot \mathrm{~F}_{12}=\mathrm{A}_{1} \cdot \mathrm{~F}_{13}+\mathrm{A}_{1} \cdot \mathrm{~F}_{14} \tag{13.42}
\end{equation*}
$$

i.e.

$$
F_{12}=F_{13}+F_{14}
$$

i.e. view factor from the emitting surface 1 to a sub-divided receiving surface is simply equal to the sum of the individual shape factors from the surface 1 to the respective parts of the receiving surface. This is known as 'Superposition rule'.

## "I studied English for 16 years but... <br> ...I finally learned to speak it in just six lessons" Jane, Chinese architect



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9) Symmetry rule: If two (or more) surfaces are symmetrically located w.r.t. the radiating surface 1 , then the view factors from surface 1 to these symmetrically located surfaces are identical. A close inspection of the geometry will reveal if there is any symmetry in a given problem.
10) Summation rule: Since radiation energy is emitted from a surface in all directions, invariably, we consider the emitting surface to be part of an enclosure. Even if there is an opening, we consider the opening as a surface with the radiative properties of that opening. Then, the conservation of energy principle requires that sum of all the view factors from the surface 1 to all other surfaces forming the enclosure, must be equal to 1 . See Fig. 13.16, where the interior surface of a completely enclosed space is sub-divided into n parts, each of area A1, A2, A3...An.


$$
\sum_{j=1}^{n} F_{i j}=1 \quad i=1,2,3, \ldots . n
$$

Fig. 13.16 View factor - summation rule for radiation in an enclosure

Then,

$$
\begin{align*}
& \mathrm{F}_{11}+\mathrm{F}_{12}+\ldots . .+\mathrm{F}_{1 \mathrm{n}}=1 \\
& \mathrm{~F}_{21}+\mathrm{F}_{22}+\ldots .+\mathrm{F}_{2 \mathrm{n}}=1 \\
& \mathrm{~F}_{\mathrm{n} 1}+\mathrm{F}_{\mathrm{n} 2}+\ldots . .+\mathrm{F}_{\mathrm{nn}}=1 \ldots \tag{13.43}
\end{align*}
$$

i.e.

$$
\sum_{j=1}^{n} F_{i j}=1 i=1,2,3, \ldots \ldots n \ldots \ldots .(13.44)
$$

11) In an enclosure of ' $n$ ' black surfaces, maintained at temperatures $T_{1}, T_{2}, \ldots . T_{n}$, net radiation from any surface, say surface 1 , is given by summing up the net radiation heat transfers from surface 1 to each of the other surfaces of the enclosure:

$$
\begin{align*}
\mathrm{Q}_{1_{\mathrm{net}}}= & \mathrm{A}_{1} \cdot \mathrm{~F}_{12} \cdot \sigma \cdot\left(\mathrm{~T}_{1}{ }^{4}-\mathrm{T}_{2}^{4}\right)+\mathrm{A}_{1} \cdot \mathrm{~F}_{13} \cdot \sigma \cdot\left(\mathrm{~T}_{1}^{4}-\mathrm{T}_{3}^{4}\right)+\mathrm{A}_{1} \cdot \mathrm{~F}_{14} \cdot \sigma \cdot\left(\mathrm{~T}_{1}^{4}-\mathrm{T}_{4}^{4}\right) \ldots \\
& +\mathrm{A}_{1} \cdot \mathrm{~F}_{1 \mathrm{n}} \cdot \sigma \cdot\left(\mathrm{~T}_{1}^{4}-\mathrm{T}_{\mathrm{n}}^{4}\right) \tag{13.45}
\end{align*}
$$

Note: Often, while solving radiation problems, determination of the view factor is the most difficult part. It will be useful to keep in mind the definition of view factor, summation rule, reciprocity relation, superposition rule and symmetry rule while attempting to find out the view factors.

## View factor relations for a few geometries are given below: (Ref: Cengel)

TABLE 22-1
View factor expressions for some common geometries of finite size (3D)

| Geometry | Relation |
| :---: | :---: |
| Aligned parallel rectangles | $\bar{X}=X / L \text { and } \bar{Y}=Y / L$ $\begin{aligned} F_{i \rightarrow j} & =\frac{2}{\pi \bar{X} \bar{Y}}\left\{\ln \left[\frac{\left(1+\bar{X}^{2}\right)\left(1+\bar{Y}^{2}\right)}{1+\bar{X}^{2}+\bar{Y}^{2}}\right]^{1 / 2}\right. \\ & +\bar{X}\left(1+\bar{Y}^{2}\right)^{1 / 2} \tan ^{-1} \frac{\bar{X}}{\left(1+\bar{Y}^{2}\right)^{1 / 2}} \\ & +\bar{Y}\left(1+\bar{X}^{2}\right)^{1 / 2} \tan ^{-1} \frac{\bar{Y}}{\left(1+\bar{X}^{2}\right)^{1 / 2}} \\ & \left.-\bar{X} \tan ^{-1} \bar{X}-\bar{Y} \tan ^{-1} \bar{Y}\right\} \end{aligned}$ |
| Coaxial parallel disks | $\begin{aligned} R_{i} & =r_{i} / L \text { and } R_{j}=r_{j} / L \\ S & =1+\frac{1+R_{j}^{2}}{R_{i}^{2}} \\ F_{i \rightarrow j} & =\frac{1}{2}\left\{S-\left[S^{2}-4\left(\frac{r_{j}}{r_{i}}\right)^{2}\right]^{1 / 2}\right\} \end{aligned}$ |


| Perpendicular rectangles <br> with a common edge | $H=Z / X$ and $W=Y / X$ <br> $\quad$$F_{i \rightarrow j}$ $=\frac{1}{\pi W}\left(W \tan ^{-1} \frac{1}{W}+H \tan ^{-1} \frac{1}{H}\right.$ <br>  $-\left(H^{2}+W^{2}\right)^{1 / 2} \tan ^{-1} \frac{1}{\left(H^{2}+W^{2}\right)^{1 / 2}}$ <br>  $+\frac{1}{4} \ln \left\{\frac{\left(1+W^{2}\right)\left(1+H^{2}\right)}{1+W^{2}+H^{2}}\right.$ <br>  $\times\left[\frac{W^{2}\left(1+W^{2}+H^{2}\right)}{\left(1+W^{2}\right)\left(W^{2}+H^{2}\right)}\right]^{W^{2}}$ <br>  $\left.\left.\times\left[\frac{H^{2}\left(1+H^{2}+W^{2}\right)}{\left(1+H^{2}\right)\left(H^{2}+W^{2}\right)}\right]^{H^{2}}\right\}\right)$ |
| :--- | :--- |

TABLE 22-2
View factor expressions for some infinitely long (2D) geometries

| Geometry | Relation |
| :---: | :---: |
| Parallel plates with midlines connected by perpendicular line | $\begin{aligned} W_{i} & =w_{i} / L \text { and } W_{j}=w_{j} / L \\ F_{i \rightarrow j} & =\frac{\left.\left[\left(W_{i}+W_{j}\right)^{2}+4\right]^{1 / 2}-\left(W_{j}-W_{i}\right)^{2}+4\right]^{1 / 2}}{2 W_{i}} \end{aligned}$ |
| Inclined plates of equal width and with a common edge | $F_{i \rightarrow j}=1-\sin \frac{1}{2} \alpha$ |
| Perpendicular plates with a common edge | $F_{i \rightarrow j}=\frac{1}{2}\left\{1+\frac{w_{j}}{w_{i}}-\left[1+\left(\frac{w_{j}}{w_{i}}\right)^{2}\right]^{1 / 2}\right\}$ |


| Three-sided enclosure | $F_{i \rightarrow j}=\frac{w_{i}+w_{j}-w_{k}}{2 w_{i}}$ |
| :---: | :---: |
| Infinite plane and row of cylinders $\oplus \oplus \oplus \oplus \oplus_{\not{-s-1} \overbrace{j}^{D}}^{-\digamma^{-1}}$ | $\begin{aligned} F_{i \rightarrow j} & =1-\left[1-\left(\frac{D}{s}\right)^{2}\right]^{1 / 2} \\ & +\frac{D}{s} \tan ^{-1}\left(\frac{s^{2}-D^{2}}{D^{2}}\right)^{1 / 2} \end{aligned}$ |



View factor for two aligned parallel, rectangles of equal size:


View factor between two coaxial parallel discs:


View factor between two perpendicular rectangles with a common edge:


View factors for two concentric cylinders of finite length: (a) outer cylinder to inner cylinder (b) outer cylinder to itself:


## Radiation heat exchange between gray surfaces: (Ref: 1, 3)

## Radiation exchange between small, gray surfaces:

$$
\begin{equation*}
\mathrm{Q}_{12}=\varepsilon_{1} \varepsilon_{2} \cdot \mathrm{~A}_{1} \cdot \mathrm{~F}_{12} \cdot \sigma \cdot\left(\mathrm{~T}_{1}^{4}-\mathrm{T}_{2}^{4}\right\}=_{1} \varepsilon_{2} \cdot \mathrm{~A}_{2} \cdot \mathrm{~F}_{21} \cdot \sigma \cdot\left(\mathrm{~T}_{1}^{4}-\mathrm{T}_{2}^{4}\right\} \tag{13.47}
\end{equation*}
$$

## Remember that:

$$
A_{1} \cdot F_{12}=A_{2} \cdot F_{21} \quad \ldots \text { by reciprocity }
$$

The product, $\left(\varepsilon_{1} \cdot \varepsilon_{2}\right)$ is known as 'equivalent emissivity ( $\varepsilon_{\text {eq }}$ ) for a system of two 'small' gray bodies.

## The electrical network method:

## Definitions:

1. Irradiation, (G): is the total radiation incident upon a surface per unit time, per unit area ( $\mathrm{W} / \mathrm{m}^{2}$ ).
2. Radiosity, ( J ): is the total radiation leaving a surface, with no regard for its origin (i.e., reflected plus emitted from the surface) per unit time, per unit area $\left(\mathrm{W} / \mathrm{m}^{2}\right)$.



Fig. 13.25 Irradiation and Radiosity
3. Surface resistance:

$$
Q=\frac{E_{b}-J}{\frac{(1-\varepsilon)}{A \cdot \varepsilon}} \quad W \ldots .(13.48)
$$

By analogy with Ohm's law:

$$
R=\frac{(1-\varepsilon)}{A \cdot \varepsilon} \quad \text {....surface resistance }
$$


$\mathrm{J}_{\mathrm{i}}=\mathrm{E}_{\mathrm{bi}} \quad$....for a black body....(13.49)
$\mathrm{J}_{\mathrm{i}}=\mathrm{E}_{\mathrm{bi}}=\sigma \cdot \mathrm{T}_{\mathrm{i}}^{4} \quad$....for a re-radiating surface....(13.50)
4. Space resistance:


$$
\begin{equation*}
\mathrm{Q}_{\mathrm{ij}}=\frac{\mathrm{J}_{\mathrm{i}}-\mathrm{J}_{\mathrm{j}}}{\frac{1}{\left(\mathrm{~A}_{\mathrm{i}} \mathrm{~F}_{\mathrm{ij}}\right)}} \tag{13.52}
\end{equation*}
$$

By analogy with Ohm's law:

$$
\begin{align*}
Q_{i j} & =\frac{J_{i}-J_{j}}{R_{i j}} \quad W \\
\text { where, } \quad R_{i j} & =\frac{1}{A_{i} F_{i j}} \quad \ldots(1 \tag{13.53}
\end{align*}
$$

$\mathrm{R}_{\mathrm{ij}}$ is known as 'space resistance' and it represents the resistance to radiative heat flow between the radiosity potentials of the two surfaces, due to their relative orientation and spacing.

For a $\mathbf{N}$ surface enclosure, net heat transfer from surface i should be equal to the sum of net heat transfers from that surface to the remaining surfaces. i.e.

$$
\begin{equation*}
Q_{i}=\sum_{j=1}^{N} Q_{i j}=\sum_{j=1}^{N} A_{i} \cdot F_{i j}\left(J_{i}-J_{j}\right)=\sum_{j=1}^{N} \frac{J_{i}-J_{j}}{R_{i j}} \tag{13.54}
\end{equation*}
$$

i.e. $\frac{E_{b i}-J_{i}}{R_{i}}=\sum_{j=1}^{N} \frac{J_{i}-J_{j}}{R_{i j}}$
where $\mathrm{R}_{\mathrm{i}}$ is the surface resistance and $\mathrm{R}_{\mathrm{ij}}$ is the space resistance.
This situation is shown in Fig. (13.27).


Fig. 13.27 Radiation heat transfer from surface $i$ to other surfaces in a $N$-surface enclosure

Radiation heat exchange in two-zone enclosures: (Ref: 1, 3)


$$
\begin{aligned}
& E_{b 1} \xrightarrow[\sim N i]{\stackrel{\dot{Q}_{1}}{\sim}} \quad J_{1} \xrightarrow{J_{12}} \quad J_{2} \quad \xrightarrow{\dot{Q}_{2}} E_{b 2} \\
& R_{1}=\frac{1-\varepsilon_{1}}{A_{1} \varepsilon_{1}} \quad R_{12}=\frac{1}{A_{1} F_{12}} \quad R_{2}=\frac{1-\varepsilon_{2}}{A_{2} \varepsilon_{2}}
\end{aligned}
$$

$$
Q_{12}=Q_{1}=-Q_{2}=\frac{E_{b 1}-E_{b 2}}{R_{1}+R_{12}+R_{2}}
$$

i.e. $\quad Q_{12}=\frac{E_{b 1}-E_{b 2}}{\frac{1-\varepsilon_{1}}{A_{1} \varepsilon_{1}}+\frac{1}{A_{1} F_{12}}+\frac{1-\varepsilon_{2}}{A_{2} \varepsilon_{2}}}$
i.e. $\quad Q_{12}=\frac{\sigma \cdot\left(T_{1}{ }^{4}-T_{2}{ }^{4}\right)}{\frac{1-\varepsilon_{1}}{A_{1} \cdot \varepsilon}+\frac{1}{A_{1} \cdot F_{12}}+\frac{1-\varepsilon_{2}}{A_{2} \varepsilon_{2}}}$

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## A few special cases of two-surface enclosure:

Here, basic, governing eqn. is eqn.(13.56), which is modified depending upon the case considered.

## Case (i): Radiant heat exchange between two black surfaces:

For a black body, $\varepsilon=1$, and $J=E_{b}$, as explained earlier. i.e. surface resistance
$[=(1-\varepsilon) /($ A. $\varepsilon)]$ of a black body is zero.

Then, from eqn. (13.56), we get:

$$
\begin{aligned}
& Q_{12}=\frac{\sigma \cdot\left\{\mathrm{T}_{1}{ }^{4}-\mathrm{T}_{2}^{4}\right\}}{\frac{1}{\mathrm{~A}_{1} \cdot \mathrm{~F}_{12}}} \\
& \text { i.e. } \quad \mathrm{Q}_{12}=\mathrm{A}_{1} \cdot \mathrm{~F}_{12} \cdot \sigma \cdot\left(\mathrm{~T}_{1}{ }^{4}-\mathrm{T}_{2}{ }^{4}\right\} \quad \begin{array}{l}
\text { W...for two black surfaces forming an } \\
\text { enclosure....(13.57) }
\end{array}
\end{aligned}
$$

Next, we shall consider four cases of practical interest where the view factor between the inner surface 1 and the outer surface 2 (i.e. $F_{12}$ ) is equal to 1 , and also the net radiation from a gray cavity:

## Case (ii): Radiant heat exchange for a small object in a large cavity:

A practical example of a small object in a large cavity is the case of a steam pipe passing through a large plant room.

For this case, we have:

$$
\begin{aligned}
\frac{A_{1}}{A_{2}} & =0 \\
\text { and, } \quad F_{12} & =1
\end{aligned}
$$

And, eqn.(13.56) becomes:

$$
\begin{equation*}
\mathrm{Q}_{12}=\mathrm{A}_{1} \cdot \sigma \cdot \varepsilon 1 \cdot\left(\mathrm{~T}_{1}^{4}-\mathrm{T}_{2}^{4}\right) \quad \text {...for small object in a large cavity... } \tag{13.58}
\end{equation*}
$$

## Case (iii): Radiant heat exchange between infinitely large parallel plates:

In this case, $A_{1}=A_{2}=A$, say, and $F_{12}=1$

Then, eqn. (13.56) becomes:
$Q_{12}=\frac{A \cdot \sigma \cdot\left(T_{1}^{4}-T_{2}^{4}\right\}}{\frac{1}{\varepsilon_{1}}+\frac{1}{\varepsilon_{2}}-1} \quad$...for infinitely large parallel plates..

## Case (iv): Radiant heat exchange between infinitely long concentric cylinders:

In this case:

$$
\mathrm{F}_{12}=1
$$

Then, eqn. (13.56) becomes:

$$
\begin{equation*}
\mathrm{Q}_{12}=\frac{\mathrm{A}_{1} \cdot \sigma \cdot\left(\mathrm{~T}_{1}^{4}-\mathrm{T}_{2}^{4}\right)}{\frac{1}{\varepsilon_{1}}+\left(\frac{\mathrm{A}_{1}}{\mathrm{~A}_{2}}\right) \cdot\left(\frac{1}{\varepsilon_{2}}-1\right)} \tag{13.60}
\end{equation*}
$$

....for infinitely long concentric cylinders
where

$$
\frac{\mathrm{A}_{1}}{\mathrm{~A}_{2}}=\frac{\mathrm{r}_{1}}{\mathrm{r}_{2}}
$$

Remember that $A_{1}$ refers to the inner (or enclosed) surface.

## Case (v): Radiant heat exchange between concentric spheres:

In this case:

$$
\mathrm{F}_{12}=1
$$

Then, eqn. (13.56) becomes

$$
\mathrm{Q}_{12}=\frac{\mathrm{A}_{1} \cdot \sigma \cdot\left(\mathrm{~T}_{1}^{4}-\mathrm{T}_{2}^{4}\right)}{\frac{1}{\varepsilon_{1}}+\left(\frac{\mathrm{A}_{1}}{\mathrm{~A}_{2}}\right) \cdot\left(\frac{1}{\varepsilon_{2}}-1\right)} \quad \ldots . \text { for concentric spheres.....(13.61) }
$$

where

$$
\frac{\mathrm{A}_{1}}{\mathrm{~A}_{2}}=\left(\frac{\mathrm{r}_{1}}{\mathrm{r}_{2}}\right)^{2}
$$

Remember, again, that $A_{1}$ refers to the inner (or enclosed) surface.

## Case (vi): Energy radiated from a gray cavity:


(a) Gray cavity

(b) Radiation network

$$
\begin{aligned}
& \mathrm{R}_{1}=\left(1-\epsilon_{1}\right) /\left(\mathrm{A}_{1} \cdot \epsilon_{1}\right) \\
& \mathrm{R}_{12}=1 /\left(\mathrm{A}_{1} \cdot \mathrm{~F}_{12}\right)
\end{aligned}
$$

$$
Q_{12}=A_{1} \cdot \varepsilon \cdot \cdot \cdot \cdot T_{1}^{4} \cdot\left[\frac{1-F_{11}}{1-\left(1-\varepsilon_{1}\right) \cdot F_{11}}\right]
$$

W....net radiation from
gray cavity..(13.62)

Eqn. (13.62) is an important result, which gives net radiation from a gray cavity to surrounding space. If it is desired, for example, to calculate the net radiation from a blind hole drilled in a flange, then the relation to use is the eqn. (13.62).



## Radiation heat exchange in three zone enclosures: [Ref: 1]




## Temperature of each surface is known; i.e. emissive power $\mathrm{E}_{\mathrm{b}}$ for each surface is known:

Then, the problem reduces to determining the radiosities $J_{1}, J_{2}$ and $J_{3}$.
This is done by applying Kirchoff's law of d.c. circuits to each node: i.e. sum of the currents (or, rate of heat transfers) entering into each node is zero:

$$
\begin{equation*}
\text { Node } \mathrm{J}_{1}: \quad \frac{\mathrm{E}_{\mathrm{b} 1}-\mathrm{J}_{1}}{\mathrm{R}_{1}}+\frac{\mathrm{J}_{2}-\mathrm{J}_{1}}{\mathrm{R}_{12}}+\frac{\mathrm{J}_{3}-\mathrm{J}_{1}}{\mathrm{R}_{13}}=0 \tag{13.63,a}
\end{equation*}
$$

Node $\mathrm{J}_{2}: \quad \frac{\mathrm{E}_{\mathrm{b} 2}-\mathrm{J}_{2}}{\mathrm{R}_{2}}+\frac{\mathrm{J}_{1}-\mathrm{J}_{2}}{\mathrm{R}_{12}}+\frac{\mathrm{J}_{3}-\mathrm{J}_{2}}{\mathrm{R}_{23}}=0$

Node $\mathrm{J}_{3}: \quad \frac{\mathrm{E}_{\mathrm{b} 3}-\mathrm{J}_{3}}{\mathrm{R}_{3}}+\frac{\mathrm{J}_{1}-\mathrm{J}_{3}}{\mathrm{R}_{13}}+\frac{\mathrm{J}_{2}-\mathrm{J}_{3}}{\mathrm{R}_{23}}=0$
Solving these three eqns. simultaneously, we get $\mathrm{J}_{1}, \mathrm{~J}_{2}$ and $\mathrm{J}_{3}$.

Once the radiosities are known, expressions for net heat flows between the surfaces are:

$$
\begin{align*}
& Q_{12}=\frac{J_{1}-J_{2}}{R_{12}}=\frac{J_{1}-J_{2}}{\frac{1}{A_{1} \cdot F_{12}}} \\
& Q_{13}=\frac{J_{1}-J_{3}}{R_{13}}=\frac{J_{1}-J_{3}}{\frac{1}{A_{1} \cdot F_{13}}} \\
& Q_{23}=\frac{J_{2}-J_{3}}{R_{23}}=\frac{J_{1}-J_{2}}{\frac{1}{A_{2} \cdot F_{23}}} \tag{13.64,~b}
\end{align*}
$$

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And, net heat flow from each surface is:

$$
\begin{align*}
& Q_{1}=\frac{E_{b 1}-J_{1}}{R_{1}}=\frac{E_{b 1}-J_{1}}{\left(\frac{1-\varepsilon_{1}}{A_{1} \cdot \varepsilon_{1}}\right)}  \tag{13.65,a}\\
& Q_{2}=\frac{E_{b 2}-J_{2}}{R_{2}}=\frac{E_{b 2}-J_{2}}{\left(\frac{1-\varepsilon_{2}}{A_{2} \cdot \varepsilon_{2}}\right)}  \tag{13.65,b}\\
& Q_{3}=\frac{E_{b 3}-J_{3}}{R_{3}}=\frac{E_{b 3}-J_{3}}{\left(\frac{1-\varepsilon_{3}}{A_{3} \cdot \varepsilon_{3}}\right)} \tag{13.65,c}
\end{align*}
$$

To the general eqns (13.64), apply the constraints such as:

If the surface is black or re-radiating: $\mathrm{J}_{\mathrm{i}}=\mathrm{E}_{\mathrm{bi}}=\sigma \cdot T_{i}^{4}$.

And, $Q_{i}=0$ for a re-radiating surface.

If $Q_{i}$ at any surface is specified instead of the temperature (i,e. $\left.E_{b i}\right)$, then, $\left(E_{b i}-J_{i}\right) / R_{i}$ is replaced by $Q_{i}$.

## Few special cases of three-zone enclosures:

1. Two black surfaces connected to a third refractory surface:


Fig.(a)


Fig.(b)
Figure 13.33 Two black surfaces connected by a thrid re-radiating surface and its radiation network

Fig. (13.33,a) shows the radiation network for this case. The radiation network is drawn very easily by remembering the usual principles: for a black surface, the surface resistance is zero, i.e. $\mathrm{E}_{\mathrm{b}}=\mathrm{J}$. For a reradiating surface too, $\mathrm{E}_{\mathrm{b}}=\mathrm{J}$, as already explained; further, for a re-radiating surface, $\mathrm{Q}=0$. Between two given surfaces, the radiosity potentials are connected by the respective space resistances, as shown. It may be observed that the system reduces to a series-parallel circuit of resistances as shown in Fig. (13.33,b).

We get:

$$
\begin{equation*}
\left.\mathrm{Q}_{12}=\sigma \cdot\left\{\mathrm{T}_{1}^{4}-\mathrm{T}_{2}^{4}\right\} \cdot\left[\mathrm{A}_{1} \cdot \mathrm{~F}_{12}+\frac{1}{\left(\frac{1}{\mathrm{~A}_{1} \cdot \mathrm{~F}_{1 R}}+\frac{1}{\mathrm{~A}_{2} \cdot \mathrm{~F}_{2} \mathrm{R}}\right)}\right\rangle\right] \tag{13.66}
\end{equation*}
$$

Here, $Q_{12}$ is the net radiant heat transferred between surfaces 1 and 2 . Similar expressions can be written for heat transfer between surfaces 2 and $3\left(=\mathrm{Q}_{23}\right)$ and the heat transfer between surfaces 1 and $3\left(=\mathrm{Q}_{13}\right)$.

## 2. Two gray surfaces surrounded by a third re-radiating surface:



Figure 13.34 Two gray surfaces surrounded by a re-radiating surface

Here, we have:

$$
\mathrm{Q}_{3}=0
$$

i.e.

$$
\frac{\mathrm{E}_{\mathrm{b} 3}-\mathrm{J}_{3}}{\left(\frac{1-\varepsilon 3}{\mathrm{~A}_{3} \cdot \varepsilon 3}\right.}=0
$$

i.e.

$$
\mathrm{E}_{\mathrm{b} 3}=\mathrm{J}_{3}
$$

i.e. once the radiosity of the re-radiating surface is known, its temperature can easily be calculated, since $E_{b 3}=\sigma \cdot T_{3}{ }^{4}$. Further, note that $T_{3}$ is independent of the emissivity of surface 3.

Expression for heat flow rate is:

$$
\mathrm{Q}_{1}=\mathrm{Q}_{2}=\frac{\mathrm{E}_{\mathrm{b} 1}-\mathrm{E}_{\mathrm{b} 2}}{\mathrm{R}_{\mathrm{tot}}}
$$

where, $R_{\text {tot }}$ is the total resistance, given by:

$$
\mathrm{R}_{\mathrm{tot}}=\mathrm{R}_{1}+\left[\frac{1}{\frac{1}{\mathrm{R}_{12}}+\frac{1}{\left(\mathrm{R}_{13}+\mathrm{R}_{23}\right)}}\right]+\mathrm{R}_{2}
$$

i.e.

$$
\begin{equation*}
\mathrm{R}_{\mathrm{tot}}=\left(\frac{1-\varepsilon_{1}}{\mathrm{~A}_{1} \cdot \varepsilon_{1}}\right)+\frac{1}{\left[\mathrm{~A}_{1} \cdot \mathrm{~F}_{12}+\frac{1}{\left(\frac{1}{\mathrm{~A}_{1} \cdot \mathrm{~F}_{13}}+\frac{1}{\mathrm{~A}_{2} \cdot \mathrm{~F}_{23}}\right)}\right]}+\left(\frac{1-\varepsilon_{2}}{\mathrm{~A}_{2} \varepsilon_{2}}\right) \tag{13.67}
\end{equation*}
$$

## Radiation heat exchange in four zone enclosures:

a) When all the four surfaces are black: Remembering the principles already explained, if the radiation network for an enclosure comprising of four black surfaces is drawn, it will look as shown in Fig. (13.35), below:


Fig 13.35 Radiation network for an enclosure of four black surfaces

## The Wake

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Expression for net radiant heat flow rate from surface 1 is:

$$
\mathrm{Q}_{1}=\frac{\mathrm{J}_{1}-\mathrm{J}_{2}}{\mathrm{R}_{12}}+\frac{\mathrm{J}_{1}-\mathrm{J}_{3}}{\mathrm{R}_{13}}+\frac{\mathrm{J}_{1}-\mathrm{J}_{4}}{\mathrm{R}_{14}}
$$

i.e.

$$
Q_{1}=\frac{E_{b 1}-E_{b 2}}{R_{12}}+\frac{E_{b 1}-E_{b 3}}{R_{13}}+\frac{E_{b 1}-E_{b 4}}{R_{14}} \quad \ldots \text { since } E_{b 1}=J_{1} \text { ec. }
$$

i.e.

$$
\begin{equation*}
\mathrm{Q}_{1}=\mathrm{A}_{1} \cdot \mathrm{~F}_{12} \cdot\left(\mathrm{E}_{\mathrm{b} 1}-\mathrm{E}_{\mathrm{b} 2}\right)+\mathrm{A}_{1} \cdot \mathrm{~F}_{13} \cdot\left(\mathrm{E}_{\mathrm{b} 1}-\mathrm{E}_{\mathrm{b} 3}\right)+\mathrm{A}_{1} \cdot \mathrm{~F}_{14} \cdot\left(\mathrm{E}_{\mathrm{b} 1}-\mathrm{E}_{\mathrm{b} 4}\right) \tag{13.68}
\end{equation*}
$$

Similar expressions can be written for the net heat flow from other three surfaces.
b) When all the four surfaces are gray: Now, for each surface, a surface resistance has also to be included, and the radiation network for this system will be as shown in Fig. (13.36):


$$
\begin{array}{lll}
\mathrm{R}_{1}=\left(1-\epsilon_{1}\right) /\left(\mathrm{A}_{1} \cdot \epsilon_{1}\right) & \mathrm{R}_{12}=1 /\left(\mathrm{A}_{1} \cdot \mathrm{~F}_{12}\right) & \mathrm{R}_{23}=1 /\left(\mathrm{A}_{2} \cdot \mathrm{~F}_{23}\right) \\
\mathrm{R}_{2}=\left(1-\epsilon_{2}\right) /\left(\mathrm{A}_{2} \cdot \epsilon_{2}\right) & \mathrm{R}_{13}=1 /\left(\mathrm{A}_{1} \cdot \mathrm{~F}_{13}\right) & \mathrm{R}_{24}=1 /\left(\mathrm{A}_{2} \cdot \mathrm{~F}_{24}\right) \\
\mathrm{R}_{3}=\left(1-\epsilon_{3}\right) /\left(\mathrm{A}_{3} \cdot \epsilon_{3}\right) & \mathrm{R}_{14}=1 /\left(\mathrm{A}_{1} \cdot \mathrm{~F}_{14}\right) & \mathrm{R}_{43}=1 /\left(\mathrm{A}_{4} \cdot \mathrm{~F}_{43}\right) \\
\mathrm{R}_{4}=\left(1-\epsilon_{4}\right) /\left(\mathrm{A}_{4} \cdot \epsilon_{4}\right) & &
\end{array}
$$

Fig 13.36 Radiation network for an enclosure of four gray surfaces

Expression for net radiant heat flow rate from surface 1 is:

$$
Q_{1}=\frac{E_{b 1}-J_{1}}{R_{1}}=\frac{J_{1}-J_{2}}{R_{12}}+\frac{\mathrm{J}_{1}-\mathrm{J}_{3}}{\mathrm{R}_{13}}+\frac{\mathrm{J}_{1}-\mathrm{J}_{4}}{\mathrm{R}_{14}}
$$

i.e.

$$
\begin{equation*}
\mathrm{Q}_{1}=\frac{\mathrm{E}_{\mathrm{b} 1}-\mathrm{J}_{1}}{\left(\frac{1-\varepsilon}{1-\mathrm{A}_{1}}\right.}=_{1} \mathrm{~A}_{1} \cdot \mathrm{~F}_{12} \cdot\left(\mathrm{~J}_{1}-\mathrm{J}_{2}\right)+\mathrm{A}_{1} \cdot \mathrm{~F}_{13} \cdot\left(\mathrm{~J}_{1}-\mathrm{J}_{3}\right)+\mathrm{A}_{1} \cdot \mathrm{~F}_{14} \cdot\left(\mathrm{~J}_{1}-\mathrm{J}_{4}\right) \tag{13.69}
\end{equation*}
$$

Similar expressions can be written for the net heat flow from other three surfaces.

## Radiation shielding:

For two infinite, parallel plates 1 and 2, radiation shield 3, be placed between the plates:


Fig.(a)


Fig.(b)
Fig 13.37 Radiation shield between two parallel plates, and associated radiation network

When there is no shield, the radiation heat transfer between plates 1 and 2 is already shown to be:

$$
\begin{equation*}
\mathrm{Q}_{12}=\frac{\mathrm{A} \cdot \sigma \cdot\left(\mathrm{~T}_{1}^{4}-\mathrm{T}_{2}^{4}\right)}{\frac{1}{\varepsilon_{1}}+\frac{1}{\varepsilon_{2}}-1} \text {...for infinitely large parallel plates.. } \tag{13.59}
\end{equation*}
$$

With one shield placed between plates 1 and 2, the radiation network will be as shown in Fig. (b) above. Note that now all the relevant resistances are in series. Net heat transfer between plates 1 and 2 is given as:
$Q_{12 \text {-one shield }}=\left(E_{b 1}-E_{b 2}\right) / R_{\text {tot }}$ where $R_{\text {tot }}$ is the total resistance.
i.e.

$$
\mathrm{Q}_{12 \_ \text {one_shield }}=\frac{\mathrm{E}_{\mathrm{b} 1}-\mathrm{E}_{\mathrm{b} 2}}{\frac{1-\varepsilon 1}{\mathrm{~A}_{1} \cdot \varepsilon} 1}+\frac{1}{\mathrm{~A}_{1} \cdot \mathrm{~F}_{13}}+\frac{1-\varepsilon_{3} 1}{\mathrm{~A}_{3} \cdot \varepsilon_{3} 1}+\frac{1-\varepsilon_{3} 2}{\mathrm{~A}_{3} \cdot \varepsilon_{3} 2}+\frac{1}{\mathrm{~A}_{3} \cdot \mathrm{~F}_{32}}+\frac{1-\varepsilon_{2}}{\mathrm{~A}_{2} \cdot \varepsilon_{2}}
$$

....for two gray surfaces with one radiation shield placed in between.....(13.70)

Now, for two large parallel plates, we note:

$$
\mathrm{F}_{13}=\mathrm{F}_{32}=1 \quad \text { and, } \quad \mathrm{A}_{1}=\mathrm{A}_{2}=\mathrm{A}_{3}=\mathrm{A}
$$

Then, eqn. (13.70) simplifies to:

$$
\begin{equation*}
\mathrm{Q}_{12 \_ \text {one_shield }}=\frac{\mathrm{A} \cdot \sigma \cdot\left(\mathrm{~T}_{1}^{4}-\mathrm{T}_{2}^{4}\right)}{\left(\frac{1}{\varepsilon_{1}}+\frac{1}{\varepsilon_{2}}-1\right)+\left(\frac{1}{\varepsilon_{3 \_1}}+\frac{1}{\varepsilon_{3 \_2}}-1\right)} \tag{13.71}
\end{equation*}
$$

Note that as compared to eqn. (13.59) for the case of no-shield, we have, with one shield, an additional term appearing in the denominator of eqn. (13.71). Therefore, if there are N radiation shields, we have, for net radiation heat transfer:

$$
\begin{equation*}
\mathrm{Q}_{12_{\mathrm{N} \_ \text {shields }}}=\frac{\mathrm{A} \cdot \sigma \cdot\left(\mathrm{~T}_{1}{ }^{4}-\mathrm{T}_{2}{ }^{4}\right)}{\left(\frac{1}{\varepsilon_{1}}+\frac{1}{\varepsilon_{2}}-1\right)+\left(\frac{1}{\varepsilon_{3} 1}+\frac{1}{\varepsilon_{3 \_2}}-1\right)+\ldots . .+\left(\frac{1}{\varepsilon_{\mathrm{N}} 1}+\frac{1}{\varepsilon_{\mathrm{N} \_2}}-1\right)} \tag{13.72}
\end{equation*}
$$

If emissivities of all surfaces are equal, eqn. (13.72) becomes:

$$
\begin{equation*}
\mathrm{Q}_{12_{\mathrm{N} \_ \text {shields }}}=\frac{\mathrm{A} \cdot \sigma \cdot\left(\mathrm{~T}_{1}{ }^{4}-\mathrm{T}_{2}{ }^{4}\right)}{(\mathrm{N}+1) \cdot\left(\frac{1}{\varepsilon}+\frac{1}{\varepsilon}-1\right)}=\frac{1}{(\mathrm{~N}+1)} \cdot\left(\mathrm{Q}_{12_{\text {no_shield } /}}\right. \tag{13.73}
\end{equation*}
$$

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Note this important result, which implies that, when all emissivities are equal, presence of one radiation shield reduces the radiation heat transfer between the two surfaces to one-half, two radiation shields reduce the heat transfer to one-third, 9 radiation shields reduce the heat transfer to one-tenth etc.

For a more practical case of the two surfaces having emissivities of $\varepsilon 1$ and $\varepsilon 2$, and all shields having the same emissivity of $\varepsilon s$, eqn. (13.72) becomes:

$$
\begin{equation*}
\mathrm{Q}_{12}^{\mathrm{N} \_ \text {shields }}=\frac{\mathrm{A} \cdot \sigma \cdot\left(\mathrm{~T}_{1}^{4}-\mathrm{T}_{2}^{4}\right)}{\left(\frac{1}{\varepsilon_{1}}+\frac{1}{\varepsilon_{2}}-1\right)+\mathrm{N} \cdot\left(\frac{2}{\varepsilon_{\mathrm{s}}}-1\right)} \tag{13.74}
\end{equation*}
$$

## To determine the equilibrium temperature of the radiation shield:

Once $Q_{12}$ is determined from eqn. (13.71), the temperature of the shield is easily found out by applying the condition that in steady state:

$$
\begin{equation*}
\mathrm{Q}_{12}=\mathrm{Q}_{13}=\mathrm{Q}_{32} \tag{13.75}
\end{equation*}
$$

We can use either of the conditions: $Q_{12}=Q_{13}$ or $Q_{12}=Q_{32}$.
$Q_{13}$ or $Q_{32}$ is determined by applying eqn. (13.59); i.e. we get:

$$
\begin{equation*}
\mathrm{Q}_{12}=\mathrm{Q}_{13}=\frac{\mathrm{A} \cdot \sigma \cdot\left(\mathrm{~T}_{1}{ }^{4}-\mathrm{T}_{3}^{4}\right)}{\frac{1}{\varepsilon_{1}}+\frac{1}{\varepsilon_{3}}-1} \tag{13.76,a}
\end{equation*}
$$

or,

$$
\begin{equation*}
\mathrm{Q}_{12}=\mathrm{Q}_{32}=\frac{\mathrm{A} \cdot \sigma \cdot\left(\mathrm{~T}_{3}^{4}-\mathrm{T}_{2}^{4}\right)}{\frac{1}{\varepsilon_{3}}+\frac{1}{\varepsilon_{2}}-1} \tag{13.76,b}
\end{equation*}
$$

In both the above eqns. $\mathrm{T}_{3}$ is the only unknown, which can easily be determined.

## For a cylindrical radiation shield placed in between two, long concentric cylinders:

The radiation heat transfer between two long, concentric cylinders is:

$$
\begin{equation*}
\mathrm{Q}_{12}=\frac{\mathrm{A}_{1} \cdot \sigma \cdot\left(\mathrm{~T}_{1}^{4}-\mathrm{T}_{2}^{4}\right)}{\frac{1}{\varepsilon_{1}}+\left(\frac{\mathrm{A}_{1}}{\mathrm{~A}_{2}}\right) \cdot\left(\frac{1}{\varepsilon_{2}}-1\right)} \quad \text {...for infinitely long concentric cylinders.. } \tag{13.60}
\end{equation*}
$$

where

$$
\frac{\mathrm{A}_{1}}{\mathrm{~A}_{2}}=\frac{\mathrm{r}_{1}}{\mathrm{r}_{2}}
$$

Now, let a cylindrical radiation shield, 3, be placed in between the inner cylinder (1) and the outer cylinder (2), as shown in Fig.(13.38).


Fig.(a)


Fig.(b)
Fig 13.38 Radiation shield between two concentric cylinders, and associated radiation network

The radiation network for this system is shown in Fig. $(13.38, b)$ and it is exactly the same as shown in Fig. (13.37,b).

And, the radiation heat transfer between cylinders 1 and 2 , when the shield is present, is given by:

$$
\left.\mathrm{Q}_{12 \_ \text {one_shield }}=\frac{\mathrm{E}_{\mathrm{b} 1}-\mathrm{E}_{\mathrm{b} 2}}{\frac{1-\varepsilon_{1}}{\mathrm{~A}_{1} \cdot \varepsilon} 1}+\frac{1}{\mathrm{~A}_{1} \cdot \mathrm{~F}_{13}}+\frac{1-\varepsilon_{3} 1}{\mathrm{~A}_{3} \cdot \varepsilon_{3} 1_{-}}+\frac{1-\varepsilon_{3} 2}{\mathrm{~A}_{3} \cdot \varepsilon_{3} 2}+\frac{1}{\mathrm{~A}_{3} \cdot \mathrm{~F}_{32}}+\frac{1-\varepsilon_{2}}{\mathrm{~A}_{2} \cdot \varepsilon_{2}}\right)
$$

....for two gray surfaces with one radiation shield placed in between.....(13.70)

Now, for the cylindrical system, we have:

$$
\begin{aligned}
& \mathrm{F}_{13}=\mathrm{F}_{32}=1 \\
& \mathrm{~A}_{1}=2 \cdot \pi \cdot \mathrm{r}_{1} \cdot \mathrm{~L} \\
& \mathrm{~A}_{2}=2 \cdot \pi \cdot \mathrm{r}_{2} \cdot \mathrm{~L}
\end{aligned}
$$

and,

$$
\mathrm{A}_{3}=2 \cdot \pi \cdot \mathrm{r}_{3} \cdot \mathrm{~L}
$$

Then, eqn. (13.70) reduces to:

$$
\mathrm{Q}_{12} \text { one_shield }=\frac{\mathrm{A}_{1} \cdot \sigma \cdot\left(\mathrm{~T}_{1}{ }^{4}-\mathrm{T}_{2}^{4}\right)}{\frac{1}{\varepsilon_{1}}+\left(\frac{\mathrm{A}_{1}}{\mathrm{~A}_{2}}\right) \cdot\left(\frac{1}{\varepsilon_{2}}-1\right)+\left(\frac{\mathrm{A}_{1}}{\mathrm{~A}_{3}}\right) \cdot\left(\frac{1}{\varepsilon_{3 \_1}}+\frac{1}{\varepsilon_{3-2}}-1\right)}
$$

...for concentric cylinders with one radiation shield...(13.77)

In eqn. (13.77), we have: $\left(A_{1} / A_{2}\right)=\left(r_{1} / r_{2}\right)$, and $\left(A_{1} / A_{3}\right)=\left(r_{1} / r_{3}\right)$.

Note that as compared to the relation for two concentric cylinders with no shield (i.e. eqn. 13.60), an additional term appears in the denominator of eqn. (13.77) (i.e. the third term) when one radiation shield is introduced; if there is a second radiation shield, say (4), then one more similar term will have to be added in the denominator to take care of the resistance of that shield.

In this case too, the equilibrium temperature of the shield is determined by applying the principle that, in steady state,

$$
\mathrm{Q}_{12}=\mathrm{Q}_{13}=\mathrm{Q}_{32}
$$

## For a spherical radiation shield placed in between two concentric spheres:

This case is also represented by Fig. ( $13.38, \mathrm{a}$ ), where inner sphere 1 is enclosed by an outer sphere 2, and a radiation shield 3, is placed in between. The radiation network for this system is shown in Fig. (13.38,b).

When there is no radiation shield, radiation heat transfer between surfaces 1 and 2 is given by eqn. (13.61), viz.

$$
\begin{equation*}
\mathrm{Q}_{12}=\frac{\mathrm{A}_{1} \cdot \sigma \cdot\left(\mathrm{~T}_{1}{ }^{4}-\mathrm{T}_{2}{ }^{4}\right)}{\frac{1}{\varepsilon_{1}}+\left(\frac{\mathrm{A}_{1}}{\mathrm{~A}_{2}}\right) \cdot\left(\frac{1}{\varepsilon_{2}}-1\right)} \tag{13.61}
\end{equation*}
$$

where

$$
\frac{\mathrm{A}_{1}}{\mathrm{~A}_{2}}=\left(\frac{\mathrm{r}_{1}}{\mathrm{r}_{2}}\right)^{2}
$$



Again, when the radiation shield is present, the general relation for radiation heat transfer between surfaces 1 and 2 is eqn. (13.70). Remembering that for concentric spheres,

$$
\begin{aligned}
& \mathrm{F}_{13}=\mathrm{F}_{32}=1 \\
& \mathrm{~A}_{1}=4 \cdot \pi \cdot \mathrm{r}_{1}^{2} \\
& \mathrm{~A}_{2}=4 \cdot \pi \cdot \mathrm{r}_{2}^{2}
\end{aligned}
$$

and,

$$
\mathrm{A}_{3}=4 \cdot \pi \cdot \mathrm{r}_{3}^{2}
$$

Relation for radiant heat transfer between surfaces 1 and 2 , is exactly as eqn. (13.77), i.e.

$$
\mathrm{Q}_{12} \text { one_shield }=\frac{\mathrm{A}_{1} \cdot \sigma \cdot\left(\mathrm{~T}_{1}^{4}-\mathrm{T}_{2}^{4}\right)}{\frac{1}{\varepsilon_{1}}+\left(\frac{\mathrm{A}_{1}}{\mathrm{~A}_{2}}\right) \cdot\left(\frac{1}{\varepsilon_{2}}-1\right)+\left(\frac{\mathrm{A}_{1}}{\mathrm{~A}_{3}}\right) \cdot\left(\frac{1}{\varepsilon_{3 \_1}}+\frac{1}{\varepsilon_{3-2}}-1\right)}
$$

...for concentric spheres with one radiation shield...(13.78)

In eqn. (13.78), we have:

$$
\frac{\mathrm{A}_{1}}{\mathrm{~A}_{2}}=\left(\frac{\mathrm{r}_{1}}{\mathrm{r}_{2}}\right)^{2} \quad \text { and, } \quad \frac{\mathrm{A}_{1}}{\mathrm{~A}_{3}}=\left(\frac{\mathrm{r}_{1}}{\mathrm{r}_{3}}\right)^{2}
$$

In this case also, equilibrium temperature of the shield is determined by applying the principle that, in steady state,

$$
\mathrm{Q}_{12}=\mathrm{Q}_{13}=\mathrm{Q}_{32}
$$

## Radiation error in temperature measurement:


(a) Thermometer without radiation shield

(b) Thermometer with radiation shield

Fig 13.39 Radiation shielding of thermometers

Let the reading shown by the thermometer be $\mathrm{T}_{c}$. This reading, however, does not represent the true temperature of the fluid $\mathrm{T}_{\mathrm{f}}$, since the thermometer bulb will lose heat by radiation to the walls of the channel which are at a lower temperature $\mathrm{T}_{\mathrm{w}}$ (which is usually the case).

The temperature $T_{c}$ shown by the thermometer will be some value in between $T_{f}$ and $T_{w}$.

We wish to find out the true temperature of the fluid $\mathrm{T}_{\mathrm{f}}$ by knowing the thermometer reading $\mathrm{T}_{\mathrm{c}}$.

Making an energy balance on the thermometer bulb, in steady state, we have:

## Without radiation shield:

$\mathrm{q}_{\text {conv }}$ to the bulb $=\mathrm{q}_{\mathrm{rad}}$ from the bulb
i.e. $\quad h \cdot A_{c} \cdot\left(T_{f}{ }^{-} T_{c}\right)=\varepsilon c_{c} \cdot A_{c} \cdot \sigma \cdot\left(T_{c}{ }^{4}-T_{w}{ }^{4}\right)$
i.e. $\quad \mathrm{T}_{\mathrm{f}}=\mathrm{T}_{\mathrm{c}}+\frac{\varepsilon_{\mathrm{c}} \cdot \sigma \cdot\left(\mathrm{T}_{\mathrm{c}}{ }^{4}-\mathrm{T}_{\mathrm{w}}{ }^{4}\right)}{\mathrm{h}}$
where $A_{c}=$ surface area of thermometer bulb,
$\varepsilon_{c}=$ emissivity of thermometer bulb surface

Eqn. (13.79) gives the true temperature of the fluid $\mathrm{T}_{\mathrm{f}}$

Second term on the RHS of eqn. (13.79) represents the error in temperature measurement due to radiation effect.

## Energy balance on the thermometer bulb:

Heat transferred to the bulb from the fluid by convection $=$
Heat transferred from the bulb to the shield by radiation

$$
\text { i.e. } \quad h \cdot A_{c} \cdot\left(T_{f}-T_{c}\right)=\frac{\sigma \cdot\left(T_{c}^{4}-T_{s}^{4}\right)}{\left(\frac{1-\varepsilon_{c}}{A_{c} \cdot \varepsilon_{c}}\right)+\frac{1}{A_{c} \cdot F_{c s}}+\left(\frac{1-\varepsilon_{s}}{A_{s} \cdot \varepsilon_{s}}\right)}
$$

In eqn. (13.80), $\mathrm{F}_{\mathrm{cs}}=$ view factor of thermometer bulb w.r.t the shield and is, generally equal to 1.

## Making an energy balance on the shield:

heat transferred to shield from the fluid by convection + heat transferred to shield from bulb by radiation $=$ heat transferred from shield to walls by radiation

$$
\begin{equation*}
\text { i.e. } 2 \cdot \mathrm{~A}_{\mathrm{s}} \cdot \mathrm{~h} \cdot\left(\mathrm{~T}_{\mathrm{f}}-\mathrm{T}_{\mathrm{s}}\right)+\frac{\sigma \cdot\left(\mathrm{T}_{\mathrm{c}}^{4}-\mathrm{T}_{\mathrm{s}}^{4}\right)}{\left(\frac{1-\varepsilon_{\mathrm{c}}}{\mathrm{~A}_{\mathrm{c}} \cdot \varepsilon_{\mathrm{c}}}\right)+\frac{1}{\mathrm{~A}_{\mathrm{c} \cdot} \cdot \mathrm{~F}_{\mathrm{cs}}}+\left(\frac{1-\varepsilon_{\mathrm{s}}}{\mathrm{~A}_{\mathrm{s}} \cdot \varepsilon_{\mathrm{s}}}\right)}=\varepsilon_{\mathrm{s}} \cdot \mathrm{~A}_{\mathrm{s}} \cdot \sigma^{\cdot} \cdot\left(\mathrm{T}_{\mathrm{s}}^{4}-\mathrm{T}_{\mathrm{w}}^{4}\right) \tag{13.81}
\end{equation*}
$$

$$
\text { where, } \begin{aligned}
\mathrm{A}_{\mathrm{S}} & =\text { area of shield on one side } \\
\varepsilon_{\mathrm{S}} & =\text { emissivitty of shield surface } \\
\mathrm{A}_{\mathrm{c}} & =\text { area of bulb surface } \\
\varepsilon_{\mathrm{c}} & =\text { emissivity of bulb surface } \\
\mathrm{F}_{\mathrm{cs}} & =\text { view factor of bulb w.r.t. shield }
\end{aligned}
$$

In the first term of the above eqn. factor 2 appears since convective heat transfer to the shield occurs on both surfaces of the shield. Also, in writing the RHS, the inherent assumption is that:

$$
\mathrm{F}_{\mathrm{SW}}=1 \quad \ldots \text { view factor between the shield and the walls }
$$

and,

$$
\frac{\mathrm{A}_{\mathrm{s}}}{\mathrm{~A}_{\mathrm{w}}}=0 \quad \begin{aligned}
& \text {..i.e. surface area of shield is negligible compared to the area of the channel } \\
& \text { walls }
\end{aligned}
$$

Solving eqns. (13.80) and (13.81) simultaneously, we obtain the shield temperature $\mathrm{T}_{\mathrm{s}}$ and the thermometer reading $\mathrm{T}_{\mathrm{c}^{\prime}}$ (if $\mathrm{T}_{\mathrm{f}}$ is known), or $\mathrm{T}_{\mathrm{f}}$ (if $\mathrm{T}_{\mathrm{c}}$ is known).



## 5A Laws of Radiation and Radiation properties:

## Prob.5A.1: Draw a plot of Planck's Law for spectral distribution:

## Mathcad Solution:

## Planck's Law:

$$
\begin{equation*}
E_{\mathrm{b} \lambda}=\frac{\mathrm{C}_{1} \cdot \lambda^{-5}}{\exp \left(\frac{C_{2}}{\lambda \cdot T}\right)-1} \quad \ldots \mathrm{~W} /\left(m^{2} \cdot \mu \mathrm{~m}\right) \tag{13.7}
\end{equation*}
$$

where:
We have, costants $\mathrm{C}_{1}$ and $\mathrm{C}_{2}$ in Planck's Law:

$$
\begin{aligned}
& \mathrm{C}_{1}:=3.742 \cdot 10^{8} \quad \mathrm{~W} \cdot \mu \mathrm{~m}^{4} / \mathrm{m}^{2} \\
& \mathrm{C}_{2}=1.4387 \cdot 10^{4} \quad \mu \mathrm{~m} . \mathrm{K}
\end{aligned}
$$

To plot the Planck's disribution for a black body, using Mathcad, first, define $\mathrm{E}_{\mathrm{b} \text {, }}$ as a function of T and h as follows:

$$
\mathrm{E}_{\mathrm{b} \lambda}(\lambda, \mathrm{~T}):=\frac{\mathrm{C}_{1} \cdot \lambda^{-5}}{\exp \left(\frac{\mathrm{C}_{2}}{\lambda \cdot \mathrm{~T}}\right)-1} \quad \ldots . \mathrm{E}_{\mathrm{b} \lambda} \text { as a function of } \lambda \text { and } \mathrm{T}
$$

To draw the plot:

Define a range variable $\lambda$ varying from $0.01 \mu \mathrm{~m}$ o $1000 \mu \mathrm{~m}$ :

$$
\lambda:=0.01,0.02 \ldots 1000 \quad \text {.... define the range variable }
$$

Now, select the $x$-y plot from the graph palette; on $x$-axis place holder, type $\lambda$ and on the $y$-axis place holder, fill in $\mathrm{E}_{\mathrm{b} \lambda}(2,100), \mathrm{E}_{\mathrm{b} \lambda 2}(2,500), \mathrm{E}_{\mathrm{b} \lambda}(2,1000)$, and $\mathrm{E}_{\mathrm{b} \lambda}(\lambda, 5800)$. Click anywhere outside the graph region, and the curves appear immediately:


See the ease with which this plot is drawn with Mathcad.

Prob.5A.2. Radiation strikes an object with transmissivity of 0.03 and reflectivity of 0.5 . The absorbed flux is indirectly measured to be $94 \mathrm{~W} / \mathrm{m} 2$. Compute the incident flux. (M.U.)

## Mathcad Solution:

## Data:

$$
\begin{array}{ll}
\tau:=0.03 & \text { transmissivity } \\
\rho:=0.5 & \text { reflectivity } \\
Q_{a}:=94 & \mathrm{~W} / \mathrm{m}^{2} \ldots \text { absorbed flux }
\end{array}
$$

Let $\mathrm{G}=$ incident flux

Now, we have: $\quad \alpha+\tau+\rho=1$ where, $\alpha=Q_{g} / G$

Therefore,

$$
\alpha=\frac{Q_{a}}{G}=1-\tau-p
$$

i.e. $\quad G:=\frac{Q_{a}}{1-\tau-\rho}$
i.e. $\quad \mathrm{G}=200 \quad \mathrm{~W} / \mathbf{m}^{2}$....Incident flux....Ans.

Prob.5A.3. A hole of area $\mathrm{dA}=2 \mathrm{~cm} 2$ is opened on the surface of a large spherical cavity whose inside is maintained at 1000 K . Calculate: (a) the radiation energy streaming through the hole in all directions into space, (b) the radiation energy streaming per unit solid angle in a direction making a 60 deg . angle with the normal to the surface of the opening. (M.U.)


## Mathcad Solution:



Data:
$\mathrm{dA}:=2 \cdot 10^{-4} \quad \mathrm{~m}^{2} \ldots$ area of the hole
$\mathrm{T}:=1000 \quad \mathrm{~K} . .$. temperature
$\sigma:=5.67 \cdot 10^{-8} \mathrm{~W} /\left(\mathrm{m}^{2}\right.$.K4) ...Stefan-Boltzann constant
$\theta=60 \mathrm{deg} . \quad . .$. angle made by the desired direction, with the normal to the surface
i.e. $\quad \theta:=60 \cdot \frac{\pi}{180} \quad$..radians
(a) Radiation streaming out in all directions:

Since the spherical cavity can be considered as a black body, energy streaming out is given by Stefan-Boltzmann law:
i.e. $\quad Q:=d A \cdot \sigma \cdot T^{4}$
or, $\quad Q=11.34 \quad$ W...Radiaion energy streaming through the hole...Ans.
(b) Radiation streaming out through unit solid angle, in a direction making 60 deg. with normal:

Now, we have the relation: $\mathrm{E}_{\mathrm{b}}=\pi . \mathrm{I}_{\mathrm{b}}$, where $\mathrm{E}_{\mathrm{b}}=$ Emissive power, and $\mathrm{I}_{\mathrm{b}}=$ Intensity of radiation.

$$
\begin{aligned}
\text { Therefore, } & I_{b}=\frac{E_{b}}{\pi} \\
\text { But, } & E_{b}=\sigma \cdot T^{4} \quad \mathrm{~W} / \mathrm{m}^{2} \ldots \text { for a black body, by Stefan-Boltzmann Law } \\
\text { i.e. } & I_{b}:=\frac{\sigma \cdot \mathrm{T}^{4}}{\pi} \mathrm{~W} /\left(\mathrm{m}^{2} . \text { sr) } \ldots\right. \text { Intensity of radiation } \\
\text { i.e. } & I_{b}=1.805 \times 10^{4} \mathrm{~W} /\left(\mathrm{m}^{2} . \mathrm{sr}\right)
\end{aligned}
$$

And, radiation streaming out in that direction:
$\mathrm{Q}:=\mathrm{I}_{\mathrm{b}} \cdot \mathrm{dA} \cdot \cos (\theta) \mathrm{W}$.... where $\theta$ is in radians (Note: while using Mathcad, $\theta$ must be in radians while calculating $\cos (\theta)$.)
i.e. $Q=1.805 \quad$ W...radiaion through a solid angle of unity, in a direction of 60 deg. with normal to the surface...Ans.

Prob.5A.4. It is observed that intensity of radiation is maximum in case of solar radiation at a wavelength of 0.49 microns. Assuming the Sun as a black body, estimate its surface temperature and emissive power. Wein displacement constant $=0.289 \times 10-2 \mathrm{~m} . \mathrm{K}$. (M.U.)

## Mathcad Solution:

Data:

$$
\begin{array}{rl}
\lambda_{\max } & :=0.49 \quad \text { microns } \ldots \text { wave length of max. radiation } \\
\sigma:=5.67 \cdot 10^{-8} & \mathrm{~W} /\left(\mathrm{m}^{2} . \mathrm{K}^{4}\right) \ldots \text { Stefan-Boltzmann const. }
\end{array}
$$

By Wein's displacement law, we have:

$$
\lambda_{\max } \cdot \mathrm{T}=2890 \text { microns. } \mathrm{K}
$$

Therefore,

$$
\mathrm{T}:=\frac{2890}{\lambda_{\max }}
$$

$$
\text { i.e. } \quad T=5.898 \times 10^{3} \quad K \text {...surface temp. of Sun...Ans. }
$$

Heat flux at the surface, $\mathrm{E}_{\mathrm{b}}$ :
Sun an be considered as a black body; then, from Stefan-Boltzmann Law:

$$
\begin{aligned}
E_{b} & :=\sigma \cdot T^{4} \\
\text { i.e. } \quad E_{b} & =6.861 \times 10^{7} \quad W / m^{2} \ldots \text { heat flux at the surface of Sun...Ans. }
\end{aligned}
$$

Prob.5A.5. The temperature of a body of area 0.1 m 2 is 900 K . Calculate the total rate of energy emission, intensity of normal radiation in $\mathrm{W} /(\mathrm{m} 2 . \mathrm{sr})$, maximum monochromatic emissive power, and wavelength at which it occurs. (M.U.)

## Mathcad Solution:

Data:

$$
\begin{aligned}
\mathrm{T} & :=900 \mathrm{~K} \text {...temperature } \\
\mathrm{A} & :=0.1 \quad \mathrm{~m}^{2} \ldots \text { area } \\
\sigma & :=5.67 \cdot 10^{-8} \mathrm{~W} /\left(\mathrm{m}^{2} \cdot \mathrm{~K}^{4}\right) \text { Stefan-Boltzmann const. }
\end{aligned}
$$

## Total rate of energy emission:

$$
\begin{aligned}
E_{b} & :=\sigma \cdot T^{4} \quad \text { from Stefan-Boltzmann Law } \\
\text { i.e. } E_{b} & =3.72 \times 10^{4} \quad \mathrm{~W} / \mathrm{m}^{2} \ldots \text { Total emissive power }
\end{aligned}
$$

Therefore,

$$
\begin{aligned}
Q & :=E_{b} \cdot A \\
\text { i.e. } Q & =3.72 \times 10^{3} \quad \text { W...Total energy emission from surface...Ans. }
\end{aligned}
$$

Intensity of normal radiation:
(Note that for a black (diffuse) surface, intensity is the same in all directions.)

$$
\begin{aligned}
\quad I & =\frac{E_{b}}{\pi} \\
\text { i.e. } \quad I & =1.184 \times 10^{4} \quad W /\left(m^{2} . s r\right) \ldots . . \text { Ans. }
\end{aligned}
$$

Wave length of max. monochromatic emissive power:
From Wein's displacement law, we have:

$$
\lambda_{\mathrm{m}} \cdot \mathrm{~T}=2898 \quad \mu \mathrm{~m} \cdot \mathrm{~K}
$$

Therefore,

$$
\lambda_{\mathrm{m}}:=\frac{2898}{\mathrm{~T}} \quad \mu \mathrm{~m}
$$

i.e. $\quad \lambda_{\mathrm{m}}=3.22 \quad \mu \mathrm{~m} . .$. wave length for max. monochromatic emissive power at 900 K...Ans

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Max. monochromatic emissive power:
We use Planck's law, with values of constants, C1 and C2:

$$
\left.\mathrm{C} 1:=3.742 \cdot 10^{8} \quad \mathrm{~W} \cdot \mu \mathrm{~m}^{4} / \mathrm{m}^{2}\right)
$$

and, $\quad C 2:=1.4387 \cdot 10^{4} \quad \mu \mathrm{~m} . \mathrm{K}$

$$
\mathrm{E}_{\mathrm{b} \lambda \max }:=\frac{\mathrm{C} 1 \cdot \lambda_{\mathrm{m}}^{-5}}{\exp \left(\frac{\mathrm{C} 2}{\lambda_{\mathrm{m}} \cdot \mathrm{~T}}\right)-1} \ldots \text { Planck's Law }
$$

i.e. $\quad E_{b\rangle \text { max }}=7.6 \times 10^{3} \quad W /\left(m^{2}\right.$.micron)...Max. monochr. emissive power....Ans.

## Alternatively:

We can directly apply eqn.(13.12):

$$
\begin{array}{ll} 
& \mathrm{E}_{\mathrm{b} / \mathrm{m}}:=1.287 \cdot 10^{-5} \cdot \mathrm{~T}^{5} \mathrm{~W} / \mathrm{m}^{3} \ldots .(13.12)  \tag{13.12}\\
\text { i.e. } & \mathrm{E}_{\mathrm{b} / \mathrm{m}}=7.6 \times 10^{9} \quad \begin{array}{l}
\mathrm{W} / \mathrm{m}^{3}=7.6 \times 10^{-3} \mathrm{~W} /\left(\mathrm{m}^{2} . \mu \mathrm{m}\right) . . \text { same as } \\
\text { obtained above...Ans. }
\end{array}
\end{array}
$$

Prob.5A.6. Write a Mathcad program to read the Radiation Functions (i.e. radiation emitted in a wave band as a fraction of Total emissive power) against $\lambda \mathrm{T}$ from the Table 13.2 and return its value at the desired $\lambda \mathrm{T}$ by interpolation:

We have

$$
\begin{gathered}
\mathrm{E}_{\mathrm{b} \lambda}(\lambda, \mathrm{~T}):=\frac{\mathrm{C}_{1} \cdot \lambda^{-5}}{\exp \left(\frac{C_{2}}{\lambda \cdot \mathrm{~T}}\right)-1} \\
\mathrm{~F}_{0_{-} \lambda}=\frac{\int_{0}^{\lambda} \mathrm{E}_{\mathrm{b} \lambda} \mathrm{~d} \lambda}{\sigma \cdot T^{4}}=\frac{\int_{0}^{\lambda \cdot \mathrm{T}} \mathrm{E}_{\mathrm{b} \lambda} \mathrm{~d}(\lambda \cdot \mathrm{~T})}{\sigma \cdot \mathrm{T}^{5}}
\end{gathered}
$$

First, write the Table 13.2 for $\mathrm{F}_{-} 0 \_\lambda$ vs $\lambda \mathrm{T}$ in two columns only, and then be set it equal to Matrix M1. First column of the Matrix is $\lambda T$ values and the second column is $F_{-} 0 \_\lambda$ values.

Then:

$$
\mathrm{F}_{0-\lambda}(\mathrm{M} 1, \lambda \mathrm{~T}):=\left\lvert\, \begin{aligned}
& (\text { return } 0) \text { if } \lambda \mathrm{T} \leq 600 \\
& (\text { return } 1) \text { if } \lambda \mathrm{T}>100000 \\
& \text { LambdaT_vector } \leftarrow \mathrm{M1}^{\langle 0\rangle} \\
& \mathrm{F}_{-} \text {vector } \leftarrow \mathrm{M1}^{\langle 1\rangle} \\
& \mathrm{F}_{-} 0{ }_{-} \downarrow \leftarrow \operatorname{linterp}\left(\text { LambdaT_vector }, \mathrm{F}_{-} \text {vector }, \lambda \mathrm{T}\right)
\end{aligned}\right.
$$

Note that $\mathrm{F} \_0 \_\lambda=0$ for $\lambda T<600$, and $\mathrm{F} \_0 \_\lambda=1$ for $\lambda \mathrm{T}>100000$ (See the Table 13.2 at the beginning of this Chapter).

$$
\begin{array}{ll}
\mathrm{Ex}: & \lambda \mathrm{T}:=12000 \\
& \mathrm{~F}_{0-\lambda}(\mathrm{M} 1, \lambda \mathrm{~T})=0.9451
\end{array}
$$

Also, if we need to get fraction between two given wave lengths $\boldsymbol{\lambda} 1$ and $\boldsymbol{\lambda} 2$ :

$$
\begin{aligned}
& \mathrm{F}_{\lambda .1 \_2}(\mathrm{M} 1, \lambda .1 \mathrm{~T}, \lambda .2 \mathrm{~T}):=\mathrm{F}_{0 \_\lambda}(\mathrm{M} 1, \lambda .2 \mathrm{~T})-\mathrm{F}_{0_{-} \lambda}(\mathrm{M} 1, \lambda .1 \mathrm{~T}) \\
& \text { Ex: } \quad \lambda .1 \mathrm{~T}:=1600 \quad \quad \lambda .2 \mathrm{~T}:=12000 \\
& \mathrm{~F}_{\lambda .1 \_\lambda .2}(\mathrm{M} 1, \lambda .1 \mathrm{~T}, \lambda .2 \mathrm{~T})=0.925
\end{aligned}
$$

## Let us use this Function in the following problem:

Prob.5A.7. Window glass transmits radiant energy in the wave length range $0.4 \mu \mathrm{~m}$ to $2.5 \mu \mathrm{~m}$. Determine the fraction of total radiant energy which is transmitted, when the source temperature is: (a) 5800 K (i.e. Sun's surface temp.), and (b) 300 K (i.e. room temp.)

## Mathcad Solution:

## Data:

$\mathrm{T} 1:=5800 \quad \mathrm{~K} \ldots$. source temperature in first case
$\mathrm{T} 2:=300 \mathrm{~K} \ldots$...source temperature in second case
$\lambda_{1}:=0.4 \quad \mu \mathrm{~m} . .$. lower end of wave length band in which glass is transparent
$\lambda_{2}:=2.5 \quad \mu \mathrm{~m} \ldots$ upper end of wave length band in which glass is transparent $\sigma:=5.67 \cdot 10^{-8} \mathrm{~W} /\left(\mathrm{m}^{2} . \mathrm{K}^{4}\right) \ldots$ Stefan-Boltzmann const.

Case (a): Source temp. T1 $=5800 \mathrm{~K}$
We use the Mathcad Functions written above, or Table 13.2 where radiation functions are tabulated against the product ( $\lambda . \mathrm{T}$ ).

We have:

$$
\begin{aligned}
& \lambda_{1} \cdot \mathrm{~T} 1=2.32 \times 10^{3} \quad \mu \mathrm{~m} . \mathrm{K} \\
& \lambda_{2} \cdot \mathrm{~T} 1=1.45 \times 10^{4} \quad \mu \mathrm{~m} . \mathrm{K}
\end{aligned}
$$



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Corresponding to these $\lambda T$ values, we get:

$$
\mathrm{F}_{0 \_\lambda 1}:=\mathrm{F}_{0 \_\lambda}\left(\mathrm{M} 1, \lambda_{1} \cdot \mathrm{~T} 1\right) \quad \text {....using the Mathcad Function written above }
$$

i.e. $\quad F_{0 \_\lambda .1}=0.1245$
and,

$$
\mathrm{F}_{0_{-} \lambda 2}:=\mathrm{F}_{0 \_} \lambda\left(\mathrm{M} 1, \lambda_{2} \cdot \mathrm{~T} 1\right) \quad \ldots . . \text { using the Mathcad Function written above }
$$

i.e. $\quad F_{0 \lambda, 2}=0.9661$

Therefore, fraction transmmitted is equal to:

$$
0.9661-0.12454=0.842
$$

Alternatively: $\quad \mathrm{F}_{\lambda_{1} \_\lambda}\left(\mathrm{M} 1, \lambda_{1} \cdot \mathrm{~T} 1, \lambda_{2} \cdot \mathrm{~T} 1\right)=0.842$
i.e. $84.2 \%$ of the energy coming from the Sun (at 5800 K ) is transmitted through the window glass.....Ans.

Case (b): Source temp. T2 $=300 \mathrm{~K}$
Again, we use the Mathcad Functions already written or Table 13.2 where radiation functions are tabulated against the product ( $\lambda . \mathrm{T}$ ).

We have:

$$
\begin{array}{ll}
\lambda_{1} \cdot \mathrm{~T} 2=120 & \mu \mathrm{~m} \cdot \mathrm{~K} \\
\lambda_{2} \cdot \mathrm{~T} 2=750 & \mu \mathrm{~m} \cdot \mathrm{~K}
\end{array}
$$

Corresponding to these $\lambda T$ values, we get:

$$
\mathrm{F}_{0 \_\lambda 1}:=\mathrm{F}_{0 \_-} \lambda\left(\mathrm{M} 1, \lambda_{1} \cdot \mathrm{~T} 2\right) \quad \text {....using the Mathcad Function written above }
$$

i.e. $\quad F_{0 \_\lambda 1}=0$
and, $\quad \mathrm{F}_{0 \_\lambda} \lambda_{2}:=\mathrm{F}_{0-\lambda}\left(\mathrm{M} 1, \lambda_{2} \cdot \mathrm{~T} 2\right) \quad$....using the Mathcad Function written above
i.e. $\quad F_{0 \_} \lambda .2=1.2 \times 10^{-5}=$ almost zero
i.e. practically no energy will be transmitted through the window glass in this wave length range, if the source temp. $=300 \mathrm{~K}$.

In other words, glass is 'opaque' to radiation at 300 K in the wave length range $0.4 \mu \mathrm{~m}$ to $2.5 \mu \mathrm{~m}$.

This is the principle of 'Greenhouse effect', wherein radiation from a high temperature source (i.e. Sun) is allowed to pass through the glass into the enclosure of the green house, while radiation at a relatively low temperature from within the enclosure, is not allowed to escape out. This, in effect, causes an increase in the temperature of the space within the enclosure.

Prob.5A.8. Spectral emissivity of a particular surface at 800 K is approximated by a step function, as follows: $\varepsilon 1=0.1$ for $\lambda=0$ to $2 \mu \mathrm{~m}, \varepsilon 2=0.5$ for $\lambda=2$ to $15 \mu \mathrm{~m}$, and $\varepsilon 3=0.8$ for $\lambda=15$ to $\infty$. Calculate: (i) the total (hemispherical) emissive power, and (ii) total hemispherical emissivity $\varepsilon$, over all wave lengths.


## Mathcad Solution:

Data:
$\mathrm{T}:=800 \mathrm{~K} \ldots$...temperature
$\varepsilon 1:=0.1 \quad$...emissiviy in wavelength range: 0 to $2 \mu \mathrm{~m}$
$\varepsilon 2:=0.5 \quad$..emissiviy in wavelength range: 2 to $15 \mu \mathrm{~m}$
$\varepsilon 3:=0.8 \quad$...emissiviy in wavelength range: $15 \mu \mathrm{~m}$ to $\infty$.
$\sigma:=5.67 \cdot 10^{-8} \mathrm{~W} /\left(\mathrm{m}^{2} . \mathrm{K}^{4}\right)$....Stefan-Boltzmann const.
$\lambda_{1}:=2 \quad \mu \mathrm{~m}$
$\lambda_{2}:=15 \quad \mu \mathrm{~m}$
$\lambda_{3}:=\infty$

Planck's Law gives spectral emissive power of a black body.
For a non-gray surface considered in this problem, we can write:
Total emissive power: $\mathrm{E}=\int_{0}^{\infty} \varepsilon_{\lambda}(\lambda) \cdot \mathrm{E}_{\mathrm{b} \lambda} \mathrm{d} \lambda$.
Variation of $\varepsilon_{\lambda}$ wih $\lambda$ is specified in the problem.
Therefore, splitting the above integral into parts:

$$
\begin{aligned}
\mathrm{E} & =\int_{0}^{2} \varepsilon_{\lambda}(\lambda) \cdot \mathrm{E}_{\mathrm{b} \lambda} \mathrm{~d} \lambda+\int_{2}^{15} \varepsilon_{\lambda}(\lambda) \cdot \mathrm{E}_{\mathrm{b} \lambda} \mathrm{~d} \lambda+\int_{15}^{\infty} \varepsilon_{\lambda}(\lambda) \cdot \mathrm{E}_{\mathrm{b} \lambda} \mathrm{~d} \lambda \\
\text { i.e. } \quad \mathrm{E} & =0.1 \cdot \int_{0}^{2} \mathrm{E}_{\mathrm{b} \lambda} \mathrm{~d} \lambda+0.5 \cdot \int_{2}^{15} \mathrm{E}_{\mathrm{b} \lambda} \mathrm{~d} \lambda+0.8 \cdot \int_{15}^{\infty} \mathrm{E}_{\mathrm{b} \lambda} \mathrm{~d} \lambda
\end{aligned}
$$

Then, $\quad \varepsilon=\frac{\mathrm{E}}{\mathrm{E}_{\mathrm{b}}}=\varepsilon 1 \cdot\left(\mathrm{~F}_{0 \_\lambda} \lambda_{1}\right)+\varepsilon 2 \cdot\left(\mathrm{~F}_{\lambda .1-\lambda}\right)+\varepsilon 3 \cdot\left(\mathrm{~F}_{\lambda \lambda_{-} \text {infinity }}\right)$

$$
\begin{equation*}
\text { i.e. } \quad \varepsilon=\varepsilon 1 \cdot\left(\mathrm{~F}_{0} \lambda_{2}\right)+\varepsilon 2 \cdot\left(\mathrm{~F}_{0_{-}} \lambda .2-\mathrm{F}_{0_{-}} \lambda .1\right)+\varepsilon 3 \cdot\left(\mathrm{~F}_{0_{-} \text {infinity }}-\mathrm{F}_{0_{-} \lambda .2}\right) \tag{a}
\end{equation*}
$$



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Values of $\mathrm{F}_{0-2,1}$ etc. are obtained from the Mathcad Function written above, or from Table 13.2.
We have:

$$
\begin{aligned}
& \lambda_{1} \cdot \mathrm{~T}=1.6 \times 10^{3} \quad \text {..correspondingly, we get: } \\
& \mathrm{F}_{0 \_\lambda 1}:=\mathrm{F}_{0-\lambda}\left(\mathrm{M} 1, \lambda_{1} \cdot \mathrm{~T}\right) \\
& \text { i.e. } \mathrm{F}_{0_{-} \lambda 1}=0.0197
\end{aligned}
$$

And,

$$
\begin{aligned}
& \lambda_{2} \cdot \mathrm{~T}=1.2 \times 10^{4} \quad \text { correspondingly, we get: } \\
& \mathrm{F}_{0-\lambda .2}:=\mathrm{F}_{0 \_\lambda}\left(\mathrm{M} 1, \lambda_{2} \cdot \mathrm{~T}\right) \quad \text { i.e. } \mathrm{F}_{0 \_\lambda 2}=0.9451
\end{aligned}
$$

And,

$$
\lambda_{3} \cdot \mathrm{~T}=\infty \quad \text {...correspondingly, we get: }
$$

$$
F_{0_{2} \lambda 3}:=F_{0_{-} \lambda} \lambda(M 1,100001) \quad \text {.....Note: For } \lambda T=\text { infinity, we have put } \lambda T>100000 \text {. }
$$

$$
\text { i.e. } F_{0} \lambda_{\lambda}=1
$$

OR:

$$
F_{0_{0} \lambda_{3}}:=1 \quad \text {..from Table } 13.2
$$

Then, from eqn. (a):

$$
\begin{aligned}
& \varepsilon:=\varepsilon 1 \cdot F_{0_{-} \lambda .1}+\varepsilon 2 \cdot\left(F_{0_{-} \lambda .2}-F_{0_{-} \lambda .1}\right)+\varepsilon 3 \cdot\left(F_{0_{-} \lambda .3}-F_{0_{0} \lambda .2}\right) \\
& \text { i.e. } \quad \varepsilon=0.509 \quad \text {...Total hemisph. emissiviy over all wave lengths...Ans. }
\end{aligned}
$$

And, total emissive power of this surface is given by:

$$
\mathrm{E}:=\varepsilon \cdot \sigma \cdot \mathrm{T}^{4} \quad \mathrm{~W} / \mathrm{m}^{2} \ldots \text { total emissive power }
$$

i.e. $\quad E=1.199 \times 10^{4} \quad W / m^{2}$....total emissive power...Ans.

Plot the variation of total emissivity $\varepsilon$ and total emissive power E as Temp T varies from 500 K to 2000 K:

First, write the relevant quantities as functions of T:

$$
\begin{aligned}
& \mathrm{F}_{0 \_\lambda 1}(\mathrm{~T}):=\mathrm{F}_{0 \_\lambda}\left(\mathrm{M} 1, \lambda_{1} \cdot \mathrm{~T}\right) \quad \text { check: } \quad \mathrm{F}_{0 \_} \lambda_{1}(\mathrm{~T})=0.0197 \\
& \mathrm{~F}_{0 \_\lambda} 2_{2}(\mathrm{~T}):=\mathrm{F}_{0 \_} \lambda\left(\mathrm{M} 1, \lambda_{2} \cdot \mathrm{~T}\right) \quad \text { check: } \mathrm{F}_{0 \_\lambda .2}(\mathrm{~T})=0.9451 \\
& \mathrm{~F}_{0 \_\lambda} \lambda:=\mathrm{F}_{0 \_} \lambda(\mathrm{M} 1,100001) \quad \ldots \text { Note: For } \lambda \mathrm{T}=\text { infinity, we have: } \quad \mathrm{F}_{0 \_} \lambda .3=1
\end{aligned}
$$

$$
\begin{aligned}
& \text {...total hemisph. emissivity } \\
& \text { check: } \quad z(T)=0.509 \\
& E(T):=\varepsilon(T) \cdot \sigma \cdot T^{4} \quad W / m^{2} \ldots \text { total emissive power check: } E(T)=1.181 \times 10^{4}
\end{aligned}
$$

To plot the results:
$T:=500,510 . .2000 \quad$...define a range variable $T$

| $\mathrm{T}=$ | $\varepsilon(T)=$ | $E(T)=$ |
| :---: | :---: | :---: |
| 500 | 0.548 | $1.943 \cdot 10^{3}$ |
| 600 | 0.532 | $3.91 \cdot 10^{3}$ |
| 700 | 0.52 | $7.076 \cdot 10^{3}$ |
| 800 | 0.509 | 1.181.104 |
| 900 | 0.497 | 1.847.104 |
| 1000 | 0.483 | $2.737 \cdot 10^{4}$ |
| 1100 | 0.467 | $3.876 \cdot 10^{4}$ |
| 1200 | 0.45 | $5.287 \cdot 10^{4}$ |
| 1300 | 0.431 | $6.987 \cdot 10^{4}$ |
| 1400 | 0.413 | $8.991 \cdot 10^{4}$ |
| 1500 | 0.394 | $1.131 \cdot 10^{5}$ |
| 1600 | 0.376 | $1.395 \cdot 10^{5}$ |
| 1700 | 0.358 | $1.693 \cdot 10^{5}$ |
| 1800 | 0.341 | $2.027 \cdot 10^{5}$ |
| 1900 | 0.324 | $2.397 \cdot 10^{5}$ |
| 2000 | 0.309 | $2.804 \cdot 10^{5}$ |

## Plot:



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Prob.5A.9. Spectral hemispherical emissivity (monochromatic emissivity) of fire brick at 750 K as a function of wave length is as follows:
$\varepsilon 1=0.1$ for $\lambda=0$ to $2 \mu \mathrm{~m}, \varepsilon 2=0.6$ for $\lambda=2$ to $14 \mu \mathrm{~m}$, and $\varepsilon 3=0.3$ for $\lambda=14$ to $\infty$.Calculate total hemispherical emissivity $\varepsilon$, over all wave lengths. [VTU - July-Aug. 2004]

## Mathcad Solution:

This problem is similar to the previous problem.
So, we will enter the data, and use the Mathcad Functions already written:

## Data:

$$
\begin{array}{ll}
\mathrm{T}:=750 & \mathrm{~K} \text {....temperature } \\
z 1:=0.1 & \text { _.emissiviy in wavelength range: } 0 \text { to } 2 \mu \mathrm{~m} \\
z 2:=0.6 & \text { _.emissiviy in wavelength range: } 2 \text { to } 14 \mu \mathrm{~m} \\
z 3:=0.3 & \text {...emissiviy in wavelength range: } 14 \mu \mathrm{~m} \text { to } \infty . \\
\sigma:=5.67 \cdot 10^{-8} \mathrm{~W} /\left(\mathrm{m}^{2} . \mathrm{K}^{4}\right) \ldots . \text { Stefan-Boltzmann const. } \\
\lambda_{1}:=2 & \mu \mathrm{~m} \\
\lambda_{2}:=14 & \mu \mathrm{~m} \\
\lambda_{3}:=\infty
\end{array}
$$

## Results:

$$
\begin{aligned}
& \mathrm{F}_{0-\lambda} \lambda_{1}(\mathrm{~T}):=\mathrm{F}_{0_{-} \lambda}\left(\mathrm{M} 1, \lambda_{1} \cdot \mathrm{~T}\right) \quad \text { i.e: } \quad \mathrm{F}_{0_{-} \lambda .1}(\mathrm{~T})=0.01375 \\
& \mathrm{~F}_{0 \_1.2}(\mathrm{~T}):=\mathrm{F}_{0 \_\lambda}\left(\mathrm{M} 1, \lambda_{2} \cdot \mathrm{~T}\right) \quad \text { i.e.: } \quad \mathrm{F}_{0} \lambda_{2}(\mathrm{~T})=0.9237 \\
& \mathrm{~F}_{0_{-} \lambda_{3}}:=\mathrm{F}_{0_{-} \lambda}(\mathrm{M} 1,100001) \quad \text {.....Note: For } \lambda T=\text { infinity, we have: } \quad \mathrm{F}_{0_{-} \lambda .3}=1 \\
& \varepsilon(\mathrm{~T}):=21 \cdot \mathrm{~F}_{0_{-} \lambda .1}(\mathrm{~T})+22 \cdot\left(\mathrm{~F}_{0_{-} \lambda_{2}}(\mathrm{~T})-\mathrm{F}_{0_{-} \lambda_{12}}(\mathrm{~T})\right)+23 \cdot\left(\mathrm{~F}_{0_{-} \lambda_{2}}-\mathrm{F}_{0_{-} \lambda_{2}}(\mathrm{~T})\right) \text {..tolal hemisph. emissivity } \\
& \text { i.e.: } \quad z(T)=0.57 \quad \text {..total hemisph. emissivity ... Ans. } \\
& \mathrm{E}(\mathrm{~T}):=\varepsilon(\mathrm{T}) \cdot \sigma \cdot \mathrm{T}^{4} \quad \mathrm{~W} / \mathrm{m}^{2} \ldots \text { total emissive power } \\
& \text { i.e.: } \quad E(T)=1.023 \times 10^{4} \quad \mathrm{~W} / \mathrm{m}^{2} \text {....total emissive power.... Ans. }
\end{aligned}
$$

Plot the variation of total emissivity $\varepsilon$ and total emissive power $E$ as Temp T varies from 500 K to 2000 K:
$\mathrm{T}:=500,510 . .2000 \quad \mathrm{~T}:=500,510 . .2000$

| $\mathrm{T}=$ | $\varepsilon(\mathrm{T})=$ | $E(T)=$ |
| :---: | :---: | :---: |
| 500 | 0.542 | $1.922 \cdot 10^{3}$ |
| 600 | 0.56 | $4.117 \cdot 10^{3}$ |
| 700 | 0.569 | $7.747 \cdot 10^{3}$ |
| 800 | 0.571 | $1.325 \cdot 10^{4}$ |
| 900 | 0.566 | $2.105 \cdot 10^{4}$ |
| 1000 | 0.555 | $3.15 \cdot 10^{4}$ |
| 1100 | 0.541 | $4.489 \cdot 10^{4}$ |
| 1200 | 0.523 | $6.148 \cdot 10^{4}$ |
| 1300 | 0.503 | $8.143 \cdot 10^{4}$ |
| 1400 | 0.481 | $1.049 \cdot 10^{5}$ |
| 1500 | 0.459 | $1.319 \cdot 10^{5}$ |
| 1600 | 0.438 | $1.626 \cdot 10^{5}$ |
| 1700 | 0.416 | $1.972 \cdot 10^{5}$ |
| 1800 | 0.396 | $2.356 \cdot 10^{5}$ |
| 1900 | 0.376 | $2.78 \cdot 10^{5}$ |
| 2000 | 0.358 | $3.246 \cdot 10^{5}$ |

Total hemisph. emissivity vs Temp (K)




Prob.5A.10. For a black body maintained at 390 K , determine:

1) total emissive power
2) wave length at which max. monochromatic emissive power occurs
3) Max. monochromatic emissive power [M.U. May 1999]

## Mathcad Solution:

Data:

$$
\begin{aligned}
& \mathrm{C} 1:=3.742 \cdot 10^{8} \quad \mathrm{~W} \cdot \mu \mathrm{~m}^{\wedge} 4 / \mathrm{m}^{\wedge} 2 \quad \mathrm{C} 2:=1.4387 \cdot 10^{4} \quad \mu \mathrm{~m} . \mathrm{K} \\
& \mathrm{~T}:=390 \mathrm{~K} \quad \sigma:=5.67 \cdot 10^{-8} \mathrm{~W} / \mathrm{m}^{\wedge} 2 . \mathrm{K}^{\wedge} 4 \ldots \text { Stefan Boltzmann constant. }
\end{aligned}
$$

## Calculations:

$$
\mathrm{E}_{\mathrm{b}}:=\sigma \cdot \mathrm{T}^{4}
$$

$$
\text { i.e. } \quad E_{b}=1.312 \times 10^{3} \quad \mathrm{~W} / \mathrm{m}^{\wedge} 2 \ldots . . \text { Total emissive power.... Ans. }
$$

We have:

$$
\lambda_{\mathrm{m}}:=\frac{2897.8}{\mathrm{~T}}
$$

i.e. $\quad \lambda_{\mathrm{m}}=7.43$ microns....wave length for max. mono. emissive power at $390 \mathrm{~K} . .$. Ans.

And:

$$
\begin{aligned}
& E_{\mathrm{b} \lambda}:=\frac{\mathrm{C} 1 \cdot \lambda_{\mathrm{m}}-5}{\exp \left(\frac{\mathrm{C} 2}{\lambda_{\mathrm{m}} \cdot \mathrm{~T}}\right)-1} \\
& \text { i.e. } \quad \mathrm{E}_{\mathrm{b} \lambda \lambda}=116.129 \quad \text { W/m^2.micron...Max. monochr. emissive power...Ans. }
\end{aligned}
$$

"Prob.5A.11. For a body, of area $0.15 \mathrm{~m}^{\wedge} 2$, at an effective temp of 900 K , calculate the following: (i) Total rate of energy emission (E) (ii) Intensity of normal radiation ( $\mathrm{I}_{\mathrm{bn}}$ ), and (iii) Wave length of max. monochromatic emissive power".

Also plot the variation of $\lambda \_m$ with $T$, varying $T$ from 500 K to 2000 K .

## "EES Solution:"

"Data:"
$\mathrm{A}=0.15\left[\mathrm{~m}^{\wedge} 2\right]$
$\mathrm{T}=900[\mathrm{~K}]$
sigma $=5.67 \mathrm{E}-08\left[\mathrm{~W} / \mathrm{m}^{\wedge} 2-\mathrm{K}^{\wedge} 4\right]$ ".. Stefan Boltzmann constant"
"Calculations:"
E_b $=$ sigma * $\mathrm{T}^{\wedge} 4$ " $\left[\mathrm{W} / \mathrm{m}^{\wedge} 2\right] \ldots$ Total Emissive power"
$\mathrm{E}=\mathrm{E} \_\mathrm{b}$ * A " $[\mathrm{W}] \ldots$ Total rate of energy emission"

I_bn = E_b / pi "[W/m^2]...Intensity of normal radiation"
lambda_m * $\mathrm{T}=2898$ "...by Wein's displacement Law.. finds lambda_max, microns"

## Results:

Unit Settings: SI C Pa J mass deg
$\begin{aligned} A & =0.15\left[\mathrm{~m}^{2}\right] \\ \mathrm{I}_{\mathrm{bn}} & =11841\left[\mathrm{~W} / \mathrm{m}^{2}\right] \\ \mathrm{T} & =900[\mathrm{~K}]\end{aligned}$

| $E=5580$ [W] |
| :--- |
| $\lambda_{\mathrm{m}}=3.22$ [microns] |

$$
\begin{aligned}
& \mathrm{E}_{\mathrm{b}}=37201\left[\mathrm{~W} / \mathrm{m}^{2}\right] \\
& \sigma=5.670 \mathrm{E}-08\left[\mathrm{~W} / \mathrm{m}^{2} \mathrm{~K}^{4}\right]
\end{aligned}
$$

Thus:
Total rate of energy emission $=\mathrm{E}=5580 \mathrm{~W}$

Intensity of normal radiation $=11841 \mathrm{~W} / \mathrm{m}^{\wedge} 2$

Wave length of max. monochromatic emissive power $=3.22$ microns $\ldots$. Ans.

## Also plot the variation of $\lambda_{-} m$ with $T$, varying $T$ from 500 K to 2000 K :

First, compute the Parametric Table:


## Now, plot the results:




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"Prob.5A.12. Considering an Industrial furnace at 2500 K as a black body, find out the following: (i) monochromatic emissive power at a wave length of 1.2 microns, (ii) wave length at which emission is maximum, (iii) max. emissive power, and (iv) total emissive power".

## "EES Solution:"

"Data:"
$\mathrm{T}=2500[\mathrm{~K}]$
lambda $=1.2$ [microns]
sigma $=5.67 \mathrm{E}-08\left[\mathrm{~W} / \mathrm{m}^{\wedge} 2-\mathrm{K}^{\wedge} 4\right]^{"} \ldots$ Stefan Boltzmann constant"
$\mathrm{C} 1=3.742 \mathrm{E} 08\left[\mathrm{~W}-\text { micron } \wedge 4 / \mathrm{m}^{\wedge} 2\right]^{"} \ldots$ const. in Planck's law eqn"
$\mathrm{C} 2=1.4388 \mathrm{E} 04[\mathrm{~m}-\mathrm{K}]^{"} \ldots$ const. in Planck's law eqn"

## "Calculations:"

E_lambda_b $=\mathrm{C} 1{ }^{\star} \operatorname{lambda} \wedge(-5) /\left(\exp \left(\mathrm{C} 2 /\left(\operatorname{lambda}{ }^{*} \mathrm{~T}\right)\right)-1\right) "[\mathrm{~W} / \mathrm{m} \wedge 2] \ldots$ monochr. emissive power at lambda $=2$ microns"
lambda_m * $\mathrm{T}=2898$ "...by Weins displacement Law.. finds lambda_max, microns"

E_lamda_b_max $=1.285 \mathrm{E}-05 * \mathrm{~T}^{\wedge} 5$ " $\left[\mathrm{W} / \mathrm{m}^{\wedge} 2\right.$ per m length $] \ldots$ Max. mono. emissive power"

E_b $=$ sigma * $\mathrm{T}^{\wedge} 4$ " $\left[\mathrm{W} / \mathrm{m}^{\wedge} 2\right] \ldots$ Total emissive power"

## Results:

Unit Settings: SI C kPa kJ mass deg
$\mathrm{C} 1=3.742 \mathrm{E}+08\left[\mathrm{~W}-\right.$ micron $\left.4 / \mathrm{m}^{2}\right]$
$\mathrm{E}_{\mathrm{b}}=2.215 \mathrm{E}+06\left[\mathrm{~W} / \mathrm{m}^{2}\right]$
$\mathrm{E}_{\text {lamda,b,max }}=1.255 \mathrm{E}+12\left[\mathrm{~W} / \mathrm{m}^{2}\right.$ per.m.length $]$
$\lambda_{\mathrm{m}}=1.159$ [microns]
$C 2=14388[\mathrm{~m}-\mathrm{K}]$
$\mathrm{E}_{\lambda, \mathrm{b}}=1.253 \mathrm{E}+06\left[\mathrm{~W} / \mathrm{m}^{2}\right]$
$\lambda=1.2[$ microns $]$
$\sigma=5.670 \mathrm{E}-08\left[\mathrm{~W} / \mathrm{m}^{2} \mathrm{~K}^{4}\right]$
$\mathrm{T}=2500[\mathrm{~K}]$

Thus:
Monochr. emissive power at $1.2 \mu \mathrm{~m}=\mathrm{E}_{-}$lambda_b $=1.253 \mathrm{E} 06 \mathrm{~W} / \mathrm{m}^{\wedge} 2$

Wave length at which emission is $\max =\lambda \_m=1.159 \mu \mathrm{~m}$

Max. mono. emissive power $=E \_$lambda_b_max $=1.255 E 12 \mathrm{~W} / \mathrm{m}^{\wedge} 2$ per m length

Total emissive power $=E_{-} b=2.215 E 06 \mathrm{~W} / \mathrm{m}^{\wedge} 2 \ldots$ Ans.

Prob.5A.13: Draw a plot of Planck's Law for spectral distribution with EXCEL:

## EXCEL Solution:

We have already stated the Planck's Law, i.e.

$$
\mathrm{E}_{\mathrm{b} \lambda}(\lambda, \mathrm{~T}):=\frac{\mathrm{C}_{1} \cdot \lambda^{-5}}{\exp \left(\frac{\mathrm{C}_{2}}{\lambda \cdot \mathrm{~T}}\right)-1} \quad \ldots \mathrm{E}_{\mathrm{b} \lambda} \text { as a function of } \lambda \text { and } \mathrm{T}
$$

Where C1 and C2 are constants, given as:

C1 $=3.742 \mathrm{E} 08\left[\mathrm{~W} \text {-micron^4 } 4 \mathrm{~m}^{\wedge} 2\right]^{\prime \prime} \ldots$ const. in Planck's law eqn"
$\mathrm{C} 2=1.4388 \mathrm{E} 04[\mathrm{~m}-\mathrm{K}]^{\prime \prime} \ldots$ const. in Planck's law eqn"

Now, let us write a VBA Function in EXCEL to produce a Table of $\lambda$ against T and plot the same for $\mathrm{T}=5800 \mathrm{~K}, 1000 \mathrm{~K}, 500 \mathrm{~K}$ and 100 K :

## Following are the steps in EXCEL:

1. In EXCEL go to: Developer - Visual Basic:


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2．Click on Visual Basic－Insert：

| ＊镸 Eile Edit View | Insert Format |  | Debus |
| :---: | :---: | :---: | :---: |
|  | $\%$ | Procedure．．． |  |
| Project－VBAProject | 包 | UserForm |  |
| 圂 國 $\square$ |  | Module |  |
| ＋题 IHTADDIN（IHTXL |  | Class Module |  |
| ＋题 TPX（tpx．xla） <br> + WS VRAProiect（Airm |  | File．．． |  |

3．Click on Module and a blank window（Module 1）presents itself．Write the code in that window as shown below：

4. Now, this Function is available in EXCEL worksheet like any other built-in Function of EXCEL, under the 'User Defined' category. Let us use it to produce the Table and plot as shown below:

Only part of the Table is shown below. Note the formula used (i.e. the VBA Function written above) in the Formula bar:

| 4 | C | - $f_{x}$ | =Plancks_Law(B10,5800) |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | A | B | C | D | E | F |
| 4 |  |  |  |  |  |  |
| 5 |  | Planck's Law for Spectral distribution: (Ref: MT, p.645) |  |  |  |  |
| 6 |  |  |  |  |  |  |
| 7 |  |  | $\mathrm{T}=5800 \mathrm{~K}$ | $\mathrm{T}=1000 \mathrm{~K}$ | $\mathrm{T}=500 \mathrm{~K}$ | $\mathrm{T}=100 \mathrm{~K}$ |
| 8 |  | Lambda (microns) | E_b_lambda (W/m^2/micron) |  |  |  |
| 9 |  |  | E_b_lambda(W |  |  |  |
| 10 |  | 0.06 | 0.000534275 |  |  |  |
| 11 |  | 0.07 | 0.090776083 |  |  |  |
| 12 |  | 0.08 | 3.905862682 |  |  |  |
| 13 |  | 0.09 | 67.94688745 |  |  |  |
| 14 |  | 0.1 | 631.4722659 |  |  |  |
| 15 |  | 0.2 | 4803754.872 |  |  |  |
| 16 |  | 0.3 | 39508670.79 |  |  |  |
| 17 |  | 0.4 | 74216005.51 | 8.7562E-06 |  |  |
| 18 |  | 0.5 | 84480620.57 | 0.00381825 |  |  |
| 19 |  | 0.6 | 78325115.48 | 0.18564775 |  |  |
| 20 |  | 0.7 | 66281071.45 | 2.64010463 |  |  |
| 21 |  | 0.8 | 53835149.43 | 17.6770775 |  |  |
| 22 |  | 0.9 | 42996046.38 | 72.3523627 |  |  |
| 23 |  | 1 | 34181499.37 | 211.304778 | 0.0001193 |  |
| 24 |  | 1.2 | 21788716.23 | 934.05258 | 0.0058015 |  |
| 25 |  | 1.4 | 14253369.15 | 2395.97531 | 0.0825033 |  |
| 26 |  | 1.6 | 9611245.643 | 4440.54435 | 0.5524087 |  |


|  | A | B | C | D | E | F |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 145 | 740 | 0.000502235 | $8.5897 \mathrm{E}-05$ | $4.253 \mathrm{E}-05$ | $7.8579 \mathrm{E}-06$ |  |
| 146 | 760 | 0.000451438 | $7.7226 \mathrm{E}-05$ | $3.825 \mathrm{E}-05$ | $7.0815 \mathrm{E}-06$ |  |
| 147 | 780 | 0.000406904 | $6.9622 \mathrm{E}-05$ | $3.449 \mathrm{E}-05$ | $6.3986 \mathrm{E}-06$ |  |
| 148 | 800 | 0.000367729 | $6.2931 \mathrm{E}-05$ | $3.118 \mathrm{E}-05$ | $5.7961 \mathrm{E}-06$ |  |
| 149 |  | 820 | 0.000333157 | $5.7025 \mathrm{E}-05$ | $2.826 \mathrm{E}-05$ | $5.2629 \mathrm{E}-06$ |
| 150 | 840 | 0.000302554 | $5.1795 \mathrm{E}-05$ | $2.568 \mathrm{E}-05$ | $4.7895 \mathrm{E}-06$ |  |
| 151 | 860 | 0.000275386 | $4.7152 \mathrm{E}-05$ | $2.338 \mathrm{E}-05$ | $4.3682 \mathrm{E}-06$ |  |
| 152 | 880 | 0.000251199 | $4.3018 \mathrm{E}-05$ | $2.133 \mathrm{E}-05$ | $3.9923 \mathrm{E}-06$ |  |
| 153 | 900 | 0.000229611 | $3.9327 \mathrm{E}-05$ | $1.951 \mathrm{E}-05$ | $3.6559 \mathrm{E}-06$ |  |
| 154 | 920 | 0.000210293 | $3.6023 \mathrm{E}-05$ | $1.787 \mathrm{E}-05$ | $3.3541 \mathrm{E}-06$ |  |
| 155 | 940 | 0.000192964 | $3.3059 \mathrm{E}-05$ | $1.64 \mathrm{E}-05$ | $3.0829 \mathrm{E}-06$ |  |
| 156 |  | 960 | 0.000177384 | $3.0394 \mathrm{E}-05$ | $1.508 \mathrm{E}-05$ | $2.8386 \mathrm{E}-06$ |
| 157 |  | 1000 | 0.000150669 | $2.5823 \mathrm{E}-05$ | $1.282 \mathrm{E}-05$ | $2.4183 \mathrm{E}-06$ |
| 158 |  |  |  |  |  |  |

Now, plot the results:


Prob.5A.14: Write an EXCEL VBA program to generate the Radiation Functions (i.e. radiation emitted in a wave band as a fraction of Total emissive power) against $\lambda \mathrm{T}$.

## EXCEL Solution:

Recollect Planck's Law:

$$
\begin{aligned}
& \mathrm{E}_{\mathrm{b} \lambda}(\lambda)=\frac{\mathrm{C}_{1} \cdot \lambda^{-5}}{\exp \left(\frac{\mathrm{C}_{2}}{\lambda \cdot \mathrm{~T}}\right)-1} \quad \mathrm{~W} /(\mathrm{m} 2 . \mu \mathrm{m}) \ldots .(13.7) \\
& \mathrm{C}_{1}=3.742 \cdot 10^{8} \quad \mathrm{~W} \cdot \mu \mathrm{~m} 4 / \mathrm{m} 2 \\
& \mathrm{C}_{2}=1.4387 \cdot 10^{4} \quad \mu \mathrm{~m} \cdot \mathrm{~K}
\end{aligned}
$$

where,
and,

## Radiation from a wave band:

Often, it is required to know the amount of radiation emitted in a given wave band, i.e. in a wave length interval between $\lambda_{1}$ and $\lambda_{2}$. This is expressed as a fraction of the total emissive power and is written as

$F \lambda_{1} \lambda_{2}$. Then, we can write:

$$
\begin{align*}
& F_{\lambda 1 \_\lambda}=\frac{\int_{-\lambda}^{\lambda}{ }^{\lambda} \mathrm{E}_{\mathrm{b} \lambda} \mathrm{~d} \lambda}{\int_{0}^{\infty} \mathrm{E}_{\mathrm{b} \lambda} \mathrm{~d} \lambda}=\frac{1}{\sigma \cdot \mathrm{~T}^{4}} \cdot \int_{\partial \lambda}^{\rho \lambda}{ }_{1} \mathrm{E}_{\mathrm{b} \lambda} \mathrm{~d} \lambda \\
& \text { i.e. } \quad F_{\lambda 1 \_\lambda 2}=\frac{1}{\sigma \cdot T^{4}} \cdot\left(\int_{0}^{\lambda \lambda} \quad E_{b \lambda} d \lambda-\int_{0}^{\rho \lambda} \begin{array}{ll}
1 & E_{b \lambda} d \lambda
\end{array}\right\} \\
& \text { i.e. } \\
& \mathrm{F}_{\lambda 1 \_\lambda 2}=\mathrm{F}_{0 \_\lambda 2}-\mathrm{F}_{0-\lambda 1}
\end{align*}
$$

Above formula is not very convenient to use, since $\mathrm{E}_{\mathrm{b}} \lambda$ depends on absolute temperature T , and it is not practicable to tabulate $F_{0} \lambda$ for each $T$. This difficulty is overcome by expressing $F_{0} \lambda$ as follows:

$$
\mathrm{F}_{0-\lambda}=\frac{\int_{0}^{\rho \lambda} \mathrm{E}_{\mathrm{b} \lambda} \mathrm{~d} \lambda}{\sigma \cdot \mathrm{~T}^{4}}=\frac{\int_{0}^{\lambda \cdot \mathrm{T}} \mathrm{E}_{\mathrm{b} \lambda} \mathrm{~d}(\lambda \cdot \mathrm{~T})}{\sigma \cdot \mathrm{T}^{5}}
$$

i.e.

$$
\mathrm{F}_{0-\lambda}=\mathrm{f}(\lambda \cdot \mathrm{~T})
$$

So, we shall, first, write a VBA Function to calculate $\mathrm{F}_{0-\lambda}$ and then, use it to write another Function for $\mathrm{F}_{\lambda 1-\lambda 2}$.

Following are the VBA Functions, where integrals are calculated by numerical method, using the Trapezoidal rule.

```
(General) - F_zero_lambdaT1_lambdaT2_alternative
    Option Explicit
    Function F(x As Double) As Double
    'This function is used to calculate the fraction F_0_lambda, Ref: Incropera
    Dim C_1 As Double, C_2 As Double, sigma As Double
    C_1 = 374200000# 'W.micron^4/m2
    C_2 = 14388# 'micron.K
    sigma = 0.0000000567 'W/m2.K^4
    F = C_1 / (sigma * x ^ 5 * (Exp (C_2 / x) - 1))
    End Function
    Function F_zero_lambdaT_Low_lambdaT_High_trapezoidal(LambdaT_Low As Double, _
    LambdaT_High As Double, n As Integer) As Double
    'Gives the fraction F_zero_Lambda by integration using Trapezoidal rule
    'Input: LambdaT_Low, LambdaT_High in micron.K, n is the no. of equal intervals
    Dim dx As Double, sum As Double, y As Double, i As Integer
    'If LambdaT Low is less than 21, F(x) goes to extremely low value and gives error!
    If LambdaT_Low < 21 Then
        LambdaT_Low = 21
    End If
    If LambdaT_High > 100000 Then
        LambdaT_High = 100000
    End If
    dx = (LambdaT_High - LambdaT_Low) / n
    sum = (F(LambdaT_Low) + F(LambdaT_High)) / 2
    For i = 1 To (n - 1)
        y = F(LambdaT_Low + (dx * i))
        sum = sum + y
    Next i
    sum = sum * dx
    F_zero_lambdaT_Low_lambdaT_High_trapezoidal = sum
    End Function
```

Now, we shall use these Functions to generate a Table of F0- $\lambda \mathrm{T}$ against $\lambda \mathrm{T}$ as shown below. Note that we have taken lower value of $\lambda \mathrm{T}$ as 20 micron. K for reasons stated in the program. In addition, we have compared the thus calculated values to the Table values given in Incropera.

Also, note that 1000 steps have been taken while implementing the Trapezoidal rule. Errors as compared to Table values in Incropera are within $1.08 \%$.




# "I studied English for 16 years but... <br> ...I finally learned to speak it in just six lessons" <br> Jane, Chinese architect 



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## Now, plot these results in EXCEL:



Prob.5A.15. Spectral hemispherical emissivity (monochromatic emissivity) of fire brick at 750 K as a function of wave length is as follows:
$\varepsilon 1=0.1$ for $\lambda=0$ to $2 \mu \mathrm{~m}, \varepsilon 2=0.6$ for $\lambda=2$ to $14 \mu \mathrm{~m}$, and $\varepsilon 3=0.3$ for $\lambda=14$ to $\infty$. Calculate total hemispherical emissivity $\varepsilon$, over all wave lengths. [VTU - July-Aug. 2004]

Also, plot the variation of total emissivity $\varepsilon$ and total emissive power E as Temp T varies from 500 K to 2000 K :

Note: This is the same as Prob.5A. 9 which was solved with Mathcad.

## EXCEL Solution:

Let us use the VBA Function written above for F0- $\lambda$ while solving this problem.

1. Set up the EXCEL worksheet, enter data and name the cells, as shown below:

| lambda2 |  |  | $\checkmark$ - $f_{x}$ |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
| - | A | B | C | D | E |
| 96 |  |  |  |  |  |
| 97 |  |  |  |  |  |
| 98 |  |  | Prob.5A.15. |  |  |
| 99 |  |  |  |  |  |
| 100 |  |  | Data: |  |  |
| 101 |  |  | T | 750 | K |
| 102 |  |  | epsilon1 | 0.1 | emissivity ( $0-2 \mu$ ) |
| 103 |  |  | epsilon2 | 0.6 | emissivity ( $2-14 \mu$ ) |
| 104 |  |  | epsilon3 | 0.3 | emissivity ( $14 \mu$ - infinity) |
| 105 |  |  | sigma | $5.67 \mathrm{E}-08$ | $\mathrm{W} / \mathrm{m}^{\wedge} 2 . \mathrm{K}^{\wedge} 4$ |
| 106 |  |  | lambda1 | 2 | microns |
| 107 |  |  | lambda2 | 14 | microns |
| 108 |  |  | lambda3 | $1.00 \mathrm{E}+10$ | microns |

2. Do the calculations:

| F_0_lambdaT_1 |  |  | - $f_{x}$ =F_zero_lambdaT_Low_lambdaT_High_trapezoidal(0,lambdaT_1,1000) |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 4 | A | B | C | D |  | E | F |
| 109 |  |  |  |  |  |  |  |
| 110 |  |  | Calculations: |  |  |  |  |
| 111 |  |  |  |  |  |  |  |
| 112 |  |  | lambdaT_1 | 1500 | micron.K | multiplying lambda and T |  |
| 113 |  |  | lambdaT_2 | 10500 | micron.K |  |  |
| 114 |  |  | lambdaT_3 | 7.5E+12 | micron.K |  |  |
| 115 |  |  |  |  |  |  |  |
| 116 |  |  | Now, get F_0_lambdaT using the VBA Function: |  |  |  |  |
| 117 |  |  |  |  |  |  |  |
| 118 |  |  | F_0_lambdaT_1 | 0.012849488 |  | Using VBA Functions for F_0_lambdaT |  |
| 119 |  |  | F_0_lambdaT_2 | 0.923721652 |  |  |  |
| 120 |  |  | F_0_lambdaT_3 | 0.999917917 |  |  |  |
| 121 |  |  |  |  |  |  |  |

Note: In the above screen shot, see the VBA Function invoked in cell D118, in the Formula bar.
3. Continuing with the calculation:


## Thus:

Total hemisph. Emissivity $=\mathrm{e}=0.571 \ldots$ Ans.

Total Emissive Power $=\mathrm{E}=10237.9 \mathrm{~W} / \mathrm{m}^{\wedge} 2 \ldots$ Ans.


To plot the variation of total emissivity $\varepsilon$ and total emissive power E as Temp T varies from 500 K to 2000 K :

1. Set up a Table as shown, and enter the first row below the column titles, i.e. row 133, using the Formulas already written. However, take care to enter T with relative reference in the formulas, so that we can drag-copy the first row to fill up the entire Table:


Note how the VBA Function is entered in cell G133, in the Formula bar. Similarly, for all other columns, use relative references to appropriate quantities.
2. Now, select row 133 i.e. from cell D133 to cell K133, and drag-copy to the end of the Table, i.e. up to cell K163. Immediately, all calculations are made and the Table is filled up:

|  | B | C | D | E | F | G | H | 1 | J | K |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 132 |  | T (K) | lambdaT_1(micron.K) | lambdaT_2(micron.K) | lambdaT_3(micron.K) | F_0_lambdaT_1 | F_0_lambdaT_2 | F_0_lambdaT_3 | epsilon | $\mathrm{E}\left(\mathrm{W} / \mathrm{m}^{\wedge} 2\right)$ |
| 133 |  | 500 | 1000 | 7000 | $5 \mathrm{E}+12$ | 0.000320733 | 0.80811908 | 0.999917917 | 0.542 | 1921.601 |
| 134 |  | 550 | 1100 | 7700 | $5.5 \mathrm{E}+12$ | 0.000911145 | 0.843645005 | 0.999917917 | 0.553 | 2867.181 |
| 135 |  | 600 | 1200 | 8400 | $6 \mathrm{E}+12$ | 0.002134036 | 0.871206025 | 0.999917917 | 0.560 | 4117.044 |
| 136 |  | 650 | 1300 | 9100 | $6.5 \mathrm{E}+12$ | 0.004316168 | 0.892826594 | 0.999917917 | 0.566 | 5725.270 |
| 137 |  | 700 | 1400 | 9800 | $7 \mathrm{E}+12$ | 0.007789953 | 0.909975128 | 0.999917917 | 0.569 | 7747.171 |
| 138 |  | 750 | 1500 | 10500 | $7.5 \mathrm{E}+12$ | 0.012849488 | 0.923721652 | 0.999917917 | 0.571 | 10237.902 |
| 139 |  | 800 | 1600 | 11200 | $8 \mathrm{E}+12$ | 0.019718434 | 0.934852175 | 0.999917917 | 0.571 | 13251.142 |
| 140 |  | 850 | 1700 | 11900 | $8.5 \mathrm{E}+12$ | 0.028533364 | 0.94394967 | 0.999917917 | 0.569 | 16837.973 |
| 141 |  | 900 | 1800 | 12600 | $9 \mathrm{E}+12$ | 0.039341265 | 0.951450973 | 0.999917917 | 0.566 | 21046.021 |
| 142 |  | 950 | 1900 | 13300 | $9.5 \mathrm{E}+12$ | 0.052107386 | 0.957686825 | 0.999917917 | 0.561 | 25918.902 |
| 143 |  | 1000 | 2000 | 14000 | $1 \mathrm{E}+13$ | 0.066729199 | 0.962910186 | 0.999917917 | 0.555 | 31495.933 |
| 144 |  | 1050 | 2100 | 14700 | $1.05 \mathrm{E}+13$ | 0.083052858 | 0.967316428 | 0.999917917 | 0.549 | 37812.099 |
| 145 |  | 1100 | 2200 | 15400 | $1.1 \mathrm{E}+13$ | 0.10088958 | 0.971057862 | 0.999917917 | 0.541 | 44898.205 |
| 146 |  | 1150 | 2300 | 16100 | $1.15 \mathrm{E}+13$ | 0.120030358 | 0.974254285 | 0.999917917 | 0.532 | 52781.176 |
| 147 |  | 1200 | 2400 | 16800 | 1.2E+13 | 0.140258237 | 0.977000715 | 0.999917917 | 0.523 | 61484.448 |
| 148 |  | 1250 | 2500 | 17500 | $1.25 \mathrm{E}+13$ | 0.161357934 | 0.979373116 | 0.999917917 | 0.513 | 71028.425 |
| 149 |  | 1300 | 2600 | 18200 | $1.3 \mathrm{E}+13$ | 0.18312292 | 0.981432674 | 0.999917917 | 0.503 | 81430.949 |
| 150 |  | 1350 | 2700 | 18900 | $1.35 \mathrm{E}+13$ | 0.20536028 | 0.983229021 | 0.999917917 | 0.492 | 92707.775 |
| 151 |  | 1400 | 2800 | 19600 | $1.4 \mathrm{E}+13$ | 0.227893746 | 0.984802687 | 0.999917917 | 0.481 | 104873.028 |
| 152 |  | 1450 | 2900 | 20300 | $1.45 \mathrm{E}+13$ | 0.250565263 | 0.986186975 | 0.999917917 | 0.471 | 117939.633 |
| 153 |  | 1500 | 3000 | 21000 | $1.5 \mathrm{E}+13$ | 0.273235479 | 0.987409416 | 0.999917917 | 0.460 | 131919.699 |
| 154 |  | 1550 | 3100 | 21700 | $1.55 \mathrm{E}+13$ | 0.295783431 | 0.988492894 | 0.999917917 | 0.449 | 146824.876 |
| 155 |  | 1600 | 3200 | 22400 | $1.6 \mathrm{E}+13$ | 0.318105697 | 0.989456534 | 0.999917917 | 0.438 | 162666.663 |
| 156 |  | 1650 | 3300 | 23100 | $1.65 \mathrm{E}+13$ | 0.340115205 | 0.990316401 | 0.999917917 | 0.427 | 179456.677 |
| 157 |  | 1700 | 3400 | 23800 | $1.7 \mathrm{E}+13$ | 0.361739836 | 0.99108605 | 0.999917917 | 0.416 | 197206.888 |
| 158 |  | 1750 | 3500 | 24500 | $1.75 \mathrm{E}+13$ | 0.382920952 | 0.991776974 | 0.999917917 | 0.406 | 215929.819 |
| 159 |  | 1800 | 3600 | 25200 | $1.8 \mathrm{E}+13$ | 0.40361191 | 0.992398958 | 0.999917917 | 0.396 | 235638.708 |
| 160 |  | 1850 | 3700 | 25900 | $1.85 \mathrm{E}+13$ | 0.423776624 | 0.992960368 | 0.999917917 | 0.386 | 256347.652 |
| 161 |  | 1900 | 3800 | 26600 | $1.9 \mathrm{E}+13$ | 0.443388211 | 0.993468383 | 0.999917917 | 0.376 | 278071.719 |
| 162 |  | 1950 | 3900 | 27300 | $1.95 \mathrm{E}+13$ | 0.462427733 | 0.993929189 | 0.999917917 | 0.367 | 300827.043 |
| 163 |  | 2000 | 4000 | 28000 | $2 \mathrm{E}+13$ | 0.480883053 | 0.994348131 | 0.999917917 | 0.358 | 324630.895 |

3. Now, plot the results in EXCEL:



Note that results obtained with EXCEL match very well with those obtained with Mathcad, as they should.


## 5B View factors (or, Shape factor, configuration factor) and Radiation energy exchange between Black bodies:

## Recollect:

Definition of view factor: it is the fraction of energy emitted by surface 1 , which is intercepted directly by surface 2 .

Note: Often, while solving radiation problems, determination of the view factor is the most difficult part. It will be useful to keep in mind the definition of view factor, summation rule, reciprocity relation, superposition rule and symmetry rule while attempting to find out the view factors.

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## Radiation energy exchange between black bodies:

Analysis of heat exchange between two black bodies is simpler since a black body absorbs all the radiation impinging on it and none is reflected.

Consider two black surfaces $A_{1}$ and $A_{2}$ at temperatures T 1 and $T 2$, exchanging radiation energy with each other.

Then:

$$
\begin{equation*}
Q_{\text {net }}=A_{1} \cdot F_{12} \cdot \sigma \cdot\left\{\mathrm{~T}_{1}^{4}-\mathrm{T}_{2}^{4}\right\}=\mathrm{A}_{2} \cdot \mathrm{~F}_{21} \cdot \sigma \cdot\left(\mathrm{~T}_{1}^{4}-\mathrm{T}_{2}^{4}\right\} \tag{13.40}
\end{equation*}
$$

## Prob.5B.1 Draw the plots for View factors of:

1) Aligned, parallel rectangles of equal size
2) Coaxial, parallel disks, and
3) Perpendicular rectangles with a common edge.

View factor relationships for a few important shapes, given at the beginning of this chapter, are reproduced below. (Ref: Cengel).

Now, we shall write Mathcad Functions for View factors, so that we need not refer to the graphs while doing calculations.

We will also use these Functions to draw the graphs.

View factors (or, Shape factor, configuration factor) and Radiation energy exchange between Black bodies:

| Geometry | Relation |
| :---: | :---: |
| Aligned parallel rectangles | $\begin{aligned} \bar{X}= & X / L \text { and } \bar{Y}=Y / L \\ F_{i \rightarrow j} & =\frac{2}{\pi \bar{X} \bar{Y}}\left\{\ln \left[\frac{\left(1+\bar{X}^{2}\right)\left(1+\bar{Y}^{2}\right)}{1+\bar{X}^{2}+\bar{Y}^{2}}\right]^{1 / 2}\right. \\ & +\bar{X}\left(1+\bar{Y}^{2}\right)^{1 / 2} \tan ^{-1} \frac{\bar{X}}{\left(1+\bar{Y}^{2}\right)^{1 / 2}} \\ & +\bar{Y}\left(1+\bar{X}^{2}\right)^{1 / 2} \tan ^{-1} \frac{\bar{Y}}{\left(1+\bar{X}^{2}\right)^{1 / 2}} \\ & \left.-\bar{X} \tan ^{-1} \bar{X}-\bar{Y} \tan ^{-1} \bar{Y}\right\} \end{aligned}$ |
| Coaxial parallel disks | $\begin{aligned} R_{i} & =r_{i} / L \text { and } R_{j}=r_{j} / L \\ S & =1+\frac{1+R_{j}^{2}}{R_{i}^{2}} \\ F_{i \rightarrow j} & =\frac{1}{2}\left\{S-\left[S^{2}-4\left(\frac{r_{j}}{r_{i}}\right)^{2}\right]^{1 / 2}\right\} \end{aligned}$ |


| Perpendicular rectangles <br> with a common edge | $H=Z / X$ and $W=Y / X$ |
| :--- | :--- |
| $F_{i \rightarrow j}$ $=\frac{1}{\pi W}\left(W \tan ^{-1} \frac{1}{W}+H \tan ^{-1} \frac{1}{H}\right.$ <br>  $-\left(H^{2}+W^{2}\right)^{1 / 2} \tan ^{-1} \frac{1}{\left(H^{2}+W^{2}\right)^{1 / 2}}$ <br>  $+\frac{1}{4} \ln \left\{\frac{\left(1+W^{2}\right)\left(1+H^{2}\right)}{1+W^{2}+H^{2}}\right.$ <br>  $\times\left[\frac{W^{2}\left(1+W^{2}+H^{2}\right)}{\left(1+W^{2}\right)\left(W^{2}+H^{2}\right)}\right]^{W^{2}}$ <br>  $\left.\left.\times\left[\frac{H^{2}\left(1+H^{2}+W^{2}\right)}{\left(1+H^{2}\right)\left(H^{2}+W^{2}\right)}\right]^{H^{2}}\right\}\right)$ |  |

1. View factors for aligned, parallel rectangles: (See the fig. above for nomenclature)

We define:

$$
\begin{aligned}
X X & =\frac{X}{L} \quad Y Y=\frac{Y}{L} \\
A(X X, Y Y) & :=\frac{2}{\pi \cdot X X \cdot Y Y} \quad B(X X, Y Y):=\ln \left[\sqrt{\left.\frac{\left(1+X X^{2}\right) \cdot\left(1+Y^{2}\right)}{1+X X^{2}+Y^{2}}\right]}\right]
\end{aligned}
$$

Software Solutions to Problems on Heat Transfer: Radiation Heat Transfer - Part I

View factors (or, Shape factor, configuration factor) and Radiation energy exchange between Black bodies:

$$
\begin{aligned}
& C(X X, Y Y):=X X \cdot\left(1+Y Y^{2}\right)^{\frac{1}{2}} \cdot \operatorname{atan}\left[\frac{X X}{\left(1+Y^{2}\right)^{\frac{1}{2}}}\right] \\
& D(X X, Y Y):=Y Y \cdot\left(1+X X^{2}\right)^{\frac{1}{2}} \cdot \operatorname{atan}\left[\frac{Y Y}{\left(1+X X^{2}\right)^{\frac{1}{2}}}\right] \\
& E(X X):=X X \cdot \operatorname{atan}(X X) \quad F(Y Y):=Y Y \cdot \operatorname{atan}(Y Y)
\end{aligned}
$$

$$
F_{i j}(X X, Y Y):=A(X X, Y Y) \cdot(B(X X, Y Y)+C(X X, Y Y)+D(X X, Y Y)-E(X X)-F(Y Y))
$$



## Now, to plot the graphs:

$$
\mathrm{XX}:=0.1,0.2 . .30 \quad \text {...define a range variable: }
$$



We write the Mathcad Function for F12 for aligned, parallel rectangles as follows:

## Here,

Inputs are: $\mathrm{X}, \mathrm{Y}$ and L .

## Output is: F12.

$$
\begin{aligned}
& \text { F12_parallel_rectangles }(X, Y, L):=\left\{\begin{array}{l}
\mathrm{XX} \leftarrow \frac{\mathrm{X}}{\mathrm{~L}} \\
\mathrm{YY} \leftarrow \frac{\mathrm{Y}}{\mathrm{~L}}
\end{array}\right. \\
& \mathrm{A} \leftarrow \frac{2}{\pi \cdot \mathrm{XX} \cdot \mathrm{YY}} \\
& B \leftarrow \ln \left[\sqrt{\frac{\left(1+X X^{2}\right) \cdot\left(1+Y Y^{2}\right)}{1+X X^{2}+Y Y^{2}}}\right] \\
& \mathrm{C} \leftarrow \mathrm{XX} \cdot\left(1+\mathrm{YY}^{2}\right)^{\frac{1}{2}} \cdot \operatorname{atan}\left[\frac{\mathrm{XX}}{(1+\mathrm{YY})^{2}}\right] \\
& \mathrm{D} \leftarrow \mathrm{YY} \cdot\left(1+\mathrm{XX}^{2}\right)^{\frac{1}{2}} \cdot \operatorname{atan}\left[\frac{\mathrm{YY}}{\left(1+\mathrm{XX}^{2}\right)^{\frac{1}{2}}}\right] \\
& \mathrm{E} \leftarrow \mathrm{XX} \cdot \operatorname{atan}(\mathrm{XX}) \\
& \mathrm{F} \leftarrow \mathrm{YY} \cdot \operatorname{atan}(\mathrm{YY}) \\
& \mathrm{F}_{12} \leftarrow \mathrm{~A} \cdot(\mathrm{~B}+\mathrm{C}+\mathrm{D}-\mathrm{E}-\mathrm{F})
\end{aligned}
$$

Ex: $\quad$ F12_parallel_rectangles $(0.5,1,1)=0.117$
2. View factors for coaxial, parallel disks: (See the fig. above for nomenclature)

Define:

$$
\begin{gathered}
R_{i}=\frac{r_{i}}{L} \quad R_{j}=\frac{r_{j}}{L} \quad S\left(R_{i}, R_{j}\right):=1+\frac{1+R_{j}^{2}}{R_{i}^{2}} \\
F_{i j}\left(R_{i}, R_{j}\right):=\frac{1}{2} \cdot\left[S\left(R_{i}, R_{j}\right)-\left[S\left(R_{i}, R_{j}\right)^{2}-4 \cdot\left(\frac{R_{j}}{R_{i}}\right)^{2}\right]^{\frac{1}{2}}\right] \quad \text {..View factor for coaxial parallel disks }
\end{gathered}
$$

## Now, to plot the graphs:

$$
R_{\mathrm{i}}:=10,9.9 .0 .1 \quad \text {...define a range variable }
$$



We write the Mathcad Function for F12 for coaxial, parallel disks as follows:

## Here,

Inputs are: $\mathbf{r}_{\mathrm{i}}, \mathrm{r}_{\mathrm{j}}$ and L .

## Output is: F12.

$$
\text { F12_parallel_disks }\left(\mathrm{r}_{\mathrm{i}}, \mathrm{r}_{\mathrm{j}}, \mathrm{~L}\right):=\left\lvert\, \begin{aligned}
& R_{\mathrm{i}} \leftarrow \frac{\mathrm{r}_{\mathrm{i}}}{\mathrm{~L}} \\
& \mathrm{R}_{\mathrm{j}} \leftarrow \frac{\mathrm{r}_{\mathrm{j}}}{\mathrm{~L}} \\
& \mathrm{~S} \leftarrow 1+\frac{1+\mathrm{R}_{\mathrm{j}}^{2}}{\mathrm{R}_{\mathrm{i}}^{2}} \\
& \mathrm{~F} 12 \leftarrow \frac{1}{2} \cdot\left[\mathrm{~S}-\left[\mathrm{S}^{2}-4 \cdot\left(\frac{\mathrm{r}_{\mathrm{j}}}{\mathrm{r}_{\mathrm{i}}}\right)^{2}\right]^{\frac{1}{2}}\right]
\end{aligned}\right.
$$

Ex: F12_paralle1_disks $(0.5,0.6,1)=0.232$

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3. View factors for perpendicular rectangles with a common edge: (See the fig. above for nomenclature)
Define:

$$
\begin{aligned}
& H=\frac{Z}{X} \quad W=\frac{Y}{X} \quad A(W):=\frac{1}{\pi \cdot W} \quad B(W):=W \cdot \operatorname{atan}\left(\frac{1}{W}\right) \\
& C(H):=H \cdot \operatorname{atan}\left(\frac{1}{H}\right) \quad D(H, W):=\left(H^{2}+W^{2}\right)^{\frac{1}{2}} \cdot \operatorname{atan}\left[\frac{1}{\left(H^{2}+W^{2}\right)^{\frac{1}{2}}}\right] \\
& E(H, W):=\frac{\left(1+W^{2}\right) \cdot\left(1+H^{2}\right)}{\left(1+W^{2}+H^{2}\right)} \cdot\left[\frac{W^{2} \cdot\left(1+W^{2}+H^{2}\right)}{\left(1+W^{2}\right) \cdot\left(W^{2}+H^{2}\right)}\right]^{W^{2}} \cdot\left[\frac{H^{2} \cdot\left(1+H^{2}+W^{2}\right)}{\left(1+H^{2}\right) \cdot\left(H^{2}+W^{2}\right)}\right]^{H^{2}} \\
& F_{i j}(H, W):=A(W) \cdot\left(B(W)+C(H)-D(H, W)+\frac{1}{4} \cdot \ln (E(H, W))\right)
\end{aligned}
$$

...View factor for coaxial perpendicular rectangles with a comon edge

## Now, to plot the graphs:

$$
\mathrm{H}:=0.1,0.2 . .10 \quad \text {.... define a range variable }
$$



We write the Mathcad Function for F12 for perpendicular rectangles with a common edge as follows:

## Here,

## Inputs are: $\mathrm{X}, \mathrm{Y}$ and Z .

Output is: F12.

$$
\text { F12_perpendicular_rectangles }(X, Y, Z):=\left\lvert\, \begin{aligned}
& H \leftarrow \frac{Z}{X} \\
& W \leftarrow \frac{Y}{X} \\
& A \leftarrow \frac{1}{\pi \cdot W} \\
& B \leftarrow W \cdot \operatorname{atan}\left(\frac{1}{W}\right) \\
& C \leftarrow H \cdot \operatorname{atan}\left(\frac{1}{H}\right) \\
& \\
& D \leftarrow\left(H^{2}+W^{2}\right)^{\frac{1}{2}} \cdot \operatorname{atan}\left[\frac{1}{\frac{1}{2}}\right. \\
& \left.\left(H^{2}+W^{2}\right)^{2}\right] \\
& E \leftarrow \frac{\left(1+W^{2}\right) \cdot\left(1+H^{2}\right)}{\left(1+W^{2}+H^{2}\right)} \cdot\left[\frac{W^{2} \cdot\left(1+W^{2}+H^{2}\right)}{\left(1+W^{2}\right) \cdot\left(W^{2}+H^{2}\right)}\right]^{W^{2}} \cdot\left[\frac{H^{2} \cdot\left(1+H^{2}+W^{2}\right)}{\left(1+H^{2}\right) \cdot\left(H^{2}+W^{2}\right)}\right]^{H^{2}} \\
& F 12 \leftarrow A \cdot\left(B+C-D+\frac{1}{4} \cdot \ln (E)\right)
\end{aligned}\right.
$$

Ex:

```
F12_perpendicular_rectangles}(5,5,5)=0.
```

Prob.5B.2. Write Mathcad Functions for view factors of following 2_D geometries:

1) Parallel plates with mid-lines connected by perpendicular line
2) Inclined plates of equal width and with a common edge
3) Perpendicular plates with a common edge
4) Three sided enclosure
5) Infinite plane and row of cylinders

Geometries and view factor relations, given at the beginning of the chapter, are reproduced below: (Ref: Cengel)

Software Solutions to Problems on Heat Transfer: Radiation Heat Transfer - Part I

View factors (or, Shape factor, configuration factor) and Radiation energy exchange between Black bodies:

View factors for few two-dimensional geometries, infinitely long:

| Geometry | Relation |
| :---: | :---: |
| Parallel plates with midlines connected by perpendicular line | $\begin{gathered} W_{i}=w_{i} / L \text { and } W_{j}=w_{j} / L \\ F_{i \rightarrow j}=\frac{\left.\left[\left(W_{i}+W_{j}\right)^{2}+4\right]^{1 / 2}-\left(W_{j}-W_{i}\right)^{2}+4\right]^{1 / 2}}{2 W_{i}} \end{gathered}$ |
| Inclined plates of equal width and with a common edge | $F_{i \rightarrow j}=1-\sin \frac{1}{2} \alpha$ |
| Perpendicular plates with a common edge | $F_{i \rightarrow j}=\frac{1}{2}\left\{1+\frac{w_{j}}{w_{i}}-\left[1+\left(\frac{w_{j}}{w_{i}}\right)^{2}\right]^{1 / 2}\right\}$ |



## Mathcad Functions:

(a) For parallel plates with mid-lines connected by perpendicular:
$\mathrm{F}_{\mathrm{ij} \_ \text {paralle1_plates }\left(\mathrm{w}_{\mathrm{i}}, \mathrm{w}_{\mathrm{j}}, \mathrm{L}\right):=} \begin{aligned} & \mathrm{W}_{\mathrm{i}} \leftarrow \frac{\mathrm{w}_{\mathrm{i}}}{\mathrm{L}} \\ & \mathrm{W}_{\mathrm{j}} \leftarrow \frac{\mathrm{w}_{\mathrm{j}}}{\mathrm{L}} \\ & \\ & \mathrm{F}_{\mathrm{ij}} \leftarrow \frac{\left[\left(\mathrm{W}_{\mathrm{i}}+\mathrm{W}_{\mathrm{j}}\right)^{2}+4\right]^{\frac{1}{2}}-\left[\left(\mathrm{W}_{\mathrm{j}}-\mathrm{W}_{\mathrm{i}}\right)^{2}+4\right]^{\frac{1}{2}}}{2 \cdot \mathrm{~W}_{\mathrm{i}}}\end{aligned}$

Example:
$\mathrm{F}_{\mathrm{ij} \text { _parallel_plates }}(0.25,0.5,0.15)=0.835$

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(b) For inclined parallel plates of equal width and with a common edge:

$$
F_{i j \_i n c l i n e d \_p l a t e s ~}(\alpha):=1-\sin \left(\frac{\alpha \cdot \pi}{180}\right) \quad \text { Note: } \alpha \text { in degrees }
$$

Example: If the angle between plates is 30 deg.:

$$
\mathrm{F}_{\mathrm{ij} \text { _inclined_plates }}(30)=0.5
$$

(c) For perpendicular plates with a common edge:

$$
\mathrm{F}_{\mathrm{ij} \_ \text {perpendicular_plates }}\left(\mathrm{w}_{\mathrm{i}}, \mathrm{w}_{\mathrm{j}}\right):=\frac{1}{2} \cdot\left[1+\frac{\mathrm{w}_{\mathrm{j}}}{\mathrm{w}_{\mathrm{i}}}-\left[1+\left(\frac{\mathrm{w}_{\mathrm{j}}}{w_{i}}\right)^{2}\right]^{\frac{1}{2}}\right]
$$

Example: when $w_{i}=0.5 \mathrm{~m}, w_{\mathrm{j}}=0.5 \mathrm{~m}$ :

$$
\mathrm{F}_{\mathrm{ij} \text { _perpendicular_plates }}(0.5,0.5)=0.293
$$

(d) For three sided enclosure:

$$
F_{i j} \text { three_sided_enclosure }\left(w_{i}, w_{j}, w_{k}\right):=\frac{w_{i}+w_{j}-w_{k}}{2 \cdot w_{i}}
$$

Example:
$F_{i j}$ _three_sided_enclosure $(0.7,0.7,0.5)=0.643$
(e) For infinite plane and row of cylinders:
$F_{i j}$ infinite_plane_to_row_of_cyl( $\left.D, s\right):=1-\left[1-\left(\frac{D}{s}\right)^{2}\right]^{\frac{1}{2}}+\frac{D}{s} \cdot \operatorname{atan}\left(\frac{s^{2}-D^{2}}{D^{2}}\right)^{\frac{1}{2}}$
Example: If $\mathrm{D}=1 \mathrm{~cm}, \mathrm{~s}=5 \mathrm{~cm}$ :

$$
\mathrm{F}_{\mathrm{ij}} \text { infinite_plane_to_row_of_cyl }(0.01,0.05)=0.268
$$

Prob.5B.2. Find out the net heat transferred between two circular disks 1 and 2 , oriented one above the other, parallel to each other on the same centre line, as shown in Fig. Disk 1 has a radius of 0.5 m and is maintained at 1000 K , and disk 2 has a radius of 0.6 m and is maintained at 600 K . Assume both the disks to be black surfaces.


Fig. Prob.5B. 2

## Mathcad Solution:

Data:

$$
\begin{array}{ll}
\mathrm{r}_{\mathrm{i}}:=0.5 & \mathrm{~m} \ldots \text {...radius of disk } 1 \\
\mathrm{r}_{\mathrm{j}}:=0.6 & \mathrm{~m} \ldots \text {...radius of disk } 2 \\
\mathrm{~L}:=1 & \mathrm{~m} \ldots \text {.... distance between disk } 1 \text { and disk } 2
\end{array}
$$

$\mathrm{T}_{1}:=1000 \quad \mathrm{~K} . .$. temp. of disk 1
$T_{2}:=600 \quad \mathrm{~K}$. temp. of disk 2
$\sigma:=5.67 \cdot 10^{-8} \mathrm{~W} / \mathrm{m}^{2} . \mathrm{K} \ldots$ Stefan-Boltzmann const.
$A_{1}:=\pi \cdot r_{i}^{2} \quad$ i.e. $\quad A_{1}=0.785 \quad \mathrm{~m}^{2} \ldots$ area of disk 1
$A_{2}:=\pi \cdot r_{j}^{2} \quad$ i.e. $\quad A_{2}=1.131 \quad \mathrm{~m}^{2} \ldots$ area of disk 2
This is the case of heat transfer between two black surfaces. So, we use eqn. (13.40):

$$
\begin{equation*}
Q_{n e t}=A_{1} \cdot F_{12} \cdot \sigma \cdot\left(T_{1}{ }^{4}-T_{2}{ }^{4}\right)=A_{2} \cdot F_{21} \cdot \sigma \cdot\left(T_{1}{ }^{4}-T_{2}{ }^{4}\right) \tag{13.40}
\end{equation*}
$$

So, the problem reduces to calculating the view factor $\mathrm{F}_{12}$ or $\mathrm{F}_{21}$.
We can easily find out F12 using the graphs given earlier. However, we can determine F12 analytically more accurately with Mathcad Functions written above.

## We use the Mathcad Function already written above:

$$
\text { F12_parallel_disks }\left(\mathrm{r}_{\mathrm{i}}, \mathrm{r}_{\mathrm{j}}, \mathrm{~L}\right)=0.232 \quad \text {...View factor for coaxial parallel disks }
$$

## (Note: verify this result by referring to the graph.)

i.e. $\quad F_{12}:=0.232$

Therefore, net heat transfer between disks 1 and 2:

$$
Q_{n e t}:=A_{1} \cdot F_{12} \cdot \sigma \cdot\left(T_{1}^{4}-T_{2}^{4}\right) \quad W \text {...from eqn. (13.40) }
$$

i.e. $Q_{\text {net }}=8.992 \times 10^{3} \quad W$....Ans.

Prob.5B.3. Find out the net heat transferred between two aligned parallel rectangles, as shown in Fig. Surface 1 is maintained at 500 K , and surface 2 is maintained at 1000 K . Assume both the surfaces to be black surfaces.

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Fig.Prob.5B. 3

## Mathcad Solution:

## Data:

$$
\begin{aligned}
& \mathrm{X}:=0.5 \quad \mathrm{~m} \ldots . \text { horizontal } \mathrm{x} \text {-dimension of surfaces } 1 \text { and } 2 \\
& \mathrm{Y}:=1.0 \quad \mathrm{~m} \text {.....horizontal } \mathrm{y} \text {-dimension of surfaces } 1 \text { and } 2 \\
& \mathrm{~L}:=1.0 \quad \text {...vertical distane between surfaces } 1 \text { and } 2 \\
& \mathrm{~T}_{1}:=500 \quad \text { K....temp. of surface } 1 \\
& \mathrm{~T}_{2}:=1000 \quad \mathrm{~K} . . \text { temp. of surface } 2 \\
& \sigma:=5.67 \cdot 10^{-8} \quad \mathrm{~W} / \mathrm{m}^{2} . \mathrm{K} \ldots \text { Stefan-Boltzmann const. }
\end{aligned}
$$

Calculations:

$$
\begin{array}{llll}
A_{1}:=X \cdot Y & \text { i.e. } & A_{1}=0.5 & \mathrm{~m}^{2} \ldots \text { area of surface } 1 \\
A_{2}:=X \cdot Y & \text { i.e. } & A_{2}=0.5 & \mathrm{~m}^{2} \ldots \text { area of surface } 2
\end{array}
$$

This is the case of heat transfer between two black surfaces. So, we use eqn. (13.40), viz.

$$
\begin{equation*}
Q_{\text {net }}=A_{1} \cdot F_{12} \cdot \sigma \cdot\left(T_{1}^{4}-T_{2}^{4}\right)=A_{2} \cdot F_{21} \cdot \sigma \cdot\left(T_{1}^{4}-T_{2}^{4}\right) \tag{13.40}
\end{equation*}
$$

So, the problem reduces to calculating the view factor $\mathrm{F}_{12}$ or $\mathrm{F}_{21}$. We can find out $\mathrm{F}_{12}$ using the graphs. However, we can determine $\mathrm{F}_{12}$ analytically more accurately with Mathcad Function using the view factor relation given in Table for aligned parallel rectangles.

We use the the Mathcad Function already written above for view factor as follows:

```
F12_parallel_rectangles(X,Y,L)}=0.11
```

(Note: verify this result from the graph.)

$$
\text { i.e. } \quad F_{12}:=0.117
$$

## Therefore, net heat transfer between surfaces 1 and 2:

$$
\begin{aligned}
& Q_{\text {net }}:=A_{1} \cdot F_{12} \cdot \sigma \cdot\left(T_{1}^{4}-T_{2}^{4}\right) \quad W \ldots \text { from eqn. (13.40) } \\
& \text { i.e. } \quad Q_{\text {net }}=-3.11 \times 10^{3} \quad \text { W....Ans. }
\end{aligned}
$$

Note: -ve sign indicates that net heat transfer is to the surface 1 , since surface 2 is at a higher temperature than surface 1 .

Prob.5B.4. Find out the net heat transferred between the areas A1 and A2 shown in Fig. Area 1 is maintained at 900 K , and area 2 is maintained at 400 K . Assume both the surfaces to be black.


Fig.Prob.5B. 4

## Mathcad Solution:

This is the case of heat transfer between two black surfaces. So, we use eqn. (13.40), viz.

$$
\begin{equation*}
Q_{\text {net }}=A_{1} \cdot F_{12} \cdot \sigma \cdot\left(T_{1}^{4}-T_{2}^{4}\right)=A_{2} \cdot F_{21} \cdot \sigma \cdot\left(T_{1}^{4}-T_{2}^{4}\right) \tag{13.40}
\end{equation*}
$$

So, the problem reduces to calculating the view factor $F_{12}$ or $F_{21}$.
We see that to calculate $F_{12}$ for areas $A_{1}$ and $A_{2}$ as oriented in the Fig., we do not readily have an analytical relation or a graph.
Let us denote the combined areas $\left(A_{1}+A_{3}\right)$ by $A_{5}$ and $\left(A_{2}+A_{4}\right)$ by $A_{8}$.
Then, we see that $A_{5}$ and $A_{8}$ are perpendicular rectangles which have a common edge, and we have graphs or analytical relation for the view factor for such an orientation.
Then, we resort to view factor algebra, as follows:

Remember that by definition, view factor $F_{12}$ is the fraction of radiant energy emitted by surface 1 which falls directly on surface 2 . Looking at the fig., we can say that fraction of energy leaving $A_{1}$ and falling on $A_{2}$ is equal to the fraction falling on $A_{8}$ minus the fraction falling on $\mathrm{A}_{4}$.
i.e. $\quad F_{12}=F_{16}-F_{14} \quad$... by definition of view factor
i.e. $\quad F_{12}=F_{61} \cdot \frac{A_{6}}{A_{1}}-F_{41} \cdot \frac{A_{4}}{A_{1}} \quad \begin{aligned} & \text { A. since by reciprocity relation, } A_{1} \cdot F_{16}=A_{6}\end{aligned} F_{61}$, and $A_{1} \cdot F_{14}=A_{4} \cdot F_{41}$.
i.e. $\quad F_{12}=\frac{A_{6}}{A_{1}} \cdot\left(F_{65}-F_{63}\right)-\frac{A_{4}}{A_{1}} \cdot\left(F_{45}-F_{43}\right)$
eqn.(A).... using the definition of view factor, as done in first step above


Now, observe that view factors $\mathrm{F}_{65}, \mathrm{~F}_{63}, \mathrm{~F}_{45}$ and $\mathrm{F}_{43}$ refer to perpendicular rectangles with a comon edge, and can be readily obtained from the graphs, or by analytical relation given in Table.

We have already written Mathcad Function for view factor relation for perpendicular rectangles with a common edge (See Prob.5B.1).

To find $\mathrm{F}_{\underline{85}}$ :

$$
X:=5 \quad Y:=5 \quad Z:=5 \quad \text {....See. Fig. }
$$

F12_perpendicular_rectangles $(\mathrm{X}, \mathrm{Y}, \mathrm{Z})=0.2$....From the Mathcad Function

```
i.e. }\mp@subsup{F}{65}{}:=0.2 ...view factor from area A A to A5
```

Note: This value can be verified from graphs also.
To find $\mathrm{F}_{\underline{\text { 63 }}}$ :

$$
X:=5 \quad Y:=5 \quad Z:=3 \quad \text { See. Fig. }
$$

F 12 _perpendicular_rectangles $(\mathrm{X}, \mathrm{Y}, \mathrm{Z})=0.161 \quad$....From the Mathcad Function

$$
\text { i.e. } \quad F_{63}:=0.161 \ldots \text { view factor from area } A_{8} \text { to } A_{3}
$$

## Note: This value also can be verified from graphs.

## To find $F_{45:}$

$$
X:=5 \quad Y:=3 \quad Z:=5 \quad \text {..See Fig. }
$$

F12_perpendicular_rectangles $(\mathrm{X}, \mathrm{Y}, \mathrm{Z})=0.269$....From the Mathcad Function
i.e. $\quad F_{45}:=0.269$...view factor from area $\mathrm{A}_{4}$ to $\mathrm{A}_{5}$

Note: This value also can be verified fromgraphs.

To find $\mathrm{F}_{43-}$

$$
X:=5 \quad Y:=3 \quad Z:=3 \quad \text {. See Fig. }
$$

F12_perpendicular_rectangles $(X, Y, Z)=0.231$....From the Mathcad Function
i.e. $\quad F_{43}:=0.231 \quad$...view factor from area $A_{4}$ to $A_{3}$

Note: This value also can be verified fromgraphs.

## Areas:

From Fig. we have:

$$
\begin{array}{lll}
A_{1}:=10 & A_{2}:=10 & A_{3}:=15
\end{array} A_{4}:=15 \quad \mathrm{~m}^{2} \quad ~\left(A_{5}:=25 \quad A_{6}:=25 \quad \mathrm{~m}^{2} \quad . ~ \$\right.
$$

Then, from eqn.(A):

$$
F_{12}:=\left[\frac{A_{6}}{A_{1}} \cdot\left(F_{65}-F_{63}\right)-\frac{A_{4}}{A_{1}} \cdot\left(F_{45}-F_{43}\right)\right]
$$

i.e. $F_{12}=0.041$....view factor from $A_{1}$ to $A_{2} \ldots$...Ans.

Note: $F_{21}$ can be calculated, if required, by reciprocity relation, i.e. $A_{1} \cdot F_{12}=A_{2} \cdot F_{21}$

## Therefore, net heat transfer between surfaces 1 and 2:

$$
Q_{\text {net }}=A_{1} \cdot F_{12} \cdot \sigma \cdot\left(T_{1}{ }^{4}-T_{2}^{4}\right) \quad W \text {...from eqn. (13.40) }
$$

Here, we have:

$$
\begin{array}{ll}
T_{1}:=900 & K . . . \text { temp. of surface } A_{1} \\
T_{2}:=400 & K . . \text { temp. of surface } A_{2} \\
\sigma:=5.67 \cdot 10^{-8} & \mathrm{~W} / \mathrm{m}^{2} . \mathrm{K} . . \text { Stefan-Boltzmann const. }
\end{array}
$$

Therefore,

$$
\begin{aligned}
& Q_{\text {net }}=A_{1} \cdot F_{12} \cdot \sigma \cdot\left(T_{1}{ }^{4}-I_{2}{ }^{4}\right) \\
\text { i.e. } \quad Q_{\text {net }} & =1.448 \times 10^{4} \quad \text { W....Ans. }
\end{aligned}
$$

Prob.5B.5. Find out the relevant view factors for the geometries shown in Fig. below:
a) a long tube with cross-section of an equilateral triangle
b) a blackbody completely enclosed by another black body
c) diagonal partition inside a long square duct
d) sphere of diameter d inside a cubical box of sides, $\mathrm{L}=\mathrm{d}$
e) hemispherical surface closed by a plane surface, and
f) the end and surface of a circular cylinder whose length is equal to diameter.

(a) Equilateral triangle

(b) A blackbody completely enclosed by another black body

(c) Diagonal partition in a long square duct

(d) Sphere inside a cubical box

(e) Hemispherical bowl

(f) Cylinder with length $=$ diameter

Fig.Prob.5B. 5

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General principle in solving these problems is to invoke: Summation rule, reciprocity theorem, inspection of geometry for symmetry, and of course, remembering the definition of view factor:

## (a) Long tube wih cross-section of equilateral triangle: See Fig. (a)

For surface 1: $\quad F_{11}+F_{12}+F_{13}=1 \quad$....Summation rule But, $\quad F_{11}=0 \quad$...since surface 1 is flat and can not see itself. Therefore, $\quad F_{12}+F_{13}=1$

Now, by inspection of geometry, we find that surfaces 2 and 3 are located symmetrically w.r.t. surface 1 , since it is an equilateral triangle. Therefore, radiation from surface 1 is divided equally between surfaces 2 and 3.

$$
\text { i.e. } \quad F_{12}=F_{13}=0.5
$$

Similarly, for surface 2 , we write:

$$
\begin{array}{ll} 
& F_{21}+F_{23}=1 \\
\text { i.e. } & F_{23}=1-F_{21} \\
\text { But, } & F_{21}=\frac{A_{1}}{A_{2}} \cdot F_{12}=F_{12} \quad \ldots \text { since } A_{1}=A_{2} \\
\text { Therefore, } & F_{23}=1-F_{12}=0.5
\end{array}
$$

Similarly for surface 3 .
(b) Black body enclosed inside a black enlosure: See Fig. (b)

For surface 1: $\quad F_{11}+F_{12}=1 \quad$...by Summation rule

$$
\begin{aligned}
\text { and, } & A_{1} \cdot F_{12}=A_{2} \cdot F_{21} \quad \ldots \text { by reciprocity } \\
\text { i.e. } & F_{12}=\frac{A_{2}}{A_{1}} \cdot F_{21} \\
\text { Now, } & F_{11}=1-F_{12} \\
\text { i.e. } & F_{11}=1-\frac{A_{2}}{A_{1}} \cdot F_{21}
\end{aligned}
$$

But, $\quad F_{21}=1$ since all the energy radiated by surface 2 is directly intercepted by surface 1 .

Therefore, $\quad F_{11}=1-\frac{A_{2}}{A_{1}}$
(c) Diagonal partition within a long square duct: See Fig. (c)

For surface 1: $\quad F_{11}+F_{12}+F_{13}=1 \quad$...by Summation rule
But, $\quad F_{11}=0 \quad$..since surface 1 is flat and can not 'see' itself.
Therefore, $\quad F_{12}+F_{13}=1$

By symmetry: $\quad F_{3}=F_{12}=0.5$..since radiation emitted by sufface 1 is divided equally between surfaces 2 and 3

By reciprocity: $\quad F_{21}=\frac{A_{1}}{A_{2}} \cdot F_{12}$
i.e. $\quad F_{21}=\frac{\sqrt{2} \cdot \mathrm{~L}}{\mathrm{~L}} \cdot 0.5=0.71$
(d) Sphere inside a cubical box: See Fig. (d)

For surface 1: $\quad F_{11}=0 \quad \begin{aligned} & \text { itself. } \\ & \text { ince surface of the sphere is convex and can not 'see' }\end{aligned}$
and, $\quad F_{11}+F_{12}=1 \quad$...by Summation rule

$$
\text { Therefore, } F_{12}=1
$$

By reciprocity: $\quad F_{21}=\frac{A_{1}}{A_{2}} \cdot F_{12}$
i.e. $\quad F_{21}=\frac{\pi \cdot \mathrm{d}^{2}}{6 \cdot \mathrm{~d}^{2}}=\frac{\pi}{6} \quad$ since $\mathrm{L}=\mathrm{d}$
i.e. $\quad F_{21}=0.524$
(e) Hemispherical surface closed by a flat surface: See Fig. (e)

$$
\text { For surface 1: } \quad F_{11}+F_{12}=1 \quad \text {....Summation rule }
$$

Also, $\quad F_{21}=1 \quad$...since surface 2 is flat and can not 'see' itself, and all radiation emitted by surface 2 falls directly on the hemispherical surface 1 .

By reciprocity: $\quad F_{12}=\frac{A_{2}}{A_{1}} \cdot F_{21}$

$$
\text { i.e. } \quad F_{12}=\frac{\pi \cdot \mathrm{r}^{2}}{2 \cdot \pi \cdot \mathrm{r}^{2}} \cdot 1=0.5
$$

Therefore,

$$
F_{11}=1-F_{12}=0.5
$$

(f) End and sides of a circular cylinder ( $\mathrm{L}=\mathrm{d}$ ): See Fig. (f)

From the fig., note that the two end surfaces are denoted by 1 and 3 and the side surface is denoted by 2 .


View factor $\mathrm{F}_{13}$ : Surfaces 1 and 3 can be considered as two concentric parallel disks.
Therefore, $\mathrm{F}_{13}$ can be found out from graphs or by analytical relation given in Table. Let us use the analytical relation:

We have:

$$
\begin{gathered}
R_{i}=\frac{r_{i}}{L} \quad R_{j}=\frac{r_{j}}{L} \quad S\left(R_{i}, R_{j}\right):=1+\frac{1+R_{j}^{2}}{R_{i}^{2}} \\
F_{i j}\left(R_{i}, R_{j}\right):=\frac{1}{2} \cdot\left[S\left(R_{i}, R_{j}\right)-\left[S\left(R_{i}, R_{j}\right)^{2}-4 \cdot\left(\frac{R_{j}}{R_{i}}\right)^{2}\right]^{\frac{\dot{2}}{2}}\right]
\end{gathered}
$$

...View factor for coaxial parallel disks
Now, for a cylinder with $\mathrm{L}=\mathrm{d}$ :

$$
R_{i}:=0.5 \quad R_{j}:=0.5
$$

Therefore, $\mathrm{F}_{\mathrm{ij}}\left(\mathrm{R}_{\mathrm{i}}, \mathrm{R}_{\mathrm{j}}\right)=0.172$
i.e. $\quad F_{13}:=0.172$...view factor from surface 1 to surface 3

For surface 1: $\quad F_{11}+F_{12}+F_{13}=1 \quad$...by Summation rule
But, $\quad F_{11}=0 \quad$...since surface 1 is flat and can not 'see' itself.
Therefore, $\quad F_{12}=1-F_{13}$

$$
\text { i.e. } \quad F_{12}=0.828
$$

By reciprocity: $\quad F_{21}=\frac{A_{1}}{A_{2}} \cdot F_{12}$
i.e. $\quad F_{21}=\frac{\frac{\pi \cdot \mathrm{d}^{2}}{4}}{\pi \cdot \mathrm{~d} \cdot \mathrm{~L}} \cdot 0.828$
i.e. $\quad F_{21}=\frac{\frac{\pi \cdot d^{2}}{4}}{\pi \cdot d^{2}} \cdot 0.828 \quad$. since $L=d$
i.e. $\quad F_{21}=\frac{0.828}{4}=0.207$

Also, by symmetry: $\quad F_{32}=F_{12}=0.828$

$$
\text { and, } \quad F_{23}=F_{21}=0.207
$$

Prob.5B.6. Find out the view factor $\left(\mathrm{F}_{11}\right)$ of a cavity with respect to itself. Hence, find out the view factor $F_{11}$ for the following:
a) a cylindrical cavity of diameter 'd' and depth ' h '
b) a conical cavity of diameter ' $d$ ' and depth ' $h$ '
c) a hemispherical bowl of diameter 'd'


Fig.Prob.5B. 6

## Mathcad Solution:

View factor of a general cavity w.r.t. itself: See Fig.(a) above.

We desire to find $\mathrm{F}_{11}$. It is obvious from the fig. that part of the radiation emitted by the cavity surface 1 , falls on itself and therefore, $\mathrm{F}_{11}$ exists.

Close the opening (or mouth) of the cavity by a hypothetical flat surface 2 . Then, surfaces 1 and 2 together form an enclosure. We can write:

For surface 1: $F_{11}+F_{12}=1 \quad$...(a).... by Summation rule

For surface 2: $\quad F_{21}+F_{22}=1 \quad$.... by Summation rule
But, $\quad F_{22}=0 \quad$..since surface 2 is flat and can not 'see' itself.
Therefore, $\quad F_{21}=1$
Further, $\quad A_{1} \cdot F_{12}=A_{2} \cdot F_{21} \quad$....by reciprocity
i.e. $\quad F_{12}=\frac{A_{2}}{A_{1}}$

Now,

$$
F_{11}=1-F_{12} \quad \text {...from eqn.(a) }
$$

i.e. $\quad F_{11}=1-\frac{A_{2}}{A_{1}} \quad \ldots$ eqn.(b)

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Eqn.(b) is an important result, since it gives the shape factor of any general cavity w.r.t. itself.

Now, this result will be applied to following specific cavities:
(a) $\mathrm{F}_{11}$ for a cylindrical cavity of diameter ' $d$ ' and depth ' $h$ ': See Fig.(b)

$$
\text { We have: } \quad F_{11}=1-\frac{A_{2}}{A_{1}}
$$

i.e. $\quad F_{11}=1-\frac{\frac{\pi \cdot \mathrm{d}^{2}}{4}}{\frac{\pi \cdot \mathrm{~d}^{2}}{4}+\pi \cdot \mathrm{d} \cdot \mathrm{h}}$ circular surface and the cylindrical side surfaces
i.e. $\quad F_{11}=\frac{4 \cdot h}{4 \cdot h+d}$
....Ans.
(b) $\mathrm{F}_{11}$ for a conical cavity of diameter ' $d$ ' and depth ' h ': See Fig. (c)

We have: $\quad F_{11}=1-\frac{A_{2}}{A_{1}}$
i.e. $\quad F_{11}=1-\frac{\frac{\pi \cdot \mathrm{d}^{2}}{4}}{\frac{\pi \cdot \mathrm{~d} \cdot \mathrm{~L}}{2}} \quad$...where L is the slant height of the cone
i.e. $\quad F_{11}=1-\frac{d}{2 \cdot L}$
i.e. $\quad F_{11}=1-\sin (\alpha) \quad$... where $\alpha$ is the half-vertex angle of the cone.

Alternatively: To get $F_{11}$ in terms of depth ' $h$ ', we can write:

$$
\begin{aligned}
& \text { We have: } F_{11}=1-\frac{d}{2 \cdot L} \\
& \text { i.e. } F_{11}=1-\frac{d}{2 \cdot \sqrt{h^{2}+\frac{d^{2}}{4}}} \\
& \text { i.e. } F_{11}=1-\frac{d}{\sqrt{4 \cdot h^{2}+d^{2}}}
\end{aligned}
$$

(c) $F_{11}$ for a hemispherical bowl of diameter ' $d$ ': See Fig.(d)

$$
\begin{aligned}
& \text { We have: } \quad F_{11}=1-\frac{A_{2}}{A_{1}} \\
& \text { i.e. } \quad F_{11}=1-\frac{\pi \cdot d^{2}}{\frac{\pi \cdot d^{2}}{2}} \\
& \text { i.e. } \quad F_{11}=1-\frac{1}{2} \\
& \text { i.e. } \quad F_{11}=0.5 \quad \text {...Ans. }
\end{aligned}
$$

This result means that for any hemispherical cavity, half of the radiation emitted by the surface 1 falls on itself; it also means that the remaining half falls on the closing surface 2 .

Prob.5B.7. A 70 mm thick metal plate has a hole of 30 mm dia drilled through it. If the plate is held at 300 C and the surrounding is at 27 C , find the energy escaping to the surroundings from both ends of the hole. Assume the metal and surroundings as black bodies.

## Mathcad Solution:



Fig.Prob.5B. 7

Data:

| $\mathrm{r}_{\mathrm{i}}:=0.015 \mathrm{~m}$ | $\mathrm{r}_{\mathrm{j}}:=0.015 \mathrm{~m} \quad \mathrm{~L}:=0.07 \mathrm{~m}$ |
| :--- | :--- |
| $\mathrm{~A} 1:=\pi \cdot 2 \cdot \mathrm{r}_{\mathrm{i}} \mathrm{L}$ | i.e. $\mathrm{A} 1=6.597 \times 10^{-3} \quad \mathrm{~m}^{\wedge} 2$ |
| $\mathrm{~A} 2:=\pi \cdot \mathrm{r}_{\mathrm{i}}^{2}$ | i.e. $\mathrm{A} 2=7.069 \times 10^{-4} \quad \mathrm{~m}^{\wedge} 2$ |
| $\mathrm{~A} 3:=\pi \cdot \mathrm{r}_{\mathrm{j}}^{2}$ | i.e. $\mathrm{A} 3=7.069 \times 10^{-4} \quad \mathrm{~m}^{\wedge} 2$ |

$\mathrm{T} 1:=300+273 \quad \mathrm{~K} \quad \mathrm{~T}_{\mathrm{a}}:=27+273 \quad \mathrm{~K}$
$\sigma:=5.67 \cdot 10^{-8} \mathrm{~W} / \mathrm{m}^{\wedge} 2 . \mathrm{K}^{\wedge} 4 \ldots . .$. Stefan Boltzmann constant


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## Calculations:

$F_{22}+F_{23}+F_{21}=1 \quad$..eqn (a)....by summation rule
$F_{22}:=0 \quad$...for a flat surface 2

F23 is the view factor from surface 2 to surface 3, i.e. between two coaxial disks
Let us apply the Mathcad Function already written to calculate F23.

$$
\begin{aligned}
& \mathrm{F}_{23}:=\mathrm{F} 12 \text { _parallel_disks }\left(\mathrm{r}_{\mathrm{i}}, \mathrm{r}_{\mathrm{j}}, \mathrm{~L}\right) \quad \text {...using Mathcad Function } \\
& \text { i.e. } \mathrm{F}_{23}=0.042
\end{aligned}
$$

Then, from eqn. (a): $\quad F_{21}:=1-\left(F_{22}+F_{23}\right)$

$$
\text { i.e. } \quad F_{21}=0.958
$$

Now, applying reciprocal rule between surfaces 1 and 2:

$$
\begin{gathered}
A 1 \cdot F_{12}=A 2 \cdot F_{21} \quad \text {...by reciprocity } \\
\text { Therefore: } \quad F_{12}:=\frac{A 2 \cdot F_{21}}{A 1} \\
\text { i.e. } \quad F_{12}=0.103
\end{gathered}
$$

Now, by symmetry: $\quad F_{13}:=F_{12}$
Therefore: Heat lost by both ends:

$$
\begin{aligned}
& \mathrm{Q}_{\text {tot }}:=2 \cdot \mathrm{~A} 1 \cdot \mathrm{~F}_{12} \cdot \sigma \cdot\left(\mathrm{T1}^{4}-\mathrm{I}_{\mathrm{a}}^{4}\right) \quad \mathrm{W} \\
& \text { i.e. } \quad \mathrm{Q}_{\text {tot }}=7.655 \quad \mathrm{~W} . \ldots . \mathrm{Ans} .
\end{aligned}
$$

## EES Functions to determine View Factors:

Let us write EES Functions to find out View Factors for three practically important geometries:
a) Aligned, parallel rectangles,
b) Coaxial, parallel disks, and
c) Perpendicular rectangles with a common edge
a) Aligned, parallel rectangles:


## \$UnitSystem SI Pa J Radians

FUNCTION F12_parallel_rectangles(X, Y, L)
\{Gives View factor for coaxial, parallel rectangles separated by a distance L
Note: Trigonometric functions are used with Radians; so, set the angles to Radians in Unit Systems\}

$$
\begin{aligned}
& \mathrm{XX}:=\mathrm{X} / \mathrm{L} \\
& \mathrm{YY}:=\mathrm{Y} / \mathrm{L} \\
& \mathrm{~A}:=2 /\left(\mathrm{pi}^{\star} \mathrm{XX}{ }^{\star} \mathrm{YY}\right) \\
& \mathrm{B}:=\ln \left(\mathrm{sqrt}\left(\left(1+\mathrm{XX}^{\wedge} 2\right)^{\star}(1+\mathrm{YY} \wedge 2) /\left(1+\mathrm{XX}^{\wedge} 2+\mathrm{YY}^{\wedge} 2\right)\right)\right) \\
& \mathrm{C}:=\mathrm{XX}^{\star}\left(1+\mathrm{YY}^{\wedge} 2\right)^{\wedge}(1 / 2)^{\star} \arctan \left(\mathrm{XX} /(1+\mathrm{YY} \wedge 2)^{\wedge}(1 / 2)\right) \\
& \mathrm{D}:=\mathrm{YY}^{\star}\left(1+\mathrm{XX}^{\wedge} 2\right)^{\wedge}(1 / 2)^{\star} \arctan \left(\mathrm{YY} /\left(1+\mathrm{XX}^{\wedge} 2\right)^{\wedge}(1 / 2)\right) \\
& \mathrm{E}:=\mathrm{XX}^{\star} \arctan (\mathrm{XX}) \\
& \mathrm{F}:=\mathrm{YY}^{\star} \arctan (\mathrm{YY})
\end{aligned}
$$

$$
\text { F12_parallel_rectangles }:=A^{*}(\mathrm{~B}+\mathrm{C}+\mathrm{D}-\mathrm{E}-\mathrm{F})
$$

END
$\qquad$ _-"
b) Coaxial, parallel disks:


FUNCTION F12_parallel_disks(r_1, r_2, L)
\{Gives View factor for coaxial, parallel disks separated by a distance L\}

RR_1 := r_1 / L
RR_2 := r_2 / L
$S:=1+\left(1+R R \_2^{\wedge} 2\right) / R R \_1^{\wedge} 2$

F12_parallel_disks $:=(1 / 2)^{\star}\left(S-\left(S^{\wedge} 2-4^{*}\left(r \_2 / r \_1\right)^{\wedge} 2\right)^{\wedge}(1 / 2)\right)$

END
$\qquad$ -_"


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c) Perpendicular rectangles with a common edge:

\$UnitSystem SI Pa J Radians

FUNCTION F12_perpendicular_rectangles(X, Y, Z)
\{Gives View factor for coaxial, perpendicular rectangles with a common edge
Note: Trigonometric functions are used with Radians; so, set the angles to Radians in Unit Systems\}
$\mathrm{H}:=\mathrm{Z} / \mathrm{X}$
$\mathrm{W}:=\mathrm{Y} / \mathrm{X}$
$\mathrm{A}:=1 /\left(\mathrm{pi}^{*} \mathrm{~W}\right)$
$\mathrm{B}:=\mathrm{W}^{*} \arctan (1 / \mathrm{W})$
$\mathrm{C}:=\mathrm{H}^{*} \arctan (1 / \mathrm{H})$
$\mathrm{D}:=\left(\mathrm{H}^{\wedge} 2+\mathrm{W}^{\wedge} 2\right)^{\wedge}(1 / 2)^{\star} \arctan \left(1 /\left(\mathrm{H}^{\wedge} 2+\mathrm{W}^{\wedge} 2\right)^{\wedge}(1 / 2)\right)$
$\mathrm{E}:=\left(1+\mathrm{W}^{\wedge} 2\right)^{\star}\left(1+\mathrm{H}^{\wedge} 2\right) /\left(1+\mathrm{W}^{\wedge} 2+\mathrm{H}^{\wedge} 2\right)$
$\mathrm{F}:=\left(\mathrm{W}^{\wedge} 2^{\star}\left(1+\mathrm{W}^{\wedge} 2+\mathrm{H}^{\wedge} 2\right) /\left(\left(1+\mathrm{W}^{\wedge} 2\right)^{\star}\left(\mathrm{W}^{\wedge} 2+\mathrm{H}^{\wedge} 2\right)\right)\right)^{\wedge}\left(\mathrm{W}^{\wedge} 2\right)$
$\mathrm{G}:=\left(\mathrm{H}^{\wedge} 2^{*}\left(1+\mathrm{H}^{\wedge} 2+\mathrm{W}^{\wedge} 2\right) /\left(\left(1+\mathrm{H}^{\wedge} 2\right)^{\star}\left(\mathrm{H}^{\wedge} 2+\mathrm{W}^{\wedge} 2\right)\right)\right)^{\wedge}\left(\mathrm{H}^{\wedge} 2\right)$

F12_perpendicular_rectangles $:=A^{*}\left(B+C-D+(1 / 4)^{*} \ln \left(E^{*} F^{*} G\right)\right)$

END
$\qquad$

## Now, let us use these EES Functions to solve some problems:

Prob.5B.8. Consider a truncated cone with top and bottom diameters of 10 and 20 cm and a height of 10 cm . Calculate:
a) the shape factor from the top surface to the sides and also from the side surface to itself.
b) If surface 1 is at 1000 K and surface 2 is at 500 K , find the net exchange of heat between 1 and 2 (Q_12), if all the surfaces are black.
c) Plot $\mathrm{F} \_12$ and $\mathrm{Q} \_12$ for L varying from 0.1 to 1 m


## Fig.Prob.5B. 8

## EES Solution:

Let the top surface be designated as 1 , bottom surface as 2 , and the side surfaces as 3 . See the fig. above.

Then:
$\qquad$
"Data:"
$\mathrm{r} \_1=0.05[\mathrm{~m}]$
$\mathrm{r} \_2=0.1[\mathrm{~m}]$
$\mathrm{L}=0.1[\mathrm{~m}]$

## "We have to find out F13 and F33:"

## "Calculations:"

$A 1=\mathrm{pi}^{\star} \mathrm{r}_{-} 1 \wedge 2$ " $\left[\mathrm{m}^{\wedge} 2\right] \ldots$ area of top surface"
$\mathrm{A} 2=\mathrm{pi}^{*} \mathrm{r}_{-} 2^{\wedge} 2$ " $\left[\mathrm{m}^{\wedge} 2\right] \ldots$ area of bottom surface"
$\mathrm{A} 3=\mathrm{pi}^{*}\left(\mathrm{r}_{-} 1+\mathrm{r}_{-} 2\right)^{*} \operatorname{sqrt}\left(\left(\mathrm{r}_{-} 2-\mathrm{r}_{-} 1\right)^{\wedge} 2+\mathrm{L}^{\wedge} 2\right)$ " $\left[\mathrm{m}^{\wedge} 2\right] \ldots$ area of the side surface"
"Now:
$\mathrm{F} 11+\mathrm{F} 12+\mathrm{F} 13=0 \ldots$. by Summation rule

But, F11 $=0 \ldots$ since it is a flat surface.

We find F12 by the EES Function for parallel disks, written above.

So, we have:
$\mathrm{F} 11=0$

F12 = F12_parallel_disks(r_1, r_2, L) "using the EES Function for parallel disks"
$\mathrm{F} 11+\mathrm{F} 12+\mathrm{F} 13=1 " \ldots$. finds the view factor from top surface to side surface, F13"

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"Now, by reciprocal relation between surfaces 1 and 3:"
$\mathrm{A} 1^{*} \mathrm{~F} 13=\mathrm{A} 3{ }^{*} \mathrm{~F} 31$ ". . finds view factor F31"
"Now, by reciprocal relation between surfaces 1 and 2:"
$\mathrm{A} 1^{*} \mathrm{~F} 12=\mathrm{A} 2{ }^{*} \mathrm{~F} 21$ "...finds view factor from surface 2 to $1, \mathrm{~F} 21$ "
"And, by reciprocal relation between surfaces 2 and 3:"
$\mathrm{F} 22=0$ "...since surface 2 is flat"
"For surface 2, by Summation rule:"
$\mathrm{F} 22+\mathrm{F} 21+\mathrm{F} 23=1{ }^{\text {" }} .$. determines F23"
$\mathrm{A} 2{ }^{*} \mathrm{~F} 23=\mathrm{A} 3{ }^{*} \mathrm{~F} 32$ " $\ldots$.finds view factor from surface 3 to $2, \mathrm{~F} 32$ "
"Now, applying Summation rule to surface 3:"
$\mathrm{F} 33+\mathrm{F} 31+\mathrm{F} 32=1^{"} \ldots$. determines F33"

## Results:

Main F12_parallel_disks
Unit Settings: SI C Pa J mass rad
$\mathrm{A} 1=0.007854\left[\mathrm{~m}^{2}\right]$
$\mathrm{A} 2=0.03142\left[\mathrm{~m}^{2}\right]$
$\mathrm{A} 3=0.05269\left[\mathrm{~m}^{2}\right]$
$\mathrm{F} 11=0$
$\mathrm{F} 12=0.4689$
F13 $=0.5311$
$F 23=0.8828$
$F 31=0.07918$
$\mathrm{F} 21=0.1172$
$\mathrm{F} 22=0$
$\mathrm{F} 32=0.5264$
F33 $=0.3944$

Main F12_parallel_disks
Local variables in Function F12_parallel_disks (1 call, 0.00 sec )
F12 ${ }_{\text {parallel,disks }}=0.4689$
$\mathrm{L}=0.1$ [ m$]$
$\mathrm{RR}_{1}=0.5$
$\mathrm{RR}_{2}=1$
$\mathrm{r}_{1}=0.05[\mathrm{~m}]$
$\mathrm{r}_{2}=0.1[\mathrm{~m}]$
$S=9$

Thus:
Shape factor from the top surface to the sides $=\mathrm{F} 13=0.5311 \ldots$ Ans.

Shape factor from the side surface to itself $=\mathrm{F} 33=0.3944 \ldots$. Ans.

In addition: (b) If surface 1 is at 1000 K and surface 2 is at 500 K , find the net exchange of heat between 1 and 2 , if all the surfaces are black.

Add the following code at the end of the above program and run it again:
"If surface 1 is at 1000 K and surface 2 is at 500 K , find the net exchange of heat between 1 and 2 , if all the surfaces are black."
$\mathrm{T} 1=1000[\mathrm{~K}]$
$\mathrm{T} 2=500[\mathrm{~K}]$
sigma $=5.67 \mathrm{E}-08\left[\mathrm{~W} / \mathrm{m}^{\wedge} 2-\mathrm{K}^{\wedge} 4\right]$ ".. Stefan Boltzmann const."
$\mathrm{Q} \_12=\mathrm{A} 1^{*} \mathrm{~F} 12^{*} \operatorname{sigma}^{*}(\mathrm{~T} 1 \wedge 4-\mathrm{T} 2 \wedge 4) "[\mathrm{~W}] \ldots$. net radiant exchange between surfaces 1 and 2 "

Software Solutions to Problems on Heat Transfer: Radiation Heat Transfer - Part I

View factors (or, Shape factor, configuration factor) and Radiation energy exchange between Black bodies:

## We get:

## Unit Settings: SI C Pa J mass rad

| $\mathrm{A} 1=0.007854\left[\mathrm{~m}^{2}\right]$ | $\mathrm{A} 2=0.03142\left[\mathrm{~m}^{2}\right]$ | $\mathrm{A} 3=0.05269\left[\mathrm{~m}^{2}\right]$ |
| :--- | :--- | :--- |
| $\mathrm{F} 11=0$ | $\mathrm{~F} 12=0.4689$ | $\mathrm{~F} 13=0.5311$ |
| $\mathrm{~F} 21=0.1172$ | $\mathrm{~F} 22=0$ | $\mathrm{~F} 23=0.8828$ |
| $\mathrm{~F} 31=0.07918$ | $\mathrm{~F} 32=0.5264$ | $\mathrm{~F} 33=0.3944$ |
| $\mathrm{~L}=0.1[\mathrm{~m}]$ | $Q_{12}=195.7[\mathrm{~W}]$ | $\mathrm{r}_{1}=0.05[\mathrm{~m}]$ |
| $\mathrm{r}_{2}=0.1[\mathrm{~m}]$ | $\sigma=5.670 \mathrm{E}-08\left[\mathrm{~W} / \mathrm{m}^{2} \mathrm{~K}\right]$ | $\mathrm{T} 1=1000[\mathrm{~K}]$ |
| $\mathrm{T} 2=500[\mathrm{~K}]$ |  |  |

i.e. $Q \_12=195.7 \mathrm{~W} \ldots$ net heat exchange between surfaces 1 and $2 \ldots$... Ans.


## Also, plot $F_{-} 12$ and $Q_{1} 12$ against $L$, with $L$ varying from 0.1 m to 1 m :

First, compute the Parametric Table. (Here, we have included F_33 also)


## Now, plot the results:





Prob.5B.9. Consider two coaxial, parallel rectangles, each of size: $1 \mathrm{~m} \times 1 \mathrm{~m}$, with a distance of 2 m from each other. If surface 1 is at 800 K and the surface 2 at 400 K , find the net heat interchange between the two surfaces. Consider the surfaces as black.


Fig.Prob.5B. 9

## EES Solution:

"Data:"
$\mathrm{X}=1[\mathrm{~m}]$
$\mathrm{Y}=1[\mathrm{~m}]$
$\mathrm{L}=2[\mathrm{~m}]$
$\mathrm{T} 1=800[\mathrm{~K}]$
$\mathrm{T} 2=400[\mathrm{~K}]$
sigma $=5.67 \mathrm{E}-08\left[\mathrm{~W} / \mathrm{m}^{\wedge} 2-\mathrm{K}^{\wedge} 4\right]$ "...Stefan Boltzmann constant"
"We have to find out F 12:"


## "Calculations:"

$A 1=X^{*} Y^{"}\left[m^{\wedge} 2\right] \ldots$ area of top surface"
$\mathrm{A} 2=\mathrm{A} 1 "\left[\mathrm{~m}^{\wedge} 2\right] \ldots$. area of bottom surface"

F_12 = F12_parallel_rectangles(X, Y, L)
"Net exchange of heat between A1 and A2:"

Q_12 = A1 ${ }^{*} \mathrm{~F} \_12^{*} \operatorname{sigma}^{*}(\mathrm{~T} 1 \wedge 4-\mathrm{T} 2 \wedge 4)$ " $[\mathrm{W}] \ldots$ net exchange of heat between surfaces1 and 2"

## Results:

Unit Settings: SI C Pa J mass rad
A1 $=1\left[\mathrm{~m}^{2}\right]$
$\mathrm{L}=2$ [ m$]$
$\begin{aligned} \mathrm{A} 2 & =1\left[\mathrm{~m}^{2}\right] \\ \mathrm{Q}_{12} & =1493[\mathrm{M}]\end{aligned}$
$F_{12}=0.06859$
$\mathrm{T} 1=800 \quad \mathrm{~K}]$
$\mathrm{T} 2=400[\mathrm{~K}]$
$\sigma=5.670 \mathrm{E}-08\left[\mathrm{~W} / \mathrm{m}^{2} \mathrm{~K}^{4}\right]$
$Y=1[\mathrm{~m}]$
$X=1 \quad[\mathrm{~m}]$

Thus:
Net exchange of heat between A1 and A2 = Q_12 = $1493 \mathrm{~W} \ldots$ Ans.

## In addition:

Plot the variation of $F \_12$ and $Q \_12$ as the distance between the plates $(L)$ varies from 0.5 m to 2.5 m :

First, compute the Parametric Table:

| D | L $[\mathrm{m}]$ | $F_{12}$ | $Q_{12}$ <br> [W] |
| :---: | :---: | :---: | :---: |
| Run 1 | 0.5 | 0.4153 | 9041 |
| Run 2 | 0.6 | 0.3545 | 7718 |
| Run 3 | 0.7 | 0.3044 | 6628 |
| Run 4 | 0.8 | 0.263 | 5726 |
| Run 5 | 0.9 | 0.2286 | 4977 |
| Run 6 | 1 | 0.1998 | 4351 |
| Run 7 | 1.1 | 0.1757 | 3826 |
| Run 8 | 1.2 | 0.1553 | 3382 |
| Run 9 | 1.3 | 0.1381 | 3006 |
| Run 10 | 1.4 | 0.1233 | 2686 |
| Run 11 | 1.5 | 0.1107 | 2410 |
| Run 12 | 1.6 | 0.09981 | 2173 |
| Run 13 | 1.7 | 0.09036 | 1967 |
| Run 14 | 1.8 | 0.08213 | 1788 |
| Run 15 | 1.9 | 0.07492 | 1631 |
| Run 16 | 2 | 0.06859 | 1493 |
| Run 17 | 2.1 | 0.063 | 1372 |
| Run 18 | 2.2 | 0.05804 | 1264 |
| Run 19 | 2.3 | 0.05362 | 1167 |
| Run 20 | 2.4 | 0.04968 | 1082 |
| Run 21 | 2.5 | 0.04614 | 1005 |

## Now, plot the results:



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## E.OM




View Factors for two concentric cylinders of finite length:

Let us write EES Functions for the view factors for:
a) From the outer cylinder to itself,
b) From the outer cylinder to inner cylinder, and
c) From the inner cylinder to itself

We use the view factor equations provided by Modest. (Ref: 'Radiative Heat Transfer' by Modest)

$$
\begin{aligned}
& \begin{array}{l}
\text { Interior of finite-length, right-circular coaxial cylin- } \\
\text { der to itself }
\end{array} \\
& \quad F_{2-2}=1-\frac{1}{R}-\frac{\sqrt{H^{2}+4 R^{2}}-H}{4 R}+\frac{1}{\pi}\left(\frac{2}{R} \tan ^{-1} \frac{2 \sqrt{R^{2}-1}}{H}\right. \\
& \\
& \left.-\frac{H}{2 R}\left\{\frac{\sqrt{4 R^{2}+H^{2}}}{H} \sin ^{-1} \frac{H^{2}+4\left(R^{2}-1\right)-2 H^{2} / R^{2}}{H^{2}+4\left(R^{2}-1\right)}-\sin ^{-1} \frac{R^{2}-2}{R^{2}}\right\}\right)
\end{aligned}
$$



Note: In the EES Functions we write below, height of the cylinders (i.e. h in the above figs.) is taken as L.
$\qquad$

FUNCTION F22_CoaxialCyl_outer_to_itself (R1, R2, L)
"Finds the View factor F_22 between two coaxial parallel cylinders - outer surface to itself" "Ref: Radiative Heat Transfer by Modest"

R := R2/R1
H:= L / R1
$\mathrm{A}=\left(\operatorname{sqrt}\left(\mathrm{H}^{\wedge} 2+4^{\star} \mathrm{R}^{\wedge} 2\right)-\mathrm{H}\right) /\left(4^{*} \mathrm{R}\right)$
$\mathrm{B}=\left(2^{*} \operatorname{sqrt}(\mathrm{R} \wedge 2-1)\right) / \mathrm{H}$
$C=\operatorname{sqrt}\left(4^{*} \mathrm{R}^{\wedge} 2+\mathrm{H}^{\wedge} 2\right) / H$
$\mathrm{D}=\mathrm{H}^{\wedge} 2+4^{\star}(\mathrm{R} \wedge 2-1)-2^{\star} \mathrm{H}^{\wedge} 2 / \mathrm{R} \wedge 2$
$\mathrm{E}=\mathrm{H}^{\wedge} 2+4^{*}(\mathrm{R} \wedge 2-1)$
$\mathrm{F}=(\mathrm{R} \wedge 2-2) / \mathrm{R} \wedge 2$
$\mathrm{G}=(2 / \mathrm{R}) * \arctan (\mathrm{~B})$

F22_CoaxialCyl_outer_to_itself $=1-1 / \mathrm{R}-\mathrm{A}+(1 / \mathrm{pi})^{*}\left(\mathrm{G}-\left(\mathrm{H} /\left(2^{*} \mathrm{R}\right)\right)^{*}(\mathrm{C} * \arcsin (\mathrm{D} / \mathrm{E})-\right.$ $\arcsin (\mathrm{F}))$ )

END
$\qquad$ ""

FUNCTION F21_coaxialCyl_outer_to_inner(R1, R2, L)
"Finds the View factor F_21 between two coaxial parallel cylinders - outer surface to inner surface Ref: Radiative Heat Transfer by Modest"
$\mathrm{R}:=\mathrm{R} 2 / \mathrm{R} 1$
H:= L / R1
$\mathrm{A}:=\left(\mathrm{H}^{\wedge} 2+\mathrm{R} \wedge 2-1\right) /\left(4^{*} \mathrm{H}\right)$


$\mathrm{B}:=\left(\mathrm{H}^{\wedge} 2-\mathrm{R} \wedge 2+1\right) /\left(\mathrm{H}^{\wedge} 2+\mathrm{R} \wedge 2-1\right)$
$\mathrm{C}:=\operatorname{sqrt}\left((\mathrm{H} \wedge 2+\mathrm{R} \wedge 2+1) \wedge 2-4^{\star} \mathrm{R} \wedge 2\right) /(2 * \mathrm{H})$
$\mathrm{D}:=(\mathrm{H} \wedge 2-\mathrm{R} \wedge 2+1) /\left(\mathrm{R}^{*}(\mathrm{H} \wedge 2+\mathrm{R} \wedge 2-1)\right)$
$\mathrm{E}:=\left(\mathrm{H}^{\wedge} 2-\mathrm{R} \wedge 2+1\right) /\left(2^{*} \mathrm{H}\right)$

F21_coaxialCyl_outer_to_inner $=(1 / \mathrm{R})^{*}\left(1-\mathrm{A}-(1 / \mathrm{pi})^{*}\left(\arccos (\mathrm{~B})-\mathrm{C}^{*} \arccos (\mathrm{D})-\mathrm{E}^{*} \arcsin (1\right.\right.$ ( R) ))

END
$\qquad$ _-"

FUNCTION F12_Coaxial_cyl_inner_to_outer (R1, R2, L)
\{Finds the View factor F12 between two coaxial parallel cylinders - inner surface to outer surface.
Ref: F_21 From Modest, see above.

Use: A2 ${ }^{*}$ F_21 = A1 ${ }^{*}$ F_12;
i.e. $\mathrm{F} \_12=\mathrm{F} \_21^{*}(\mathrm{~A} 2 / \mathrm{A} 1)$;

And, $\mathrm{A} 2 / \mathrm{A} 1=\mathrm{R} 2 / \mathrm{R} 1\}$

F12_Coaxial_cyl_inner_to_outer $=(\mathrm{R} 2 / \mathrm{R} 1) *$ F21_coaxialCyl_outer_to_inner(R1, R2, L)

END
$\qquad$ _-"

## Now, let us use these Functions in solving a problem:

Prob.5B.10. Two concentric cylinders have diameters of 10 and 20 cm . Calculate the View factor between the two open ends.


Fig.Prob.5B. 10

## EES Solution:

Let: Surface 1: outer surface of inner cylinder

Surface 2: inner surface of outer cylinder

Surfaces 3 and 4: surfaces at the two ends

Then, we have:
"Data:"
$\mathrm{R} 1=0.05[\mathrm{~m}]$
$\mathrm{R} 2=0.1[\mathrm{~m}]$
$\mathrm{L}=0.2[\mathrm{~m}]$
"Calculations:"
$\mathrm{A} 1=2^{*} \mathrm{pi}^{*} \mathrm{R} 1^{*} \mathrm{~L}^{"}\left[\mathrm{~m}^{\wedge} 2\right] \ldots$ area of inner cylinder surface"
$\mathrm{A} 2=2^{*} \mathrm{pi}^{*} \mathrm{R} 2{ }^{*} \mathrm{~L}$ " $\left[\mathrm{m}^{\wedge} 2\right] \ldots$ area of outer cylinder surface"
$A 3=\mathrm{pi}^{*}\left(\mathrm{R} 2^{\wedge} 2-\mathrm{R} 1^{\wedge} 2\right)$ " $\left[\mathrm{m}^{\wedge} 2\right] \ldots$ area of open, end surface"
$\mathrm{A} 4=\mathrm{A} 3$ " $\left[\mathrm{m}^{\wedge} 2\right] \ldots$ area of open, end surface"

F_21 = F21_CoaxialCyl_outer_to_inner(R1, R2, L)"...finds View factor from outer to inner surface"

F_22 = F22_CoaxialCyl_outer_to_itself (R1, R2, L)"...finds View factor from outer surface to itself"
"Now, by reciprocity:"

A1 * $\mathrm{F} \_12=\mathrm{A} 2{ }^{*} \mathrm{~F} \_21$ "..finds F_12"
"For surface 2, Summation rule gives:"

F_21 + F_22 + F_23 + F_24 = 1
"But, by symmetry:"
$\mathrm{F} \_23=\mathrm{F} \_24$ "...therefore, $\mathrm{F} \_23$ and $\mathrm{F} \_24$ are calculated."
"Again, applying reciprocity between surfaces 2 and 3:"
$\mathrm{A} 2{ }^{*} \mathrm{~F} \_23=\mathrm{A} 3^{*} \mathrm{~F} \_32$ "...finds F_32"
"Now,"

F_33 $=0$ "...since surface is flat"

F_44 = 0 "...since surface is flat"
$\mathrm{F} \_11=0$ "...since surface is convex"

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"Applying Summation rule for surface 3:"
$\mathrm{F} \_31+\mathrm{F}$ - $32+\mathrm{F} \_33+\mathrm{F}$ - $34=1$ "....eq. (A)"
"In the above, $\mathrm{F}_{-} 31$ and $\mathrm{F} \_34$ are unknown."
"To find F_31:"

F_11 + F_12 + F_13 + F_14 = 1 "...by Summation rule on surface 1"
"But,"

F_13 = F_14 ". . by Symmetry"
"Therefore, $\mathrm{F} \_13$ and $\mathrm{F} \_14$ are determined, since $\mathrm{F} \_11=0$ and $\mathrm{F} \_12$ is already determined"
"Then, by reciprocity:"
$A 1{ }^{*} F \_13=A 3{ }^{*} F \_31$ "...finds $F \_31 "$
"And, from eqn.(A), F_34 is immediately calculated."

## Results:

Unit Settings: SI C Pa J mass rad

| $A 1=0.06283\left[\mathrm{~m}^{2}\right]$ | $A 2=0.1257\left[\mathrm{~m}^{2}\right]$ | $A 3=0.02356\left[\mathrm{~m}^{2}\right]$ | $A 4=0.02356\left[\mathrm{~m}^{2}\right]$ |
| :--- | :--- | :--- | :--- |
| $F_{11}=0$ | $F_{12}=0.8253$ | $F_{13}=0.08737$ | $F_{14}=0.08737$ |
| $F_{21}=0.4126$ | $F_{22}=0.3286$ | $F_{23}=0.1294$ | $F_{24}=0.1294$ |
| $F_{31}=0.233$ | $F_{32}=0.6901$ | $F_{33}=0$ | $F_{34}=0.07694$ |
| $F_{44}=0$ | $L=0.2[\mathrm{~m}]$ | $R 1=0.05[\mathrm{~m}]$ | $R 2=0.1[\mathrm{~m}]$ |

## Thus:

View factor between the two open ends $=$ F_34 $=0.07694 \ldots$ Ans.

Also, plot F_34 for different values of L, i.e. $0.1<L<0.5 \mathrm{~m}$ :

View factors (or, Shape factor, configuration factor) and Radiation energy exchange between Black bodies:

## Parametric Table:

| ${ }_{1 . .9}$ | L [m] | $\mathrm{F}_{34}$ |
| :---: | :---: | :---: |
| Run 1 | 0.1 | 0.2036 |
| Run 2 | 0.15 | 0.1194 |
| Run 3 | 0.2 | 0.07694 |
| Run 4 | 0.25 | 0.05309 |
| Run 5 | 0.3 | 0.03858 |
| Run 6 | 0.35 | 0.02918 |
| Run 7 | 0.4 | 0.02279 |
| Run 8 | 0.45 | 0.01826 |
| Run 9 | 0.5 | 0.01494 |

Now, plot the results:


Now, let us write VBA Functions for view factors for a few important geometries:
We shall write VBA Functions for:

1. Aligned, parallel rectangles,
2. Coaxial, parallel disks, and
3. Perpendicular rectangles with a common edge

Recollect the nomenclature of geometries:

(a) Aligned parallel rectangles
(b) Coaxial parallel disks

(c) Perpendicular rectangles with a common edge

Flg.13.18 Few three-dimensional geometries

And, the equations for View factors are given at the beginning of the chapter.

Now, we shall present the VBA programs for View factors:

Software Solutions to Problems on Heat Transfer: Radiation Heat Transfer - Part I

View factors (or, Shape factor, configuration factor) and Radiation energy exchange between Black bodies:

## 1. Aligned, parallel rectangles:

```
Option Explicit
Function F_ij_Aligned_parallel_rectangles(XBYL As Double, YBYL As Double) As Double
'Finds the View factor F_ij between two aligned parallel rectangles
'Note that VB does not have the ATAN function
'So, use the relation: ATAN(x) = ASIN(x/sqrt(x^2 +1))
Dim XX As Double, YY As Double, A As Double, B As Double, C As Double
Dim D As Double, E As Double, F As Double, BB As Double, CC As Double
XX = XBYL
YY = YBYL
BB = (XX / (1 + YY ^ 2) ^ (1/2))
CC = (YY / (1 + XX ^ 2) ^ (1 / 2))
A = 2 / (Application.Pi() * XX * YY)
B = Application.Ln(Sqr ((1 + XX ^ 2) * (1 + YY^ 2)/ (1 + XX^ 2 + YY ^ 2)))
'C = XX * (1 + YY ^ 2) ^ (1/ 2) * Atan(XX / (1 + YY^2) ^ (1/2))
'D = YY * (1 + XX ^ 2) ^ (1/ 2) * Atan(YY / (1 + XX ^ 2) ^ (1/ / 2))
C = XX * (1 + YY ^ 2) ^ (1 / 2) * Application.Asin(BB / Sqr (BB ^ 2 + 1))
D = YY * (1 + XX ^ 2) ^ (1 / 2) * Application.Asin(CC / Sqr(CC ^ 2 + 1))|
'E = XX * Atan(XX)
'F = YY * Atan(YY)
E = XX * Application.Asin(XX / Sqr(XX ^ 2 + 1))
F = YY * Application.Asin(YY / Sqr(YY ^ 2 + 1))
F_ij_Aligned_parallel_rectangles = A * (B + C + D - E - F)
Eñd Function
```




## 2. Coaxial, parallel disks:

```
Function F_ij_Coaxial_parallel_discs(L_by_r_i As Double, r_j_by_L As Double) As Double
'Finds the View factor F_ij between two coaxial parallel discs
Dim RR_i As Double, RR_j As Double, S As Double
RR_i = 1 / L_by_r_i
RR_j = r_j_by_L
S = 1 + (1 + RR_j ^ 2) / RR_i ^ 2
F_ij_Coaxial_parallel_discs =}=(1/2)*(S - (S ^ 2 - 4 * (RR_j / RR_i) ^ 2) ^ 0.5
End Function
```


## 3. Perpendicular rectangles with a common edge:

```
Function F_ij_Perpendicular_rectangles(ZBYX As Double, YBYX As Double) As Double
'Finds the View factor F_ij between two perpendicular rectangles with a common edge
'common edge: X; other sides:Y, Z
Dim H As Double, W As Double, A As Double, B As Double, C As Double
Dim D As Double, E As Double, E_1 As Double, E_2 As Double, E_3 As Double
H = ZBYX
W = YBYX
A = 1 / (Application.Pi() * W)
B = W * Application.Asin((1 / W) / Sqr((1 / W) ^ 2 + 1))
C = H * Application.Asin((1/H) / Sqr((1/H) ^ 2 + 1))
D = (H^2 + W ^ 2) ^ 0.5 * Application.Asin((1/ Sqr(H^2 + W ^ 2)) / Sqr((1 / _
(H^2 + W^2)) + 1))
E_1 = (1 + W^2)* (1 + H^2)/(1 + W^2 + H^2)
E_2 = (W^2* (1 + W^2 + H^2) / ((1 + W^2)* (W^2 + H ^ 2))) ^ (W^2)
```



```
E'= E_1 * E_2 * E_3
F_ij_Perpendicular_rectangles = A * (B + C - D + 0.25 * Application.Ln(E))
Eñd Function
```

Prob.5B.11. Draw the View factor graphs for Aligned, parallel rectangles:

## EXCEL Solution:

We shall use the VBA Function written above.

Following are the steps:

1. Set up the EXCEL worksheet. Set up a Table as follows:

| D82 |  |  | - $f_{x}$ | ¢ $=$ F_ij_Aligned_parallel_rectangles(C82,\$D\$81) |  |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 4 | A | B | C | D | E | F | G | H | I | 」 | K |
| 80 |  |  |  |  |  |  |  |  |  |  |  |
| 81 |  |  | XbyL\YbyL = | 0.1 | 0.2 | 0.4 | 0.6 | 1 | 2 | 4 | 10 |
| 82 |  |  | 0.1 | 0.003162057 | 0.00626 | 0.01207 | 0.01715 | 0.02492 | 0.03514 | 0.04209 | 0.04671 |
| 83 |  |  | 0.2 |  |  |  |  |  |  |  |  |
| 84 |  |  | 0.3 |  |  |  |  |  |  |  |  |
| 85 |  |  | 0.4 |  |  |  |  |  |  |  |  |
| 86 |  |  | 0.5 |  |  |  |  |  |  |  |  |
| 87 |  |  | 0.6 |  |  |  |  |  |  |  |  |
| 88 |  |  | 0.7 |  |  |  |  |  |  |  |  |
| 89 |  |  | 0.8 |  |  |  |  |  |  |  |  |
| 90 |  |  | 0.9 |  |  |  |  |  |  |  |  |
| 91 |  |  | 1 |  |  |  |  |  |  |  |  |
| 92 |  |  | 2 |  |  |  |  |  |  |  |  |
| 93 |  |  | 3 |  |  |  |  |  |  |  |  |
| 94 |  |  | 4 |  |  |  |  |  |  |  |  |
| 95 |  |  | 5 |  |  |  |  |  |  |  |  |
| 96 |  |  | 6 |  |  |  |  |  |  |  |  |
| 97 |  |  | 7 |  |  |  |  |  |  |  |  |
| 98 |  |  | 8 |  |  |  |  |  |  |  |  |
| 99 |  |  | 9 |  |  |  |  |  |  |  |  |
| 100 |  |  | 10 |  |  |  |  |  |  |  |  |
| 101 |  |  | 20 |  |  |  |  |  |  |  |  |
| 102 |  |  | 30 |  |  |  |  |  |  |  |  |
| 103 |  |  | 40 |  |  |  |  |  |  |  |  |
| 104 |  |  | 50 |  |  |  |  |  |  |  |  |
| 105 |  |  | 60 |  |  |  |  |  |  |  |  |
| 106 |  |  | 70 |  |  |  |  |  |  |  |  |
| 107 |  |  | 80 |  |  |  |  |  |  |  |  |
| 108 |  |  | 90 |  |  |  |  |  |  |  |  |
| 109 |  |  | 100 |  |  |  |  |  |  |  |  |

In the above screen shot, column C contains $\mathrm{X} / \mathrm{L}$ values and row 81 contains $\mathrm{Y} / \mathrm{L}$ values.

We use the VBA Function written above to calculate the view factor F_ij (i.e. view factor from surface i to surface j), for given values of X/L and Y/L. Formula entered in cellD82 can be seen in the Formula bar. Use relative references to $\mathrm{X} / \mathrm{L}$ so that we can drag-copy to fill up the Table later. Similarly, write formulas for other $\mathrm{Y} / \mathrm{L}$ values as shown above.

Software Solutions to Problems on Heat
Transfer: Radiation Heat Transfer - Part I

View factors (or, Shape factor, configuration factor) and Radiation energy exchange between Black bodies:

Recollect that to enter the desired VBA Function, click on 'Insert Function' mark:


Clicking on 'Insert Function', gives:


In the above window, click on 'User Defined' category and press OK:


Choose ' $\mathrm{F}_{-}$ij Aligned, parallel rectangles' and press OK. We get the following window; fill it up as shown. Note that cell C82 for $\mathrm{X} / \mathrm{L}$ is in relative reference and cell D81 for Y/L is in absolute reference, so that when we drag-copy downwards, appropriate $\mathrm{X} / \mathrm{L}$ values are automatically chosen:


Click OK. And the value of $\mathrm{F}_{-} \mathrm{ij}$ is calculated.

Software Solutions to Problems on Heat Transfer: Radiation Heat Transfer - Part I

View factors (or, Shape factor, configuration factor) and Radiation energy exchange between Black bodies:

Complete the row 82 likewise.
2. Now, select cells D82 to K82 and drag-copy to the end of the Table, i.e. up to the cell K109. The calculations are done immediately and the Table is filled up:

| XbyL\YbyL = | 0.1 | 0.2 | 0.4 | 0.6 | 1 | 2 | 4 | 10 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 0.1 | 0.0031621 | 0.00626 | 0.01207 | 0.01715 | 0.02492 | 0.03514 | 0.04209 | 0.04671 |
| 0.2 | 0.006263 | 0.0124 | 0.02392 | 0.03398 | 0.04941 | 0.06971 | 0.08353 | 0.09272 |
| 0.3 | 0.009247 | 0.01832 | 0.03532 | 0.0502 | 0.07305 | 0.10318 | 0.12372 | 0.13739 |
| 0.4 | 0.012073 | 0.02392 | 0.04614 | 0.0656 | 0.09554 | 0.13513 | 0.16219 | 0.18021 |
| 0.5 | 0.014711 | 0.02915 | 0.05625 | 0.08001 | 0.11665 | 0.16527 | 0.19859 | 0.22081 |
| 0.6 | 0.017148 | 0.03398 | 0.0656 | 0.09336 | 0.13627 | 0.19342 | 0.23271 | 0.25896 |
| 0.7 | 0.019379 | 0.0384 | 0.07417 | 0.10563 | 0.15435 | 0.21951 | 0.26447 | 0.29455 |
| 0.8 | 0.021411 | 0.04243 | 0.08199 | 0.11683 | 0.17092 | 0.24356 | 0.29387 | 0.32759 |
| 0.9 | 0.023255 | 0.04609 | 0.0891 | 0.12702 | 0.18605 | 0.26564 | 0.321 | 0.35816 |
| 1 | 0.024925 | 0.04941 | 0.09554 | 0.13627 | 0.19982 | 0.28588 | 0.34596 | 0.38638 |
| 2 | 0.035143 | 0.06971 | 0.13513 | 0.19342 | 0.28588 | 0.41525 | 0.50899 | 0.57338 |
| 3 | 0.039652 | 0.07868 | 0.15268 | 0.21888 | 0.32474 | 0.47558 | 0.58762 | 0.66599 |
| 4 | 0.042091 | 0.08353 | 0.16219 | 0.23271 | 0.34596 | 0.50899 | 0.63204 | 0.71933 |
| 5 | 0.043603 | 0.08654 | 0.16809 | 0.2413 | 0.35917 | 0.52993 | 0.6602 | 0.75363 |
| 6 | 0.044628 | 0.08858 | 0.17209 | 0.24712 | 0.36813 | 0.54421 | 0.67954 | 0.77741 |
| 7 | 0.045366 | 0.09005 | 0.17497 | 0.25132 | 0.37461 | 0.55454 | 0.69359 | 0.79482 |
| 8 | 0.045924 | 0.09116 | 0.17715 | 0.25449 | 0.3795 | 0.56236 | 0.70426 | 0.80811 |
| 9 | 0.046359 | 0.09202 | 0.17885 | 0.25697 | 0.38332 | 0.56847 | 0.71261 | 0.81856 |
| 10 | 0.046708 | 0.09272 | 0.18021 | 0.25896 | 0.38638 | 0.57338 | 0.71933 | 0.82699 |
| 20 | 0.048288 | 0.09586 | 0.18638 | 0.26795 | 0.40026 | 0.59563 | 0.7499 | 0.86563 |
| 30 | 0.048817 | 0.09691 | 0.18845 | 0.27096 | 0.40491 | 0.60309 | 0.76017 | 0.8787 |
| 40 | 0.049081 | 0.09744 | 0.18948 | 0.27246 | 0.40723 | 0.60682 | 0.76532 | 0.88526 |
| 50 | 0.049240 | 0.09775 | 0.1901 | 0.27337 | 0.40863 | 0.60906 | 0.76841 | 0.8892 |
| 60 | 0.049346 | 0.09797 | 0.19051 | 0.27397 | 0.40956 | 0.61056 | 0.77047 | 0.89183 |
| 70 | 0.049422 | 0.09812 | 0.19081 | 0.2744 | 0.41022 | 0.61162 | 0.77194 | 0.89371 |
| 80 | 0.049478 | 0.09823 | 0.19103 | 0.27472 | 0.41072 | 0.61243 | 0.77304 | 0.89512 |
| 90 | 0.049523 | 0.09832 | 0.1912 | 0.27497 | 0.41111 | 0.61305 | 0.7739 | 0.89622 |
| 100 | 0.049558 | 0.09839 | 0.19134 | 0.27517 | 0.41142 | 0.61355 | 0.77459 | 0.89709 |

3. Now draw the graphs in EXCEL:


Prob.5B.12. Draw the View factor graphs for coaxial, parallel disks:

## EXCEL Solution:

We shall use the VBA Function written above.

Following are the steps:

1. Set up the EXCEL worksheet. Set up a Table as follows:

|  | D141 |  | - | $f_{x}=F$ ij_Coaxial_parallel_discs(C141,\$D\$140) |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 4 | A | B | C | D | E | F | G | H | I | 」 |
| 139 |  |  |  |  |  |  |  |  |  |  |
| 140 |  |  | L_byR_i\R_jby_L | 0.3 | 0.6 | 1 | 1.5 | 2 | 4 | 8 |
| 141 |  |  | 0.1 | 0.000891081 | 0.00356 | 0.0099 | 0.02227 | 0.03959 | 0.15812 | 0.62344 |
| 142 |  |  | 0.2 |  |  |  |  |  |  |  |
| 143 |  |  | 0.3 |  |  |  |  |  |  |  |
| 144 |  |  | 0.4 |  |  |  |  |  |  |  |
| 145 |  |  | 0.5 |  |  |  |  |  |  |  |
| 146 |  |  | 0.6 |  |  |  |  |  |  |  |
| 147 |  |  | 0.7 |  |  |  |  |  |  |  |
| 148 |  |  | 0.8 |  |  |  |  |  |  |  |
| 149 |  |  | 0.9 |  |  |  |  |  |  |  |
| 150 |  |  | 1 |  |  |  |  |  |  |  |
| 151 |  |  | 2 |  |  |  |  |  |  |  |
| 152 |  |  | 3 |  |  |  |  |  |  |  |
| 153 |  |  | 4 |  |  |  |  |  |  |  |
| 154 |  |  | 5 |  |  |  |  |  |  |  |
| 155 |  |  | 6 |  |  |  |  |  |  |  |
| 156 |  |  | 7 |  |  |  |  |  |  |  |
| 157 |  |  | 8 |  |  |  |  |  |  |  |
| 158 |  |  | 9 |  |  |  |  |  |  |  |
| 159 |  |  | 10 |  |  |  |  |  |  |  |
| 160 |  |  | 20 |  |  |  |  |  |  |  |
| 161 |  |  | 30 |  |  |  |  |  |  |  |
| 162 |  |  | 40 |  |  |  |  |  |  |  |
| 163 |  |  | 50 |  |  |  |  |  |  |  |
| 164 |  |  | 60 |  |  |  |  |  |  |  |
| 165 |  |  | 70 |  |  |  |  |  |  |  |
| 166 |  |  | 80 |  |  |  |  |  |  |  |
| 167 |  |  | 90 |  |  |  |  |  |  |  |
| 168 |  |  | 100 |  |  |  |  |  |  |  |

In the above screen shot, column $C$ contains $L / R \_i$ values and row 140 contains $R \_j / L$ values.

We use the VBA Function written above to calculate the view factor $\mathrm{F}_{-} \mathrm{ij}$ (i.e. view factor from surface ito surface j), for given values of L/R_i and R_j/L. Formula entered in cell D141 can be seen in the Formula bar. Use relative references to $L / R \_i$ so that we can drag-copy to fill up the Table later. Similarly, write formulas for other $R_{-} j / L$ values as shown above.

Software Solutions to Problems on Heat Transfer: Radiation Heat Transfer - Part I

View factors (or, Shape factor, configuration factor) and Radiation energy exchange between Black bodies:
2. Now, select cells D141 to J141 and drag-copy to the end of the Table, i.e. up to the cell J168.

The calculations are done immediately and the Table is filled up:

| L_byR_i\R_jby_L | 0.3 | 0.6 | 1 | 1.5 | 2 | 4 | 8 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 0.1 | 0.000891081 | 0.00356 | 0.0099 | 0.02227 | 0.03959 | 0.15812 | 0.62344 |
| 0.2 | 0.003461076 | 0.01384 | 0.0384 | 0.08623 | 0.15279 | 0.58388 | 0.97538 |
| 0.3 | 0.007426605 | 0.02965 | 0.08196 | 0.18242 | 0.31803 | 0.86472 | 0.98151 |
| 0.4 | 0.012392345 | 0.0493 | 0.13502 | 0.29352 | 0.48769 | 0.91153 | 0.98301 |
| 0.5 | 0.017934496 | 0.07092 | 0.19098 | 0.39754 | 0.60961 | 0.92482 | 0.98362 |
| 0.6 | 0.023671607 | 0.09278 | 0.24388 | 0.47901 | 0.67889 | 0.93063 | 0.98394 |
| 0.7 | 0.029306344 | 0.1136 | 0.28993 | 0.53614 | 0.71722 | 0.93375 | 0.98412 |
| 0.8 | 0.03463697 | 0.13258 | 0.32784 | 0.57481 | 0.73985 | 0.93564 | 0.98424 |
| .9 | 0.039547509 | 0.14937 | 0.35811 | 0.60123 | 0.75423 | 0.93688 | 0.98432 |
| 1 | 0.043988012 | 0.16393 | 0.38197 | 0.6198 | 0.76393 | 0.93774 | 0.98438 |
| 2 | 0.068027565 | 0.23196 | 0.46887 | 0.67544 | 0.79176 | 0.94035 | 0.98456 |
| 3 | 0.075457337 | 0.24941 | 0.48612 | 0.68493 | 0.7964 | 0.94081 | 0.98459 |
| 4 | 0.078424643 | 0.25595 | 0.49219 | 0.68818 | 0.79798 | 0.94097 | 0.9846 |
| 5 | 0.079871841 | 0.25906 | 0.495 | 0.68967 | 0.79871 | 0.94105 | 0.98461 |
| 6 | 0.080678654 | 0.26077 | 0.49653 | 0.69048 | 0.79911 | 0.94109 | 0.98461 |
| 7 | 0.081172376 | 0.26181 | 0.49745 | 0.69097 | 0.79935 | 0.94111 | 0.98461 |
| 8 | 0.081495782 | 0.26248 | 0.49805 | 0.69128 | 0.7995 | 0.94113 | 0.98461 |
| 9 | 0.081718871 | 0.26295 | 0.49846 | 0.6915 | 0.7996 | 0.94114 | 0.98461 |
| 10 | 0.081879129 | 0.26328 | 0.49875 | 0.69165 | 0.79968 | 0.94114 | 0.98461 |
| 20 | 0.082395398 | 0.26435 | 0.49969 | 0.69214 | 0.79992 | 0.94117 | 0.98461 |
| 0 | 0.082491655 | 0.26455 | 0.49986 | 0.69223 | 0.79996 | 0.94117 | 0.98462 |
| 40 | 0.082525393 | 0.26462 | 0.49992 | 0.69227 | 0.79998 | 0.94117 | 0.98462 |
| 50 | 0.082541017 | 0.26465 | 0.49995 | 0.69228 | 0.79999 | 0.94118 | 0.98462 |
| 60 | 0.082549507 | 0.26467 | 0.49997 | 0.69229 | 0.79999 | 0.94118 | 0.98462 |
| 70 | 0.082554627 | 0.26468 | 0.49997 | 0.69229 | 0.79999 | 0.94118 | 0.98462 |
| 0 | 0.08255795 | 0.26468 | 0.49998 | 0.6923 | 0.79999 | 0.94118 | 0.98462 |
| 0 | 0.082560228 | 0.26469 | 0.49998 | 0.6923 | 0.8 | 0.94118 | 0.98462 |
| 100 | 0.082561858 | 0.26469 | 0.49999 | 0.6923 | 0.8 | 0.94118 | 0.98462 |



Software Solutions to Problems on Heat Transfer: Radiation Heat Transfer - Part I

View factors (or, Shape factor, configuration factor) and Radiation energy exchange between Black bodies:
3. Now, draw the graphs in EXCEL:


Prob.5B.13. Draw the View factor graphs for Perpendicular rectangles with a common edge:

## EXCEL Solution:

We shall use the VBA Function written above.

Following are the steps:

1. Set up the EXCEL worksheet. Set up a Table as follows:

| D207 |  |  |  | ${ }_{x}$ =F_ij_Perpendicular_rectangles(C207,\$D\$206) |  |  |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 4 | A | B | C | D | E | F | G | H | I | 」 | K | L |
| 205 |  |  |  |  |  |  |  |  |  |  |  |  |
| 206 |  |  | Zby $\mathrm{X} \backslash \mathrm{Yby}$ X $=$ | 0.02 | 0.05 | 0.1 | 0.2 | 0.6 | 1 | 2 | 4 | 10 |
| 207 |  |  | 0.1 | 0.443749568 | 0.37202 | 0.28189 | 0.18108 | 0.07048 | 0.04325 | 0.02188 | 0.01097 | 0.00439 |
| 208 |  |  | 0.2 |  |  |  |  |  |  |  |  |  |
| 209 |  |  | 0.3 |  |  |  |  |  |  |  |  |  |
| 210 |  |  | 0.4 |  |  |  |  |  |  |  |  |  |
| 211 |  |  | 0.5 |  |  |  |  |  |  |  |  |  |
| 212 |  |  | 0.6 |  |  |  |  |  |  |  |  |  |
| 213 |  |  | 0.7 |  |  |  |  |  |  |  |  |  |
| 214 |  |  | 0.8 |  |  |  |  |  |  |  |  |  |
| 215 |  |  | 0.9 |  |  |  |  |  |  |  |  |  |
| 216 |  |  | 1 |  |  |  |  |  |  |  |  |  |
| 217 |  |  | 1.5 |  |  |  |  |  |  |  |  |  |
| 218 |  |  | 2 |  |  |  |  |  |  |  |  |  |
| 219 |  |  | 2.5 |  |  |  |  |  |  |  |  |  |
| 220 |  |  | 3 |  |  |  |  |  |  |  |  |  |
| 221 |  |  | 3.5 |  |  |  |  |  |  |  |  |  |
| 222 |  |  | 4 |  |  |  |  |  |  |  |  |  |
| 223 |  |  | 4.5 |  |  |  |  |  |  |  |  |  |
| 224 |  |  | 5 |  |  |  |  |  |  |  |  |  |
| 225 |  |  | 5.5 |  |  |  |  |  |  |  |  |  |
| 226 |  |  | 6 |  |  |  |  |  |  |  |  |  |
| 227 |  |  | 6.5 |  |  |  |  |  |  |  |  |  |
| 228 |  |  | 7 |  |  |  |  |  |  |  |  |  |
| 229 |  |  | 7.5 |  |  |  |  |  |  |  |  |  |
| 230 |  |  | 8 |  |  |  |  |  |  |  |  |  |
| 231 |  |  | 8.5 |  |  |  |  |  |  |  |  |  |
| 232 |  |  | 9 |  |  |  |  |  |  |  |  |  |
| 233 |  |  | 9.5 |  |  |  |  |  |  |  |  |  |
| 234 |  |  | 10 |  |  |  |  |  |  |  |  |  |

In the above screen shot, column C contains $\mathrm{Z} / \mathrm{X}$ values and row 206 contains $\mathrm{Y} / \mathrm{X}$ values.

We use the VBA Function written above to calculate the view factor F_ij (i.e. view factor from surface i to surface j), for given values of $\mathrm{Z} / \mathrm{X}$ and $\mathrm{Y} / \mathrm{X}$. Formula entered in cell D207 can be seen in the Formula bar. Use relative references to $\mathrm{Z} / \mathrm{X}$ so that we can drag-copy to fill up the Table later. Similarly, write formulas for other $\mathrm{Y} / \mathrm{X}$ values as shown above.

Software Solutions to Problems on Heat Transfer: Radiation Heat Transfer - Part I

View factors (or, Shape factor, configuration factor) and Radiation energy exchange between Black bodies:
2. Now, select cells D207 to L207 and drag-copy to the end of the Table, i.e. up to the cell L234.

The calculations are done immediately and the Table is filled up:

| ZbyX YbyX $=$ | 0.02 | 0.05 | 0.1 | 0.2 | 0.6 | 1 | 2 | 4 | 10 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 0.1 | 0.443749568 | 0.37202 | 0.28189 | 0.18108 | 0.07048 | 0.04325 | 0.02188 | 0.01097 | 0.00439 |
| 0.2 | 0.466154321 | 0.42337 | 0.36216 | 0.27104 | 0.12295 | 0.07744 | 0.03971 | 0.02 | 0.00802 |
| 0.3 | 0.473183243 | 0.44044 | 0.39322 | 0.31714 | 0.16239 | 0.10512 | 0.05473 | 0.02768 | 0.01111 |
| 0.4 | 0.476459637 | 0.44852 | 0.40859 | 0.34295 | 0.19206 | 0.1277 | 0.06757 | 0.03434 | 0.0138 |
| 0.5 | 0.478288879 | 0.45305 | 0.41738 | 0.35864 | 0.21445 | 0.14619 | 0.07865 | 0.04018 | 0.01617 |
| 0.6 | 0.479423474 | 0.45587 | 0.4229 | 0.36884 | 0.23147 | 0.16138 | 0.08829 | 0.04536 | 0.01829 |
| 0.7 | 0.480178071 | 0.45775 | 0.4266 | 0.37581 | 0.24453 | 0.1739 | 0.09671 | 0.04998 | 0.02019 |
| 0.8 | 0.480705777 | 0.45906 | 0.4292 | 0.38078 | 0.25467 | 0.18425 | 0.10411 | 0.05413 | 0.02191 |
| 0.9 | 0.481089185 | 0.46002 | 0.43109 | 0.38443 | 0.26263 | 0.19286 | 0.11065 | 0.05789 | 0.02348 |
| 1 | 0.481376318 | 0.46073 | 0.43251 | 0.38719 | 0.26896 | 0.20004 | 0.11643 | 0.06131 | 0.02492 |
| 1.5 | 0.4821110417 | 0.46257 | 0.43616 | 0.39435 | 0.28678 | 0.22232 | 0.13715 | 0.07458 | 0.03069 |
| 2 | 0.482390134 | 0.46327 | 0.43756 | 0.39711 | 0.29429 | 0.23285 | 0.1493 | 0.08365 | 0.03491 |
| 2.5 | 0.482524639 | 0.4636 | 0.43823 | 0.39844 | 0.29805 | 0.23847 | 0.15682 | 0.09015 | 0.03818 |
| 3 | 0.482599192 | 0.46379 | 0.4386 | 0.39919 | 0.30019 | 0.24176 | 0.16169 | 0.09495 | 0.04082 |
| 3.5 | 0.482644682 | 0.4639 | 0.43883 | 0.39964 | 0.30151 | 0.24383 | 0.16499 | 0.09857 | 0.043 |
| 4 | 0.48267443 | 0.46398 | 0.43897 | 0.39993 | 0.30238 | 0.24522 | 0.16731 | 0.10136 | 0.04484 |
| 4.5 | 0.48269493 | 0.46403 | 0.43908 | 0.40014 | 0.30298 | 0.24619 | 0.16899 | 0.10353 | 0.04641 |
| 5 | 0.482709646 | 0.46406 | 0.43915 | 0.40029 | 0.30342 | 0.2469 | 0.17024 | 0.10526 | 0.04777 |
| 5.5 | 0.482720563 | 0.46409 | 0.43921 | 0.4004 | 0.30374 | 0.24743 | 0.17119 | 0.10664 | 0.04895 |
| 6 | 0.482728883 | 0.46411 | 0.43925 | 0.40048 | 0.30399 | 0.24783 | 0.17193 | 0.10776 | 0.04998 |
| 6.5 | 0.482735367 | 0.46413 | 0.43928 | 0.40054 | 0.30418 | 0.24815 | 0.17252 | 0.10868 | 0.05089 |
| 7 | 0.482740519 | 0.46414 | 0.4393 | 0.40059 | 0.30433 | 0.2484 | 0.17299 | 0.10944 | 0.0517 |
| 7.5 | 0.482744679 | 0.46415 | 0.43933 | 0.40064 | 0.30446 | 0.2486 | 0.17338 | 0.11008 | 0.05242 |
| 8 | 0.482748086 | 0.46416 | 0.43934 | 0.40067 | 0.30456 | 0.24877 | 0.1737 | 0.11062 | 0.05306 |
| 8.5 | 0.482750912 | 0.46417 | 0.43936 | 0.4007 | 0.30464 | 0.24891 | 0.17397 | 0.11108 | 0.05363 |
| 9 | 0.482753281 | 0.46417 | 0.43937 | 0.40072 | 0.30471 | 0.24903 | 0.17419 | 0.11147 | 0.05414 |
| 9.5 | 0.482755287 | 0.46418 | 0.43938 | 0.40074 | 0.30477 | 0.24912 | 0.17438 | 0.11181 | 0.05461 |
| 10 | 0.482757001 | 0.46418 | 0.43939 | 0.40076 | 0.30482 | 0.24921 | 0.17455 | 0.1121 | 0.05502 |

1. Now, draw the graphs in EXCEL:


Prob.5B.14. Draw the View factor graphs for two concentric cylinders of finite length:

Recollect:

View Factors for two concentric cylinders of finite length:

Let us write EES Functions for the view factors for:
a) From the outer cylinder to inner cylinder, F21,
b) From the inner cylinder to outer cylinder, F12
c) From outer cylinder to itself, F22

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We use the view factor equations provided by Modest. (Ref: 'Radiative Heat Transfer' by Modest)


## First, let us write VBA Functions for the View factors, using the above equations:

## 1. View Factor $\mathrm{F}_{\text {_ }} 21$ from outer cylinder to inner cylinder for different L/R2:

```
Function F_21_Coaxial_cyl_outer_surface_to_inner_surface(R1_by_R2 As Double, _
H_by_R2 As Double) As Doub}l
'Finds the View factor F 21 between two coaxial parallel cylinders--outer surface to
'inner surface
'Ref: Radiative Heat Transfer by Modest
Dim R As Double, H As Double, A As Double, B As Double, C As Double
Dim D As Double, E As Double
R=1/R1_by_R2
H}=\textrm{H}\mathrm{ by R }\overline{2}*\mp@subsup{*}{}{-}
A}=(\overline{H}\wedge2+R^2-1#)/(4#*H
B=(H^2-R^2 + 1#)/(H^2 + (H^2 - 1#)
C = Sqr ((H^2 + R^2 + 1) ^2 - 4* * N 2) / (2#* H)
```



```
E = (H^2-R^2 + 1#)/(2#* H)
F_21_Coaxial_cyl_outer_surface_to_inner_surface = (1# / R) * (1# - A - (1# / Application.Pi()) * _
(\overline{Application.Acos(B) -- C * Applicàtion.A}cos(D) - E * Application.Asin(1 / R)))
End Function
```


## 2. View Factor F_12 from inner cylinder to outer cylinder for different L/R2:

```
Function F_12_Coaxial_cyl_inner_surface_to_outer_surface(R1_by_R2 As Double, _
H_by_R2 As Double) As Double
'Ref: F 21 From Modest, see above
Use: A\overline{2.F_21 = A1*F_12; i.e. F_12 = F_21*(A2/A1); And, A.2/A1 = R2/R1}
    F_12_Coaxial_cyl_inner_surface_to_outer_surface =
    F_21_Coaxial_cyl_outer_surface_to_inner_surface(R1_by_R2, H_by_R2) * (1# / R1_by_R2)|
End Function
```

3. View Factor F_2 $_{2}$ from outer cylinder to itself for different $\mathbf{L} / \mathbf{R} 2$ :
```
Function F_22_Coaxial_cyl_outer_to_itself(R1_by_R2 As Double, H_by_R2 As Double) As Double
'Finds the View factor F_-22 between two coaxial parallel cylinders--outer surface to itself
'Ref: Radiative Heat Transfer by Modest
Dim R As Double, H As Double, A As Double, B As Double, C As Double
Dim D As Double, E As Double, F As Double, G As Double
R=1/R1_by_R2
H}=\mp@subsup{H}{-}{\prime}\mp@subsup{b}{_}{\prime}R2*-
A = (Sqr (H^2 + 4*R^2)-H)/(4*R)
B = (2* Sqr (R^2 - 1)) / H
C = Sqr (4* R^2 + H^2) / H
D=H^2 + 4* (R^2-1)-2*H^2/R^2
E = H^2 + 4* (R^2-1)
F=(R^2-2)/R^2
Note that VB does not have the ATAN function
'So, use the relation: ATAN(x) = ASIN(x/sqrt (x^2 +1))
'G = (2 / R) * Application.Atan(B)
G = (2/R) * Application.Asin(B/Sqr(B^2 + 1))
F_22_Coaxial_cyl_outer_to_itself = 1 - 1 / R - A + (1 / Application.Pi()) * (G - (H / (2 * R)) * _
(\overline{C}*-Application.Asin(\overline{D /- E) - Application.Asin(F)))}
End Function
```

Now, using the above Functions, let us draw the graphs for different View factors for two concentric cylinders of finite length:

Prob.5B.14. Draw the View factor graph for $\mathrm{F}_{-} 21$ for two concentric cylinders of finite length:

## EXCEL Solution:

1. Set up a Table in EXCEL worksheet as shown below:

| E382 |  | $\cdots f_{x}$ | =F_21_Coaxial_cyl_outer_surface_to_inner_surface(D382,\$E\$3i |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 4 | C | D | E | F | G | H | I | J |
| 380 |  |  |  |  |  |  |  |  |
| 381 |  | (R1_by_R2) \L/R2 | 0.1 | 0.25 | 0.5 | 1 | 2 | $1.00 \mathrm{E}+06$ |
| 382 |  | 0.001 | $6.35 \mathrm{E}-05$ | 0.000156 | 0.000295 | 0.0005 | 0.000705 | 0.001 |
| 383 |  | 0.1 |  |  |  |  |  |  |
| 384 |  | 0.15 |  |  |  |  |  |  |
| 385 |  | 0.2 |  |  |  |  |  |  |
| 386 |  | 0.25 |  |  |  |  |  |  |
| 387 |  | 0.3 |  |  |  |  |  |  |
| 388 |  | 0.35 |  |  |  |  |  |  |
| 389 |  | 0.4 |  |  |  |  |  |  |
| 390 |  | 0.45 |  |  |  |  |  |  |
| 391 |  | 0.5 |  |  |  |  |  |  |
| 392 |  | 0.55 |  |  |  |  |  |  |
| 393 |  | 0.6 |  |  |  |  |  |  |
| 394 |  | 0.65 |  |  |  |  |  |  |
| 395 |  | 0.7 |  |  |  |  |  |  |
| 396 |  | 0.75 |  |  |  |  |  |  |
| 397 |  | 0.8 |  |  |  |  |  |  |
| 398 |  | 0.85 |  |  |  |  |  |  |
| 399 |  | 0.9 |  |  |  |  |  |  |
| 400 |  | 0.95 |  |  |  |  |  |  |
| 401 |  | 1 |  |  |  |  |  |  |

In the above screen shot, column D contains R1/R2 values and row 381 contains L/R2 values.

We use the VBA Function written above to calculate the view factor F_21 (i.e. view factor from surface 2 to surface 1), for given values of R1/R2 and L/R2. Formula entered in cell E382 can be seen in the Formula bar. Use relative references to R1/R2 so that we can drag-copy to fill up the Table later. Similarly, write formulas for other L/R2 values as shown above.

Software Solutions to Problems on Heat Transfer: Radiation Heat Transfer - Part I
2. Now, select cells E382 to J382 and drag-copy to the end of the Table, i.e. up to the cell J401. The calculations are done immediately and the Table is filled up:

| (R1_by_R2) \L/R2 | 0.1 | 0.25 | 0.5 | 1 | 2 | $1.00 \mathrm{E}+06$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 0.001 | $6.35 \mathrm{E}-05$ | 0.000156 | 0.000295 | 0.0005 | 0.000705 | 0.001 |
| 0.1 | 0.006898 | 0.016903 | 0.031709 | 0.052666 | 0.072561 | 0.1 |
| 0.15 | 0.010836 | 0.026504 | 0.04945 | 0.081196 | 0.110487 | 0.15 |
| 0.2 | 0.015184 | 0.037054 | 0.0687 | 0.111382 | 0.149592 | 0.2 |
| 0.25 | 0.020024 | 0.048732 | 0.089691 | 0.143382 | 0.189938 | 0.25 |
| 0.3 | 0.02546 | 0.061759 | 0.112694 | 0.177367 | 0.231589 | 0.3 |
| 0.35 | 0.031626 | 0.076413 | 0.138032 | 0.213524 | 0.274613 | 0.35 |
| 0.4 | 0.038702 | 0.093054 | 0.166088 | 0.252053 | 0.319078 | 0.4 |
| 0.45 | 0.046929 | 0.112143 | 0.197321 | 0.29317 | 0.365058 | 0.45 |
| 0.5 | 0.056639 | 0.134292 | 0.232274 | 0.337106 | 0.412628 | 0.5 |
| 0.55 | 0.068305 | 0.160317 | 0.271598 | 0.384111 | 0.461868 | 0.55 |
| 0.6 | 0.082621 | 0.191324 | 0.316063 | 0.434452 | 0.512864 | 0.6 |
| 0.65 | 0.100646 | 0.228841 | 0.366573 | 0.488412 | 0.565709 | 0.65 |
| 0.7 | 0.124079 | 0.274996 | 0.424187 | 0.546297 | 0.620505 | 0.7 |
| 0.75 | 0.155797 | 0.332781 | 0.490118 | 0.608437 | 0.677368 | 0.75 |
| 0.8 | 0.201036 | 0.406394 | 0.565734 | 0.675197 | 0.736435 | 0.8 |
| 0.85 | 0.270142 | 0.501603 | 0.652549 | 0.746993 | 0.797878 | 0.85 |
| 0.9 | 0.385426 | 0.625941 | 0.752238 | 0.824349 | 0.86193 | 0.9 |
| 0.95 | 0.59705 | 0.788409 | 0.86678 | 0.908035 | 0.928971 | 0.95 |
| 1 | 1 | 1 | 1 | 1 | 1 | 1 |

3. Now, draw the graphs in EXCEL:


Software Solutions to Problems on Heat Transfer: Radiation Heat Transfer - Part I

View factors (or, Shape factor, configuration factor) and Radiation energy exchange between Black bodies:

Prob.5B.15. Draw the View factor graph for F_12 for two concentric cylinders of finite length:

## EXCEL Solution:

1. Set up a Table in EXCEL worksheet as shown below:



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In the above screen shot, column D contains R1/R2 values and row 483 contains L/R2 values.

We use the VBA Function written above to calculate the view factor $\mathrm{F}_{-} 12$ (i.e. view factor from surface 1 to surface 2), for given values of R1/R2 and L/R2. Formula entered in cell E484 can be seen in the Formula bar. Use relative references to R1/R2 so that we can drag-copy to fill up the Table later. Similarly, write formulas for other L/R2 values as shown above.
2. Now, select cells E484 to K484 and drag-copy to the end of the Table, i.e. up to the cell K503. The calculations are done immediately and the Table is filled up:

| (R1_by_R2)\L/R2 | 0.1 | 0.25 | 0.5 | 1 | 2 | 4 | $1.00 \mathrm{E}+07$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 0.001 | 0.063501 | 0.156076 | 0.295367 | 0.50025 | 0.705033 | 0.844159 | 1 |
| 0.1 | 0.068977 | 0.169028 | 0.317088 | 0.526657 | 0.725605 | 0.856116 | 1 |
| 0.15 | 0.072242 | 0.176691 | 0.329665 | 0.541305 | 0.736581 | 0.862386 | 1 |
| 0.2 | 0.075922 | 0.185272 | 0.343501 | 0.556909 | 0.74796 | 0.868814 | 1 |
| 0.25 | 0.080096 | 0.194929 | 0.358764 | 0.573527 | 0.759752 | 0.875401 | 1 |
| 0.3 | 0.084865 | 0.205863 | 0.375646 | 0.591225 | 0.771964 | 0.882152 | 1 |
| 0.35 | 0.09036 | 0.218324 | 0.394376 | 0.610069 | 0.784608 | 0.889069 | 1 |
| 0.4 | 0.096756 | 0.232634 | 0.415221 | 0.630133 | 0.797696 | 0.896156 | 1 |
| 0.45 | 0.104287 | 0.249207 | 0.438491 | 0.651488 | 0.81124 | 0.903421 | 1 |
| 0.5 | 0.113279 | 0.268584 | 0.464549 | 0.674212 | 0.825256 | 0.910868 | 1 |
| 0.55 | 0.124191 | 0.291485 | 0.493815 | 0.698384 | 0.83976 | 0.918508 | 1 |
| 0.6 | 0.137702 | 0.318874 | 0.526771 | 0.724086 | 0.854773 | 0.92635 | 1 |
| 0.65 | 0.154841 | 0.352063 | 0.563959 | 0.751403 | 0.870321 | 0.934408 | 1 |
| 0.7 | 0.177256 | 0.392851 | 0.605982 | 0.780424 | 0.886435 | 0.942699 | 1 |
| 0.75 | 0.20773 | 0.443708 | 0.653491 | 0.81125 | 0.903157 | 0.951246 | 1 |
| 0.8 | 0.251295 | 0.507992 | 0.707167 | 0.843996 | 0.920544 | 0.960079 | 1 |
| 0.85 | 0.317814 | 0.590121 | 0.767705 | 0.878816 | 0.93868 | 0.969245 | 1 |
| 0.9 | 0.428251 | 0.69549 | 0.83582 | 0.915943 | 0.9577 | 0.978815 | 1 |
| 0.95 | 0.628474 | 0.829904 | 0.9124 | 0.955826 | 0.977864 | 0.988926 | 1 |
| 1 | 1 | 1 | 1 | 1 | 1 | 1 | 1 |

Software Solutions to Problems on Heat Transfer: Radiation Heat Transfer - Part I

View factors (or, Shape factor, configuration factor) and Radiation energy exchange between Black bodies:
3. Now, draw the graphs in EXCEL:


Prob.5B.16. Draw the View factor graph for F_22 for two concentric cylinders of finite length:

## EXCEL Solution:

1. Set up a Table in EXCEL worksheet as shown below:


In the above screen shot, column D contains R1/R2 values and row 432 contains $\mathrm{L} / \mathrm{R} 2$ values.

We use the VBA Function written above to calculate the view factor F _ 22 (i.e. view factor from surface 2 to itself), for given values of R1/R2 and L/R2. Formula entered in cell E433 can be seen in the Formula bar. Use relative references to R1/R2 so that we can drag-copy to fill up the Table later. Similarly, write formulas for other $\mathrm{L} / \mathrm{R} 2$ values as shown above.


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Software Solutions to Problems on Heat Transfer: Radiation Heat Transfer - Part I
2. Now, select cells E433 to K433 and drag-copy to the end of the Table, i.e. up to the cell K452. The calculations are done immediately and the Table is filled up:

| (R1_by_R2) \L/R2 | 0.1 | 0.25 | 0.5 | 1 | 2 | 4 | $1.00 \mathrm{E}+07$ |
| :---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| 0.001 | 0.04872 | 0.11714 | 0.21907 | 0.38167 | 0.58529 | 0.76323 | 0.999 |
| 0.1 | 0.04557 | 0.10929 | 0.2036 | 0.35241 | 0.53573 | 0.69341 | 0.9 |
| 0.15 | 0.04396 | 0.1053 | 0.19574 | 0.33755 | 0.51061 | 0.65806 | 0.85 |
| 0.2 | 0.04235 | 0.10128 | 0.18783 | 0.32259 | 0.48536 | 0.62262 | 0.8 |
| 0.25 | 0.04071 | 0.09722 | 0.17983 | 0.30749 | 0.45994 | 0.58705 | 0.75 |
| 0.3 | 0.03906 | 0.0931 | 0.17173 | 0.29223 | 0.43431 | 0.55131 | 0.7 |
| 0.35 | 0.03738 | 0.08892 | 0.16351 | 0.27674 | 0.40842 | 0.51537 | 0.65 |
| 0.4 | 0.03566 | 0.08465 | 0.15512 | 0.261 | 0.38222 | 0.47919 | 0.6 |
| 0.45 | 0.03391 | 0.08028 | 0.14654 | 0.24493 | 0.35563 | 0.44271 | 0.55 |
| 0.5 | 0.0321 | 0.07579 | 0.13773 | 0.22848 | 0.3286 | 0.4059 | 0.5 |
| 0.55 | 0.03023 | 0.07114 | 0.12863 | 0.21157 | 0.30104 | 0.36871 | 0.45 |
| 0.6 | 0.02829 | 0.06632 | 0.11919 | 0.1941 | 0.27286 | 0.33107 | 0.4 |
| 0.65 | 0.02625 | 0.06126 | 0.10932 | 0.17596 | 0.24395 | 0.29291 | 0.35 |
| 0.7 | 0.0241 | 0.05591 | 0.09891 | 0.157 | 0.21418 | 0.25415 | 0.3 |
| 0.75 | 0.02179 | 0.05019 | 0.08782 | 0.137 | 0.18337 | 0.21471 | 0.25 |
| 0.8 | 0.01927 | 0.04397 | 0.07582 | 0.11567 | 0.1513 | 0.17444 | 0.2 |
| 0.85 | 0.01645 | 0.03703 | 0.06253 | 0.09259 | 0.11765 | 0.1332 | 0.15 |
| 0.9 | 0.01316 | 0.02895 | 0.0473 | 0.06702 | 0.08197 | 0.09074 | 0.1 |
| 0.95 | 0.00894 | 0.0187 | 0.02861 | 0.03758 | 0.04347 | 0.04669 | 0.05 |
| 1 | 0 | 0 | 0 | $5.6 \mathrm{E}-17$ | $-2.8 \mathrm{E}-17$ | $4.2 \mathrm{E}-17$ | $2.9 \mathrm{E}-12$ |

3. Now, draw the graphs in EXCEL:


Prob.5B.17. Consider a hemispherical oven of 1 m diameter. Find the View factor of curved surface to itself. (= F22). The curved surface is at 800 K and the base is at 300 K . If both the surfaces can be considered as black, determine the heat transfer rate per unit length. Assume the furnace to be very long.


## Fig.Prob.5B. 17

## EXCEL Solution:

Since the furnace is very long, there is no heat loss at the ends.

Excel worksheet is shown below. Formulas used are also shown in the worksheet.


Software Solutions to Problems on Heat Transfer: Radiation Heat Transfer - Part I

View factors (or, Shape factor, configuration factor) and Radiation energy exchange between Black bodies:


Thus:
F22 $=$ View factor of curved surface to itself $=0.36338 \ldots$ Ans.

Q12 $=$ Net heat transfer between surface 2 and $1=22765$ W...Ans.

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Prob.5B.18. Consider a thin, hollow cylinder of 10 cm dia and 10 cm length, as shown in Fig. below.
a) What is the view factor of the curved surface to itself (i.e. F_33)?
b) If $\mathrm{T} 1=800 \mathrm{~K}, \mathrm{~T} 2=400 \mathrm{~K}$, find the net heat transfer between surfaces 1 and 2 (Q_12), considering all surfaces as black.
c) Also, plot the variation of F_33 and Q_12 as L varies from 0.1 m to 0.5 m


## EXCEL Solution:

Top, bottom and side surfaces are designated as 1,2 and 3 respectively.

Let the radius be R and length, L .

Excel worksheet is shown below. Formulas used are also shown in the worksheet.

Following are the steps:

1. Set up the EXCEL worksheet, enter data:

|  | A | B | C | D | E |  | F |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 565 |  |  | Data: |  |  |  |  |
| 566 |  |  |  | R | $\mathbf{0 . 0 5}$ | m |  |
| 567 |  |  |  | T1 | $\mathbf{8 0 0}$ | K |  |
| 568 |  |  |  | T2 | $\mathbf{4 0 0}$ | K |  |
| 569 |  |  |  | L | $\mathbf{0 . 1}$ | m |  |
| 570 |  |  |  | sigma | $\mathbf{5 . 6 7 E - 0 8}$ | $\mathrm{W} / \mathrm{m}^{\wedge} 2 . \mathrm{K}^{\wedge} 4$ |  |
| 571 |  |  |  |  |  |  |  |

Software Solutions to Problems on Heat Transfer: Radiation Heat Transfer - Part I

View factors (or, Shape factor, configuration factor) and Radiation energy exchange between Black bodies:
2. Perform the calculations:


Note that F_12 is calculated using the VBA Function written above for coaxial, parallel disks. See the Formula bar in the above screen shot.

Continuing, we have:

| 4 | A | B | C | D | E | F |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 587 |  |  |  |  |  |  |
| 588 |  |  | Calculate F_31 using the reciprocity relation for surfaces 1 and 3: |  |  |  |
| 589 |  |  |  |  |  |  |
| 590 |  |  |  | i.e. $\mathrm{A} 1{ }^{*} \mathrm{~F} 13=\mathrm{A} 3$ * F 31 |  |  |
| 591 |  |  |  |  |  |  |
| 592 |  |  | Therefore: | F_31 $=\left(\right.$ A1 $\left.{ }^{*} \mathrm{~F}_{2} 13\right) / \mathrm{A} 3=$ | 0.207106781 |  |
| 593 |  |  |  |  |  |  |
| 594 |  |  | For surface 3: |  |  |  |
| 595 |  |  |  | F_31+F_32+F_33 = 1 .... By Summation rule |  |  |
| 596 |  |  |  |  |  |  |
| 597 |  |  | Also, for surface 3: | F_31 = F_32 ... by Symmetry |  |  |
| 598 |  |  | Therefore: | F_32 = | 0.207106781 |  |
| 599 |  |  |  |  |  |  |
| 600 |  |  | Therefore: | F_33 $=1$ - (F_31+F_32) |  |  |
| 601 |  |  |  |  |  |  |
| 602 |  |  | So, we get: | F_33 = | 0.5858 | ...Ans. |
| 603 |  |  |  |  |  |  |
| 604 |  |  | Net heat transfer Q_12: |  |  |  |
| 605 |  |  |  | Q_12-A1 * F_12 * sigma ${ }^{\text {* }}$ (T1^4 |  |  |
| 606 |  |  |  |  |  |  |
| 607 |  |  |  | Q_12 = | 29.340 | W ... Ans. |

## Thus:

View factor of the top surface to bottom surface (i.e. $\mathrm{F}_{-} 12$ ) $=\mathbf{0 . 1 7 1 6} \ldots$ Ans.

View factor of the curved surface to itself (i.e. F_33) $=0.5858$... Ans.

Net heat transfer between surfaces 1 and $2=$ Q_12 $=29.34 \mathrm{~W} .$. Ans.

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## To plot the variation of $F \_33$ and $Q \_12$ as $L$ varies from 0.1 m to 0.5 m :

3. Set up a Table as shown below. Remember to enter $L$ with relative reference in the equations for $\mathrm{F} \_12, \mathrm{~F} \_13 \ldots$ etc.


Equation entered for $\mathrm{F}_{-} 12$ in cell D613 can be seen in the Formula bar in the above screen shot.
4. Now, select cells D613 to H613 and drag-copy to the end of the Table, i.e. up to cell H633. Then, immediately, the calculations are done and the Table gets filled up:

| $\mathrm{L}(\mathrm{m})$ | F_12 | F_13 | F_31 | F_33 | Q_12 (W) |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 0.1 | 0.1716 | 0.8284 | 0.2071 | 0.5858 | 29.340 |
| 0.12 | 0.1311 | 0.8689 | 0.1810 | 0.6380 | 22.415 |
| 0.14 | 0.1027 | 0.8973 | 0.1602 | 0.6795 | 17.562 |
| 0.16 | 0.0823 | 0.9177 | 0.1434 | 0.7132 | 14.065 |
| 0.18 | 0.0671 | 0.9329 | 0.1296 | 0.7409 | 11.482 |
| 0.2 | 0.0557 | 0.9443 | 0.1180 | 0.7639 | 9.530 |
| 0.22 | 0.0469 | 0.9531 | 0.1083 | 0.7834 | 8.023 |
| 0.24 | 0.0400 | 0.9600 | 0.1000 | 0.8000 | 6.840 |
| 0.26 | 0.0345 | 0.9655 | 0.0928 | 0.8143 | 5.896 |
| 0.28 | 0.0300 | 0.9700 | 0.0866 | 0.8268 | 5.131 |
| 0.3 | 0.0263 | 0.9737 | 0.0811 | 0.8377 | 4.503 |
| 0.32 | 0.0233 | 0.9767 | 0.0763 | 0.8474 | 3.983 |
| 0.34 | 0.0207 | 0.9793 | 0.0720 | 0.8560 | 3.546 |
| 0.36 | 0.0186 | 0.9814 | 0.0682 | 0.8637 | 3.177 |
| 0.38 | 0.0167 | 0.9833 | 0.0647 | 0.8706 | 2.862 |
| 0.4 | 0.0152 | 0.9848 | 0.0616 | 0.8769 | 2.592 |
| 0.42 | 0.0138 | 0.9862 | 0.0587 | 0.8826 | 2.357 |
| 0.44 | 0.0126 | 0.9874 | 0.0561 | 0.8878 | 2.153 |
| 0.46 | 0.0115 | 0.9885 | 0.0537 | 0.8926 | 1.974 |
| 0.48 | 0.0106 | 0.9894 | 0.0515 | 0.8969 | 1.816 |
| 0.5 | 0.0098 | 0.9902 | 0.0495 | 0.9010 | 1.677 |

Software Solutions to Problems on Heat Transfer: Radiation Heat Transfer - Part I

View factors (or, Shape factor, configuration factor) and Radiation energy exchange between Black bodies:
5. Now, plot the results in EXCEL:



To see the Part II download:
Software Solutions to problems on heat transfer - Radiation Heat Transfer II

